



# THIRD AIAA SONIC BOOM PREDICTION WORKSHOP NEARFIELD CFD SUMMARY

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NASA Langley Research  
Center

# MOTIVATION

Commercial supersonic overland flight is currently prohibited

- Supersonic overland flight is an enabler for entry into new vehicle market



Replacing the prohibition with a certification standard requires an international effort to quantify the accuracy and reliability of prediction methods



Deficiencies in existing methods should be noted to focus research on addressing weaknesses



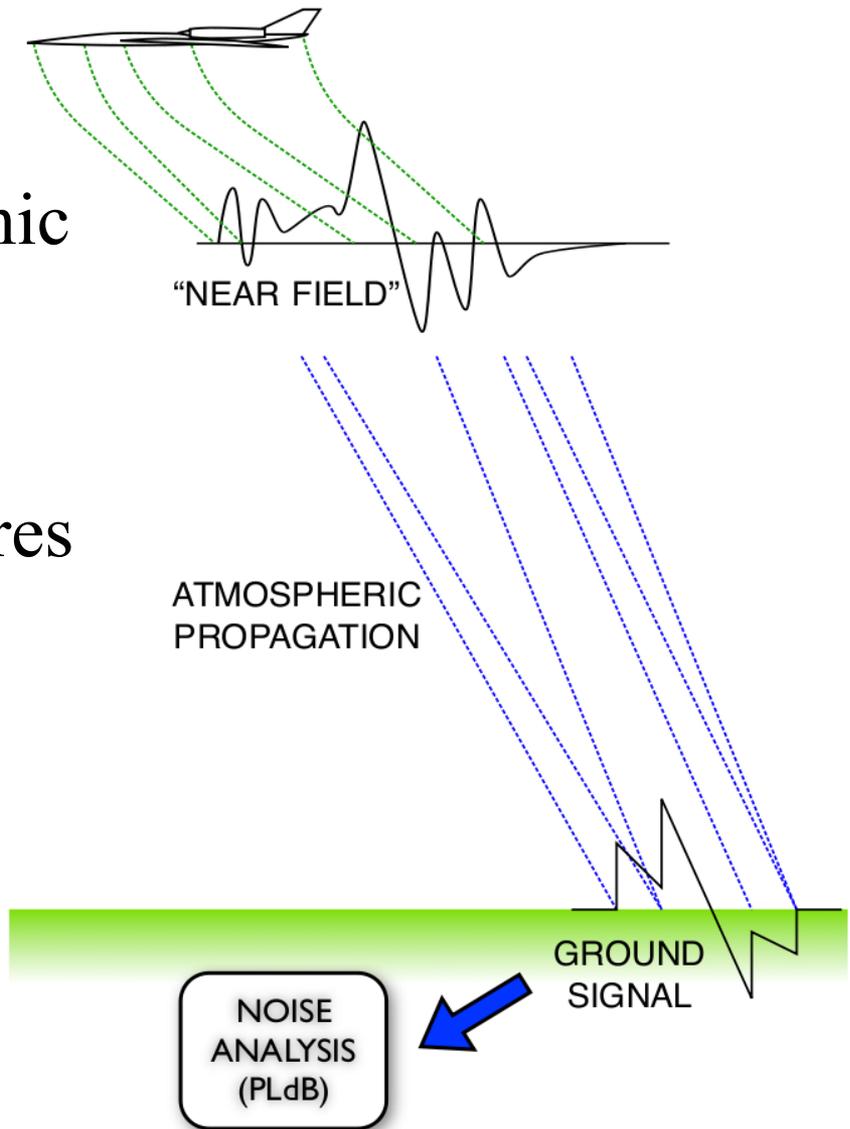
# MOTIVATION

Near field CFD is part of sonic boom prediction

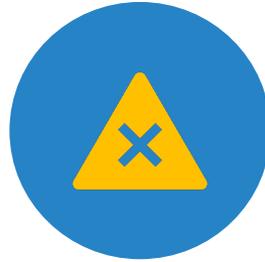
Explore the issues

Impartially compare signatures by uniform application of

- Near field statistics
- Propagation
- Loudness measures



# WORKSHOP CULTURE



Adjectives such as good, bad, right, and wrong oversimplify issues and should be avoided



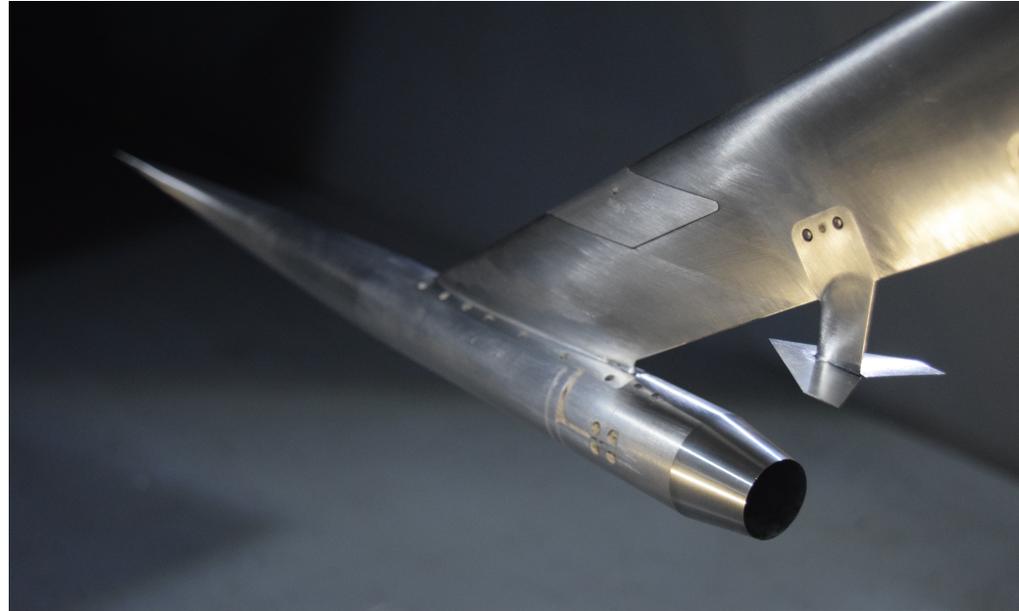
Focus on describing observed differences and communicate why things are different

# MODELS AND CASES

Ames 9'x7' UPWT  
Biconvex Plume-Shock  
Interaction Case

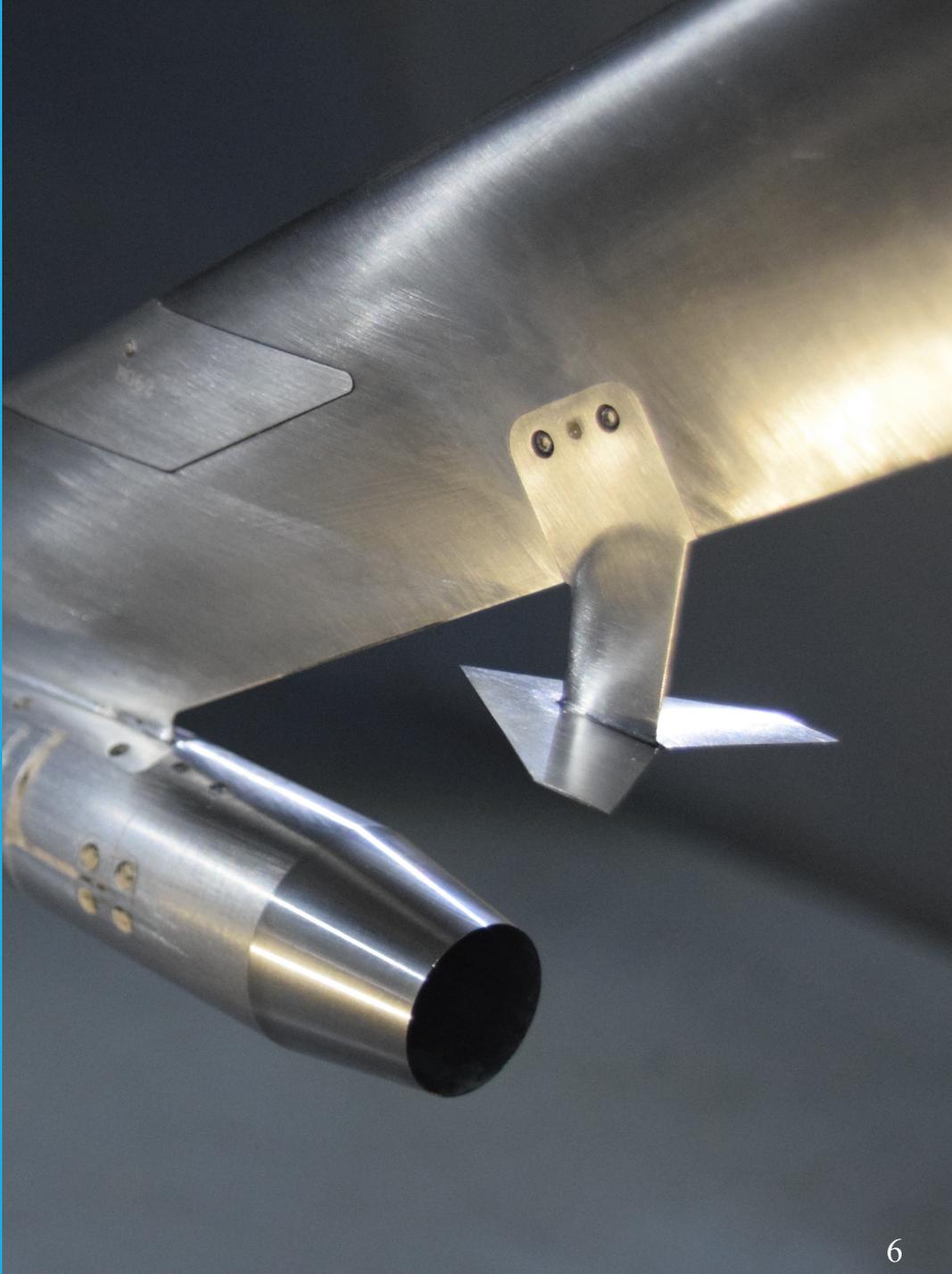
C608, an early X-59  
Prototype

IGES and STEP  
geometry files along with  
workshop generated grids  
provided



BICONVEX  
SUBMISSIONS

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# OUTLINE

- Fine-grid nearfield pressures
  - Excluded different geometry (AE, AF) and optional case with two submissions (AD, OE)
- Compare the fine-grid ensemble pointwise standard deviation to experiment
- Identify outliers

# BICONVEX $dp/p_{INF}$

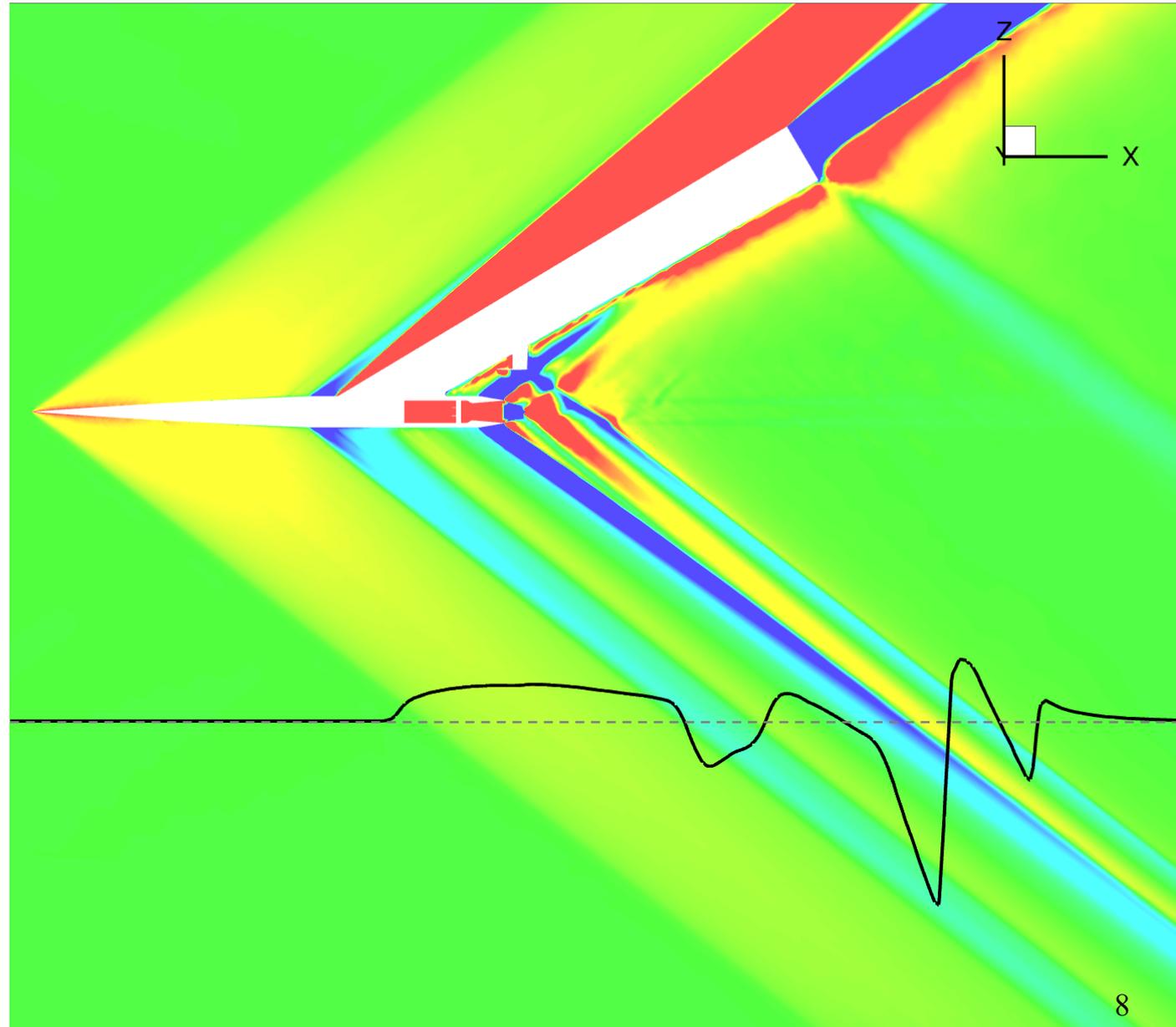
Grid 100

Tetrahedral

USM3D

Production  
code

$dp/p_{\infty}$  which  
is the pressure  
disturbance  
normalized by  
freestream  
pressure

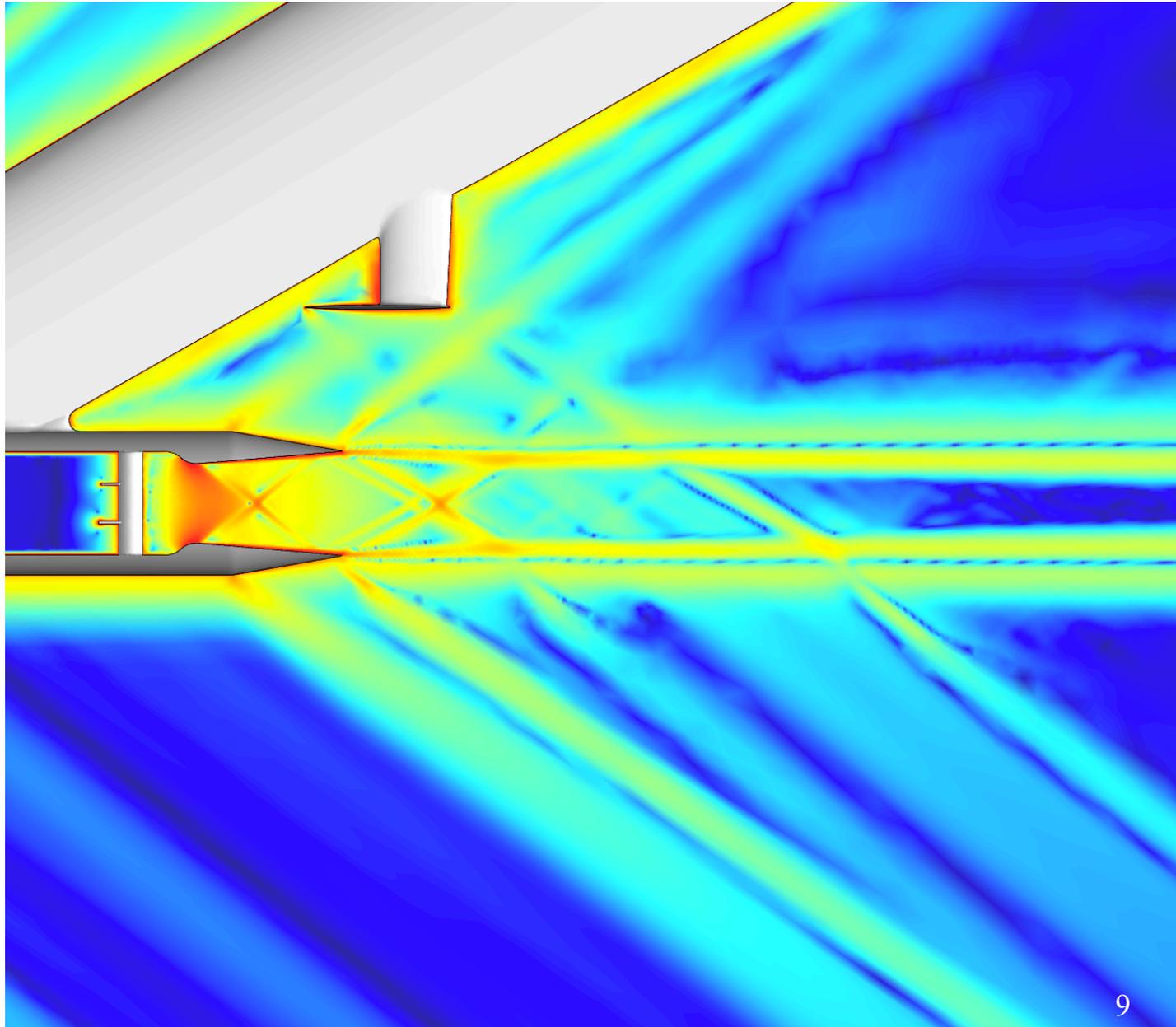


# BICONVEX DENSITY GRADIENT

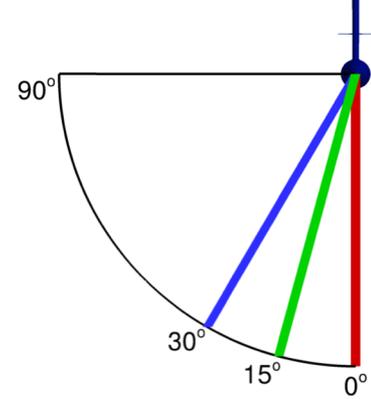
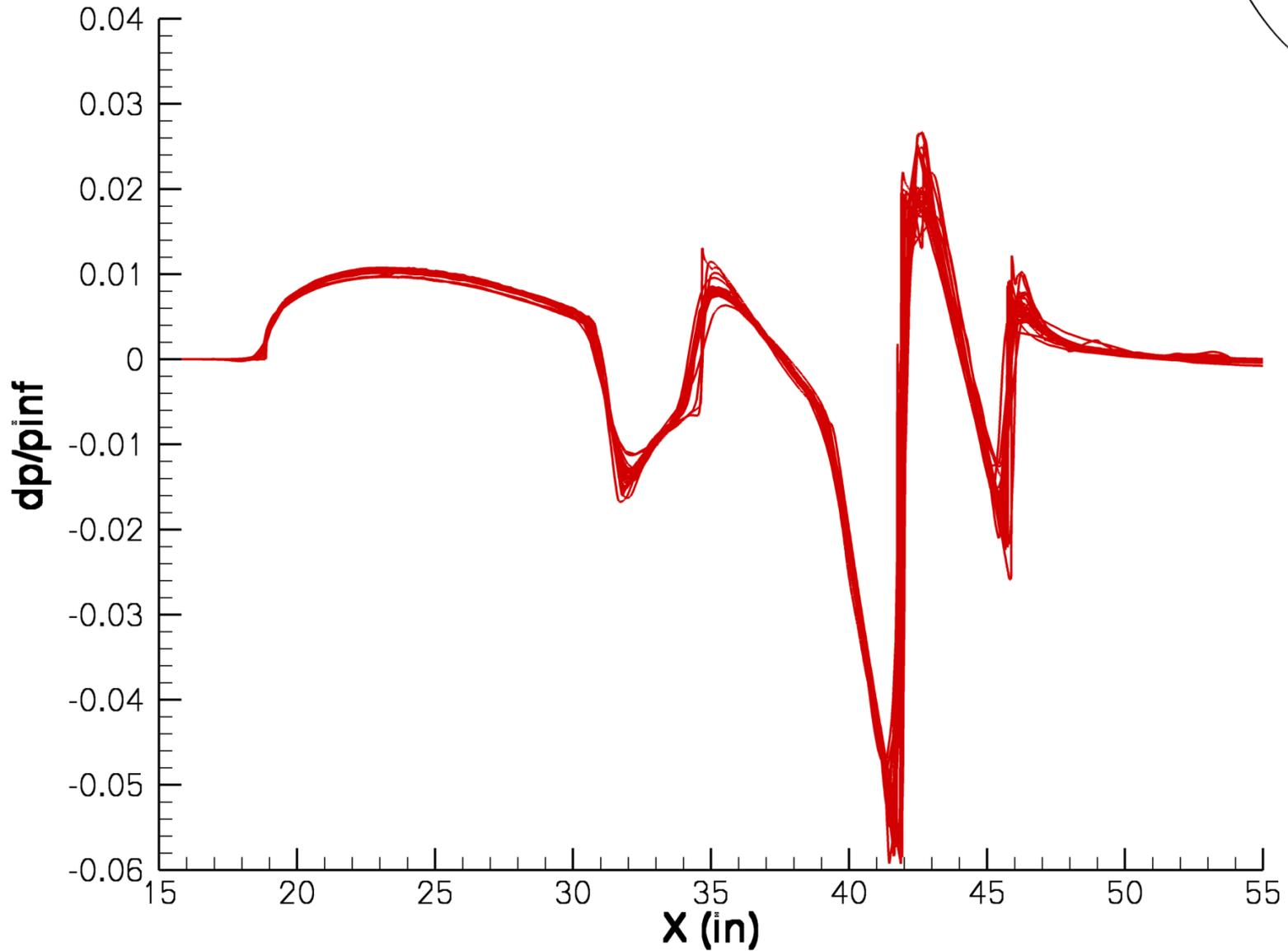
Grid 100  
Tetrahedral

USM3D  
Production  
code

Density  
gradient  
(numerical  
schlieren)



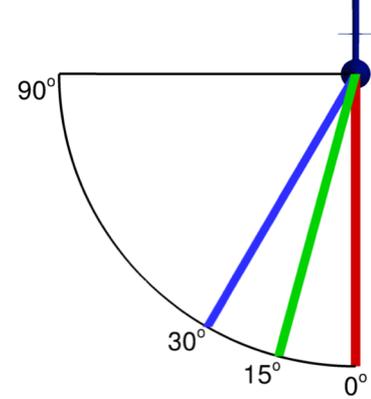
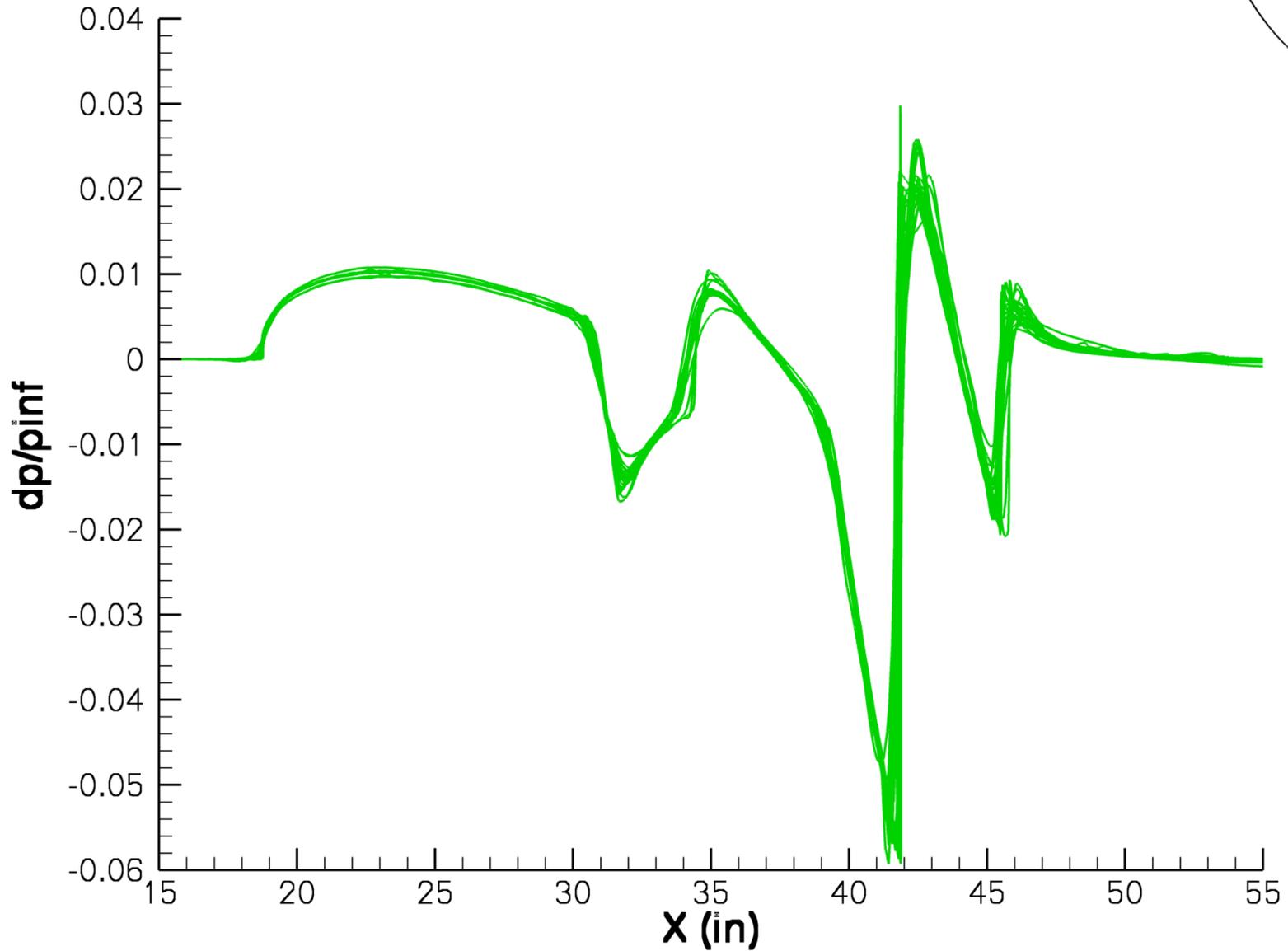
# BICONVEX FINE-GRID



PHI=00

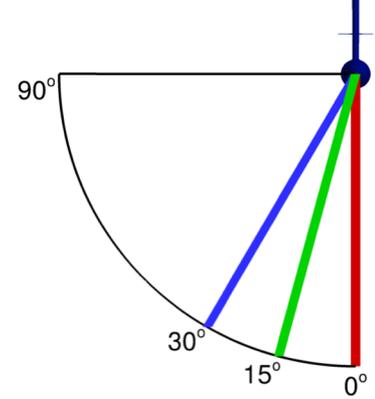
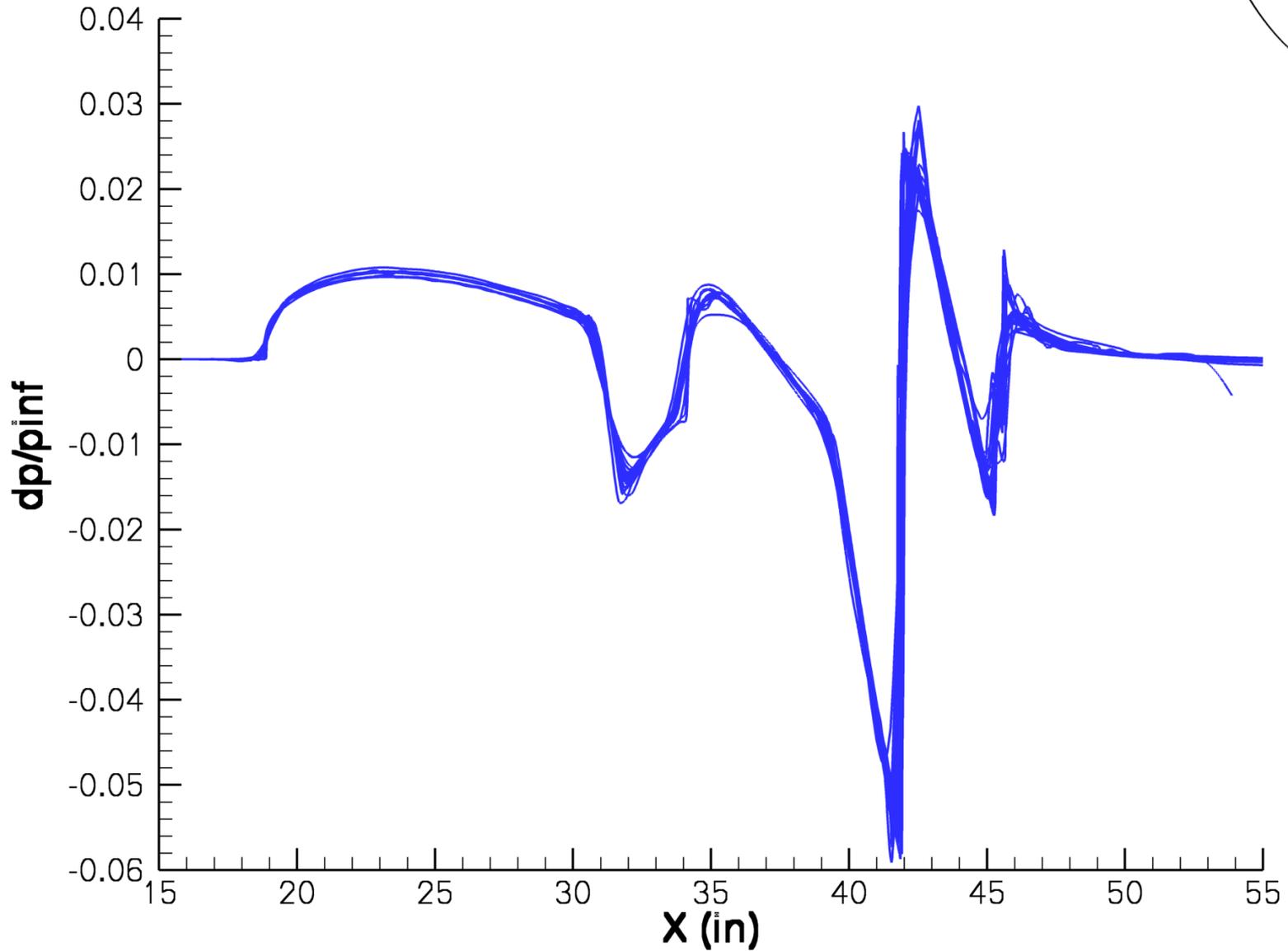
N=31

# BICONVEX FINE-GRID



$\text{PHI}=15$   
 $N=31$

# BICONVEX FINE-GRID



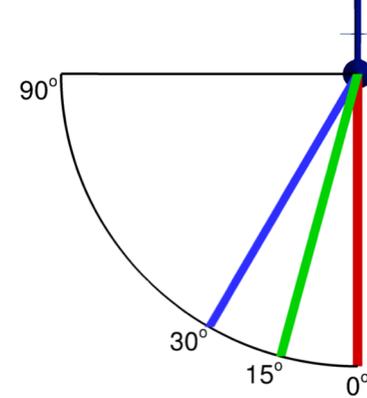
PHI=30

N=31

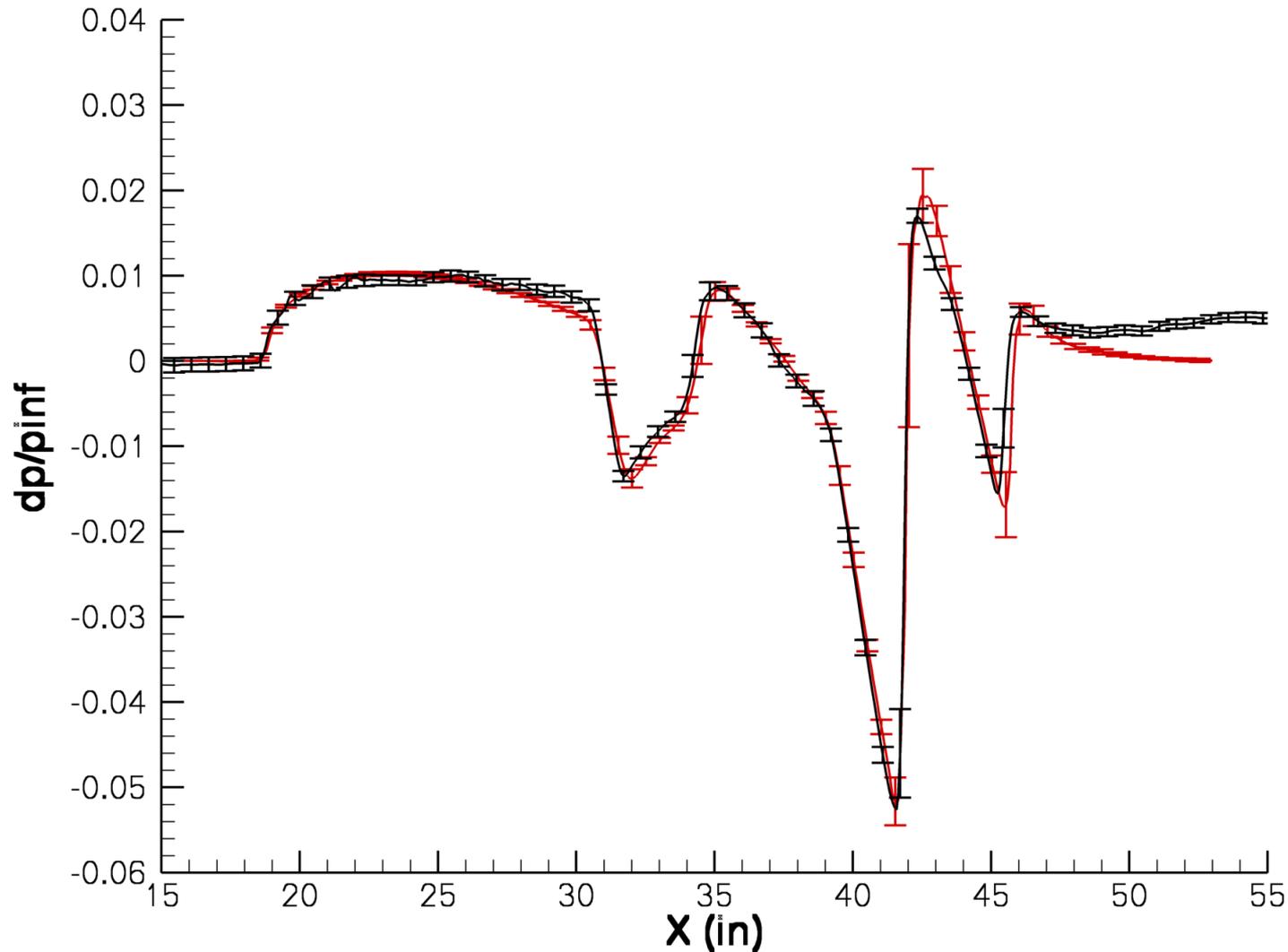
# NEARFIELD SIGNATURE STATISTICS

- Pointwise population mean and standard deviation of interpolated signature every 0.05 inch (N=32)
- Analogous to wind tunnel spatial averaging
- Finest grid solution from each participant (which vary in resolution)
- Outliers impact these statistics

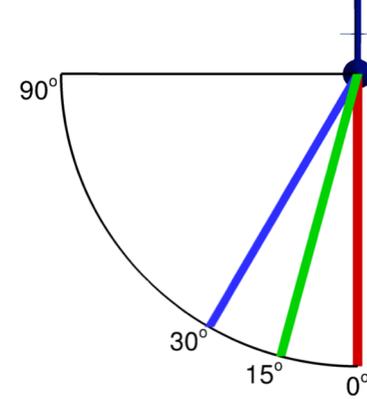
# BICONVEX FINE-GRID ENSEMBLE COMPARISON WITH EXPERIMENT



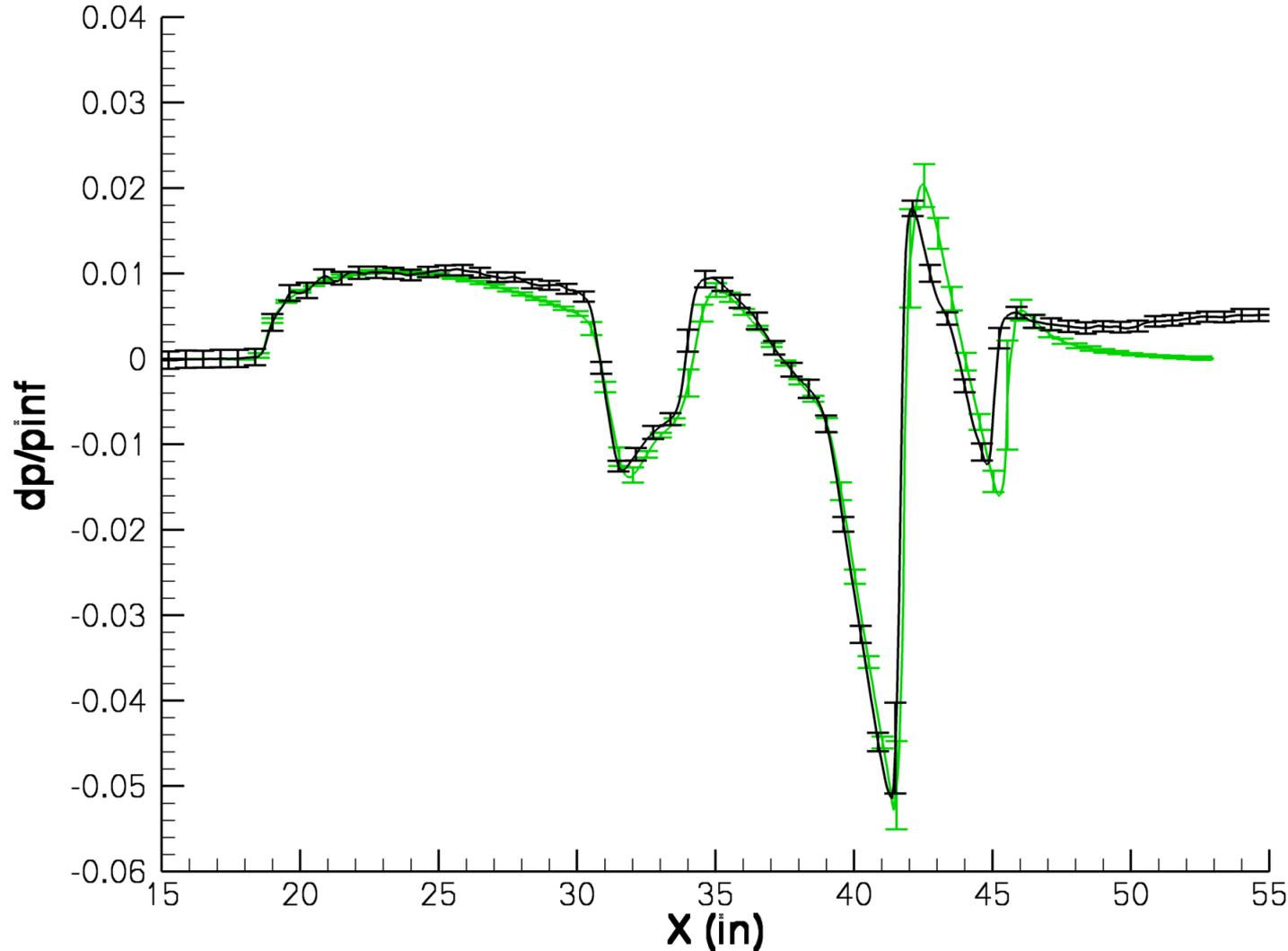
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N=31



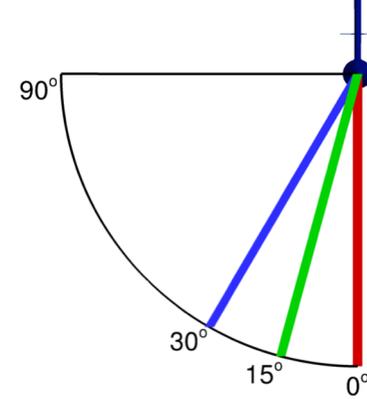
# BICONVEX FINE-GRID ENSEMBLE COMPARISON WITH EXPERIMENT



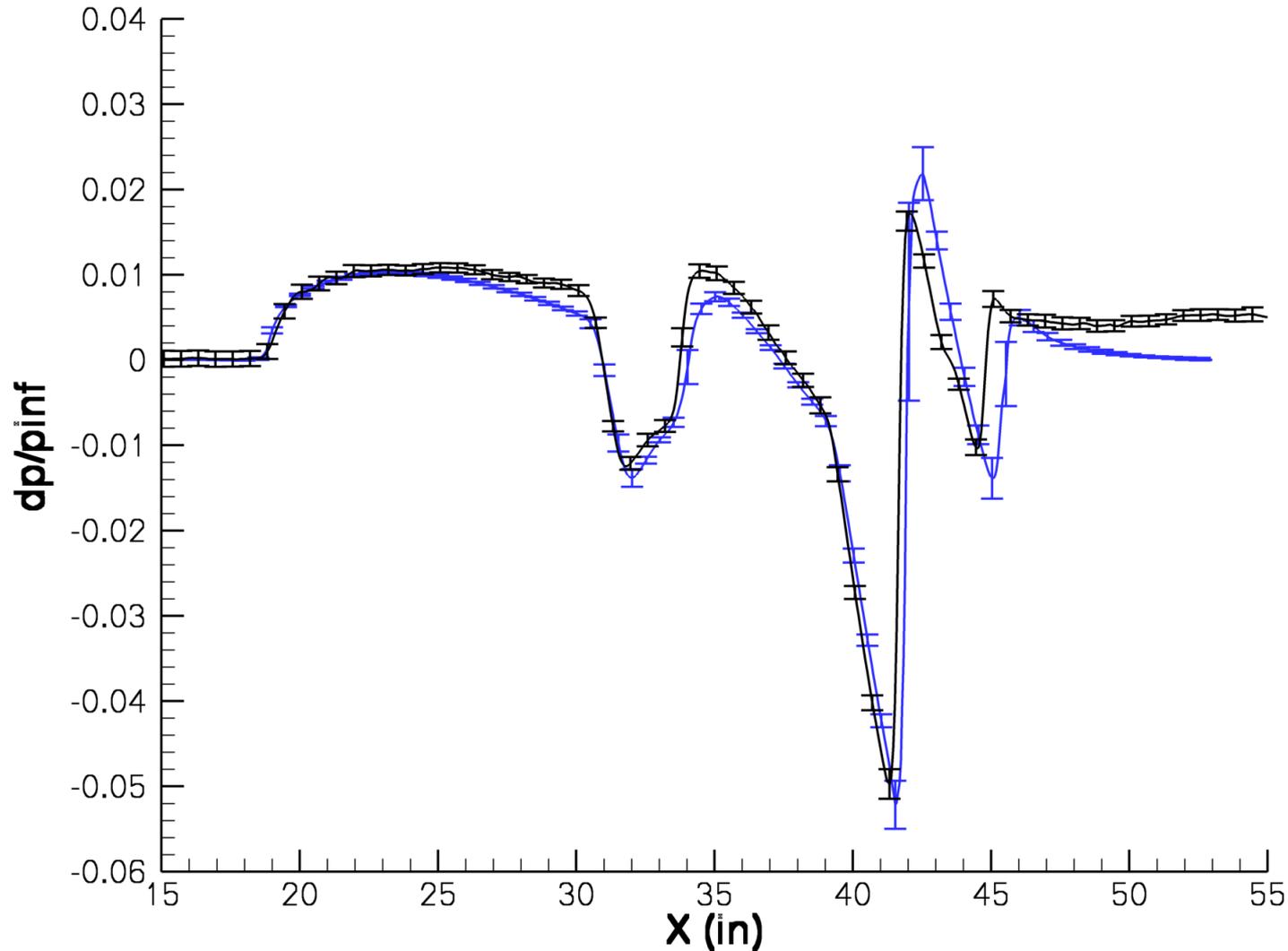
PHI=15  
N=31



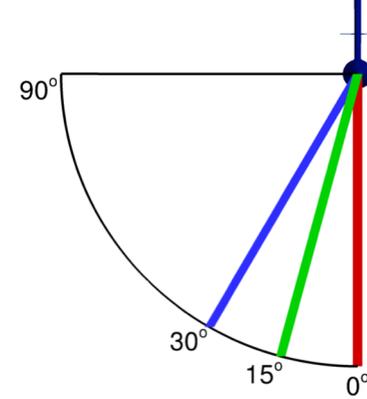
# BICONVEX FINE-GRID ENSEMBLE COMPARISON WITH EXPERIMENT



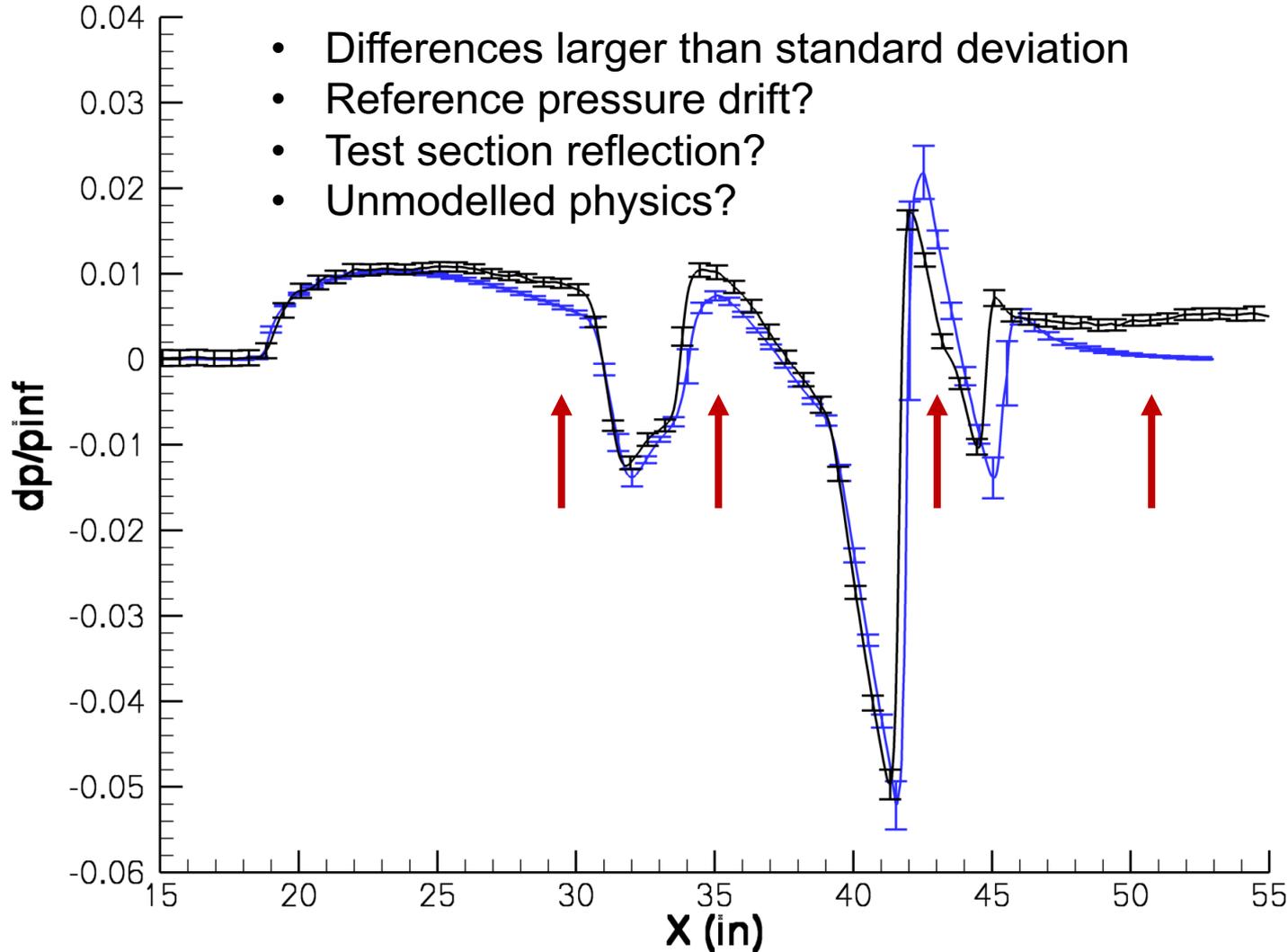
PHI=30  
N=31



# BICONVEX FINE-GRID ENSEMBLE COMPARISON WITH EXPERIMENT



PHI=30  
N=31

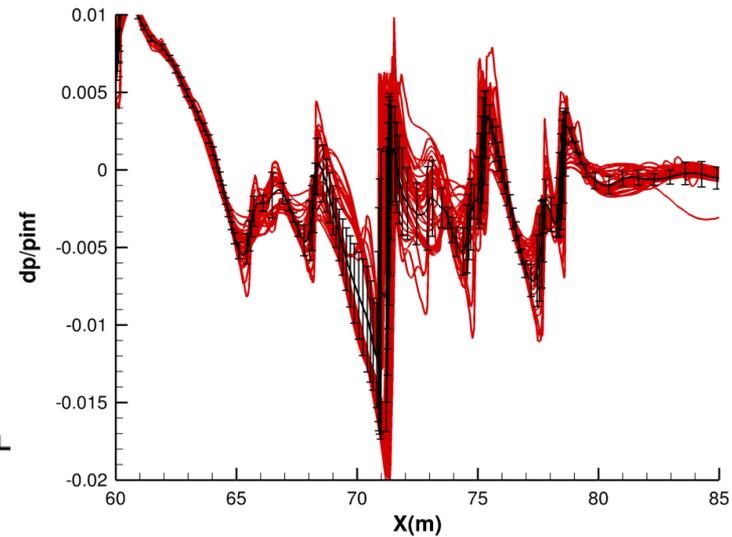
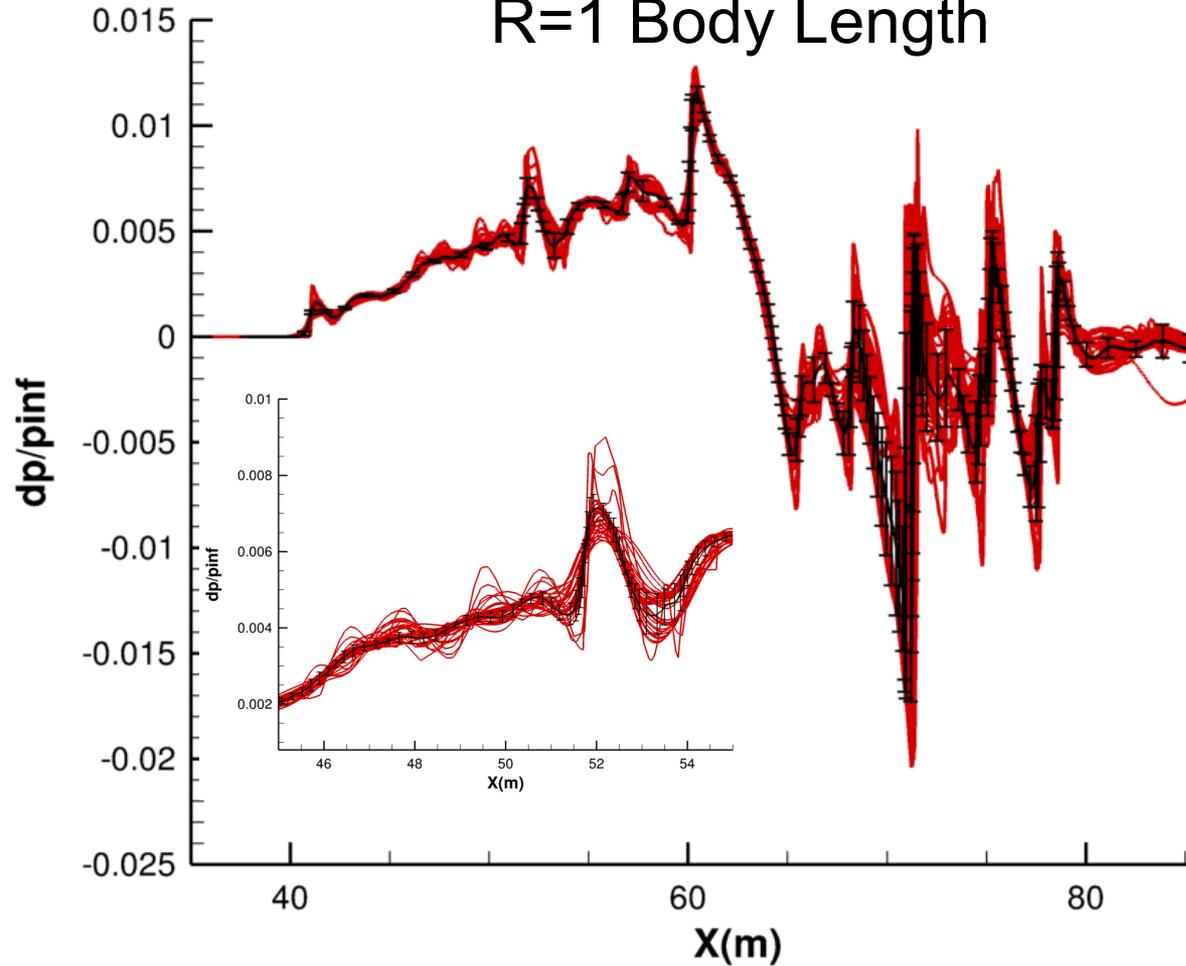
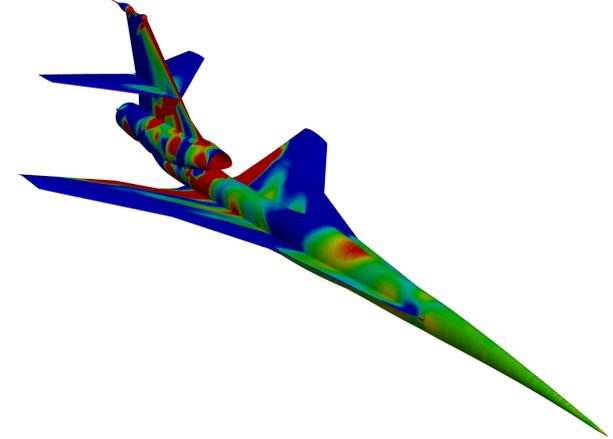


# IDENTIFICATION OF OUTLIERS

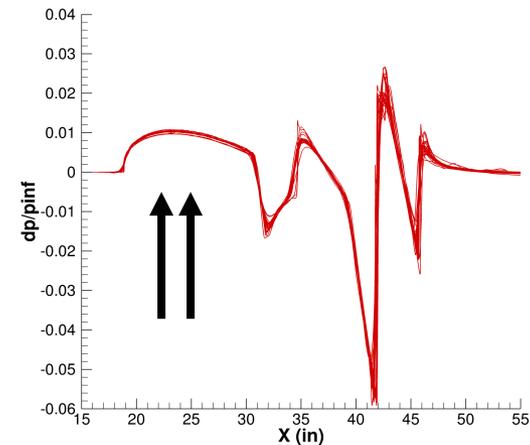
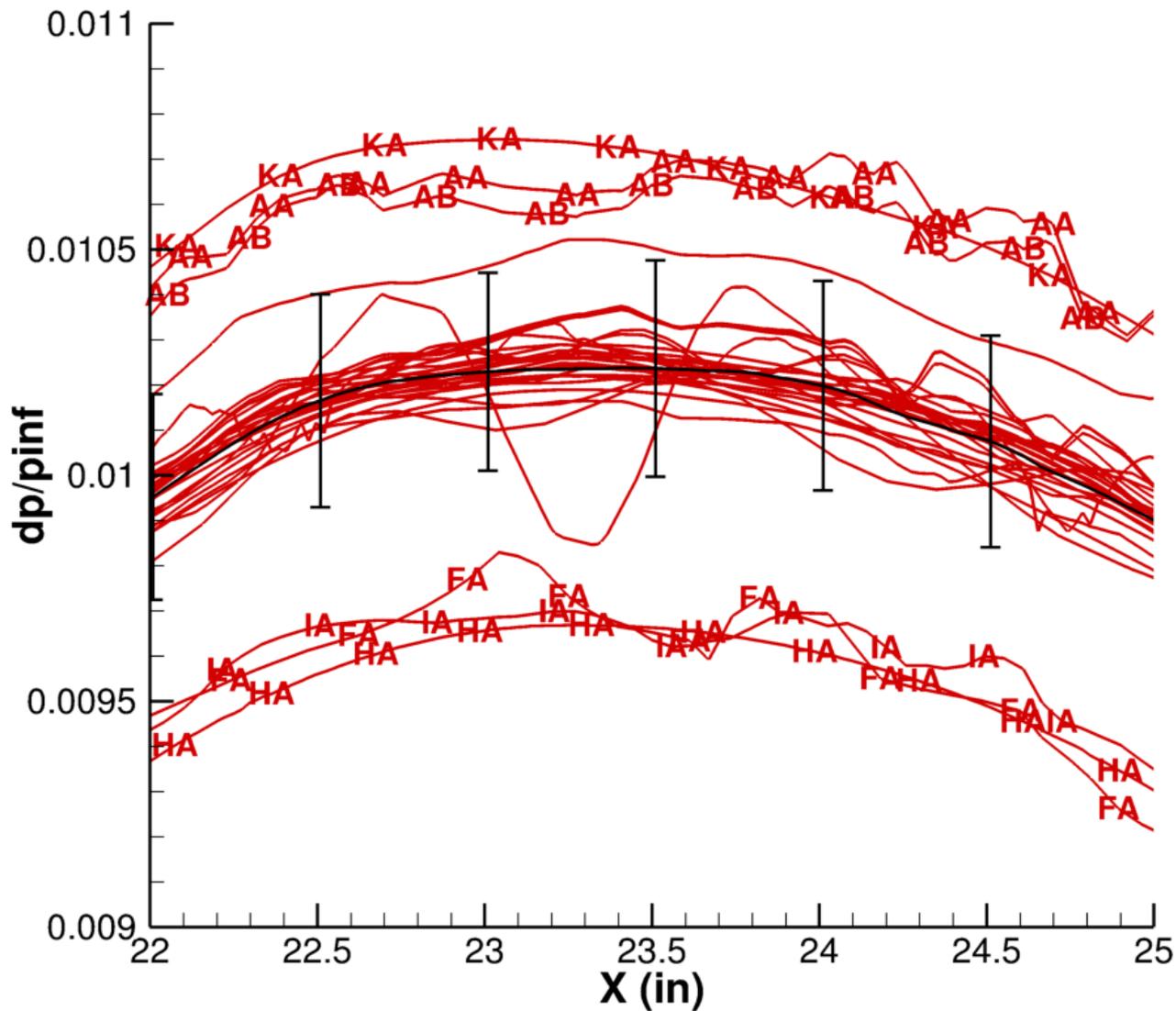
- Goal is an objective tool to identify and learn from differences in submissions
- Pointwise standard deviation is an imperfect tool
  - Not suited to small sample size
  - Distribution of submissions is not normal
  - Should use Functional Data Analysis (FDA) for shape as well as magnitude outlier identification
- Previous workshop used box and whisker plots with an effective coverage factor of 2 (exceed 95% likelihood)
- Coverage factor of 1 (exceed 68% likelihood) used here
- Focus on submissions exceeding 1 standard deviation for forebody and plume (avoid shocks)

# SBPW2 C25P CONTEXT

PHI=00  
R=1 Body Length

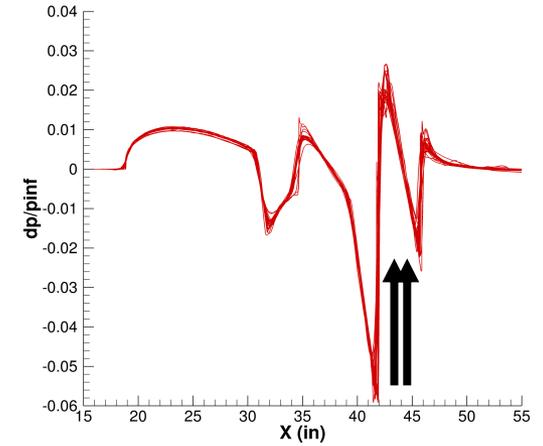
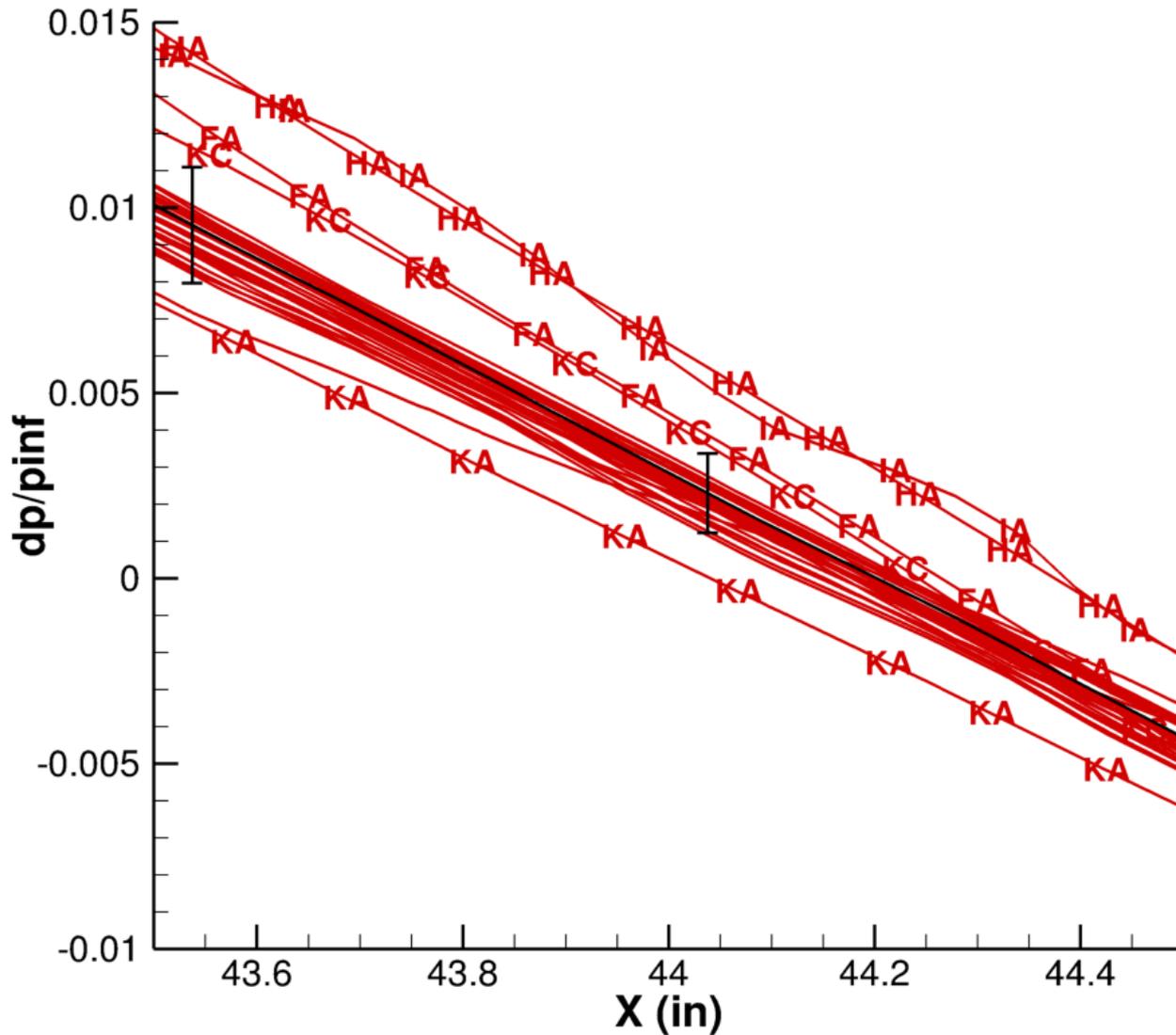


# BICONVEX FOREBODY



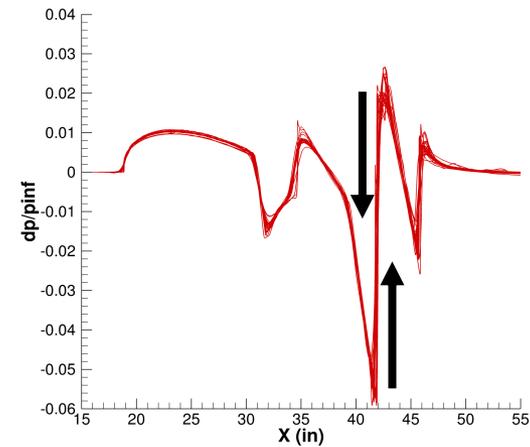
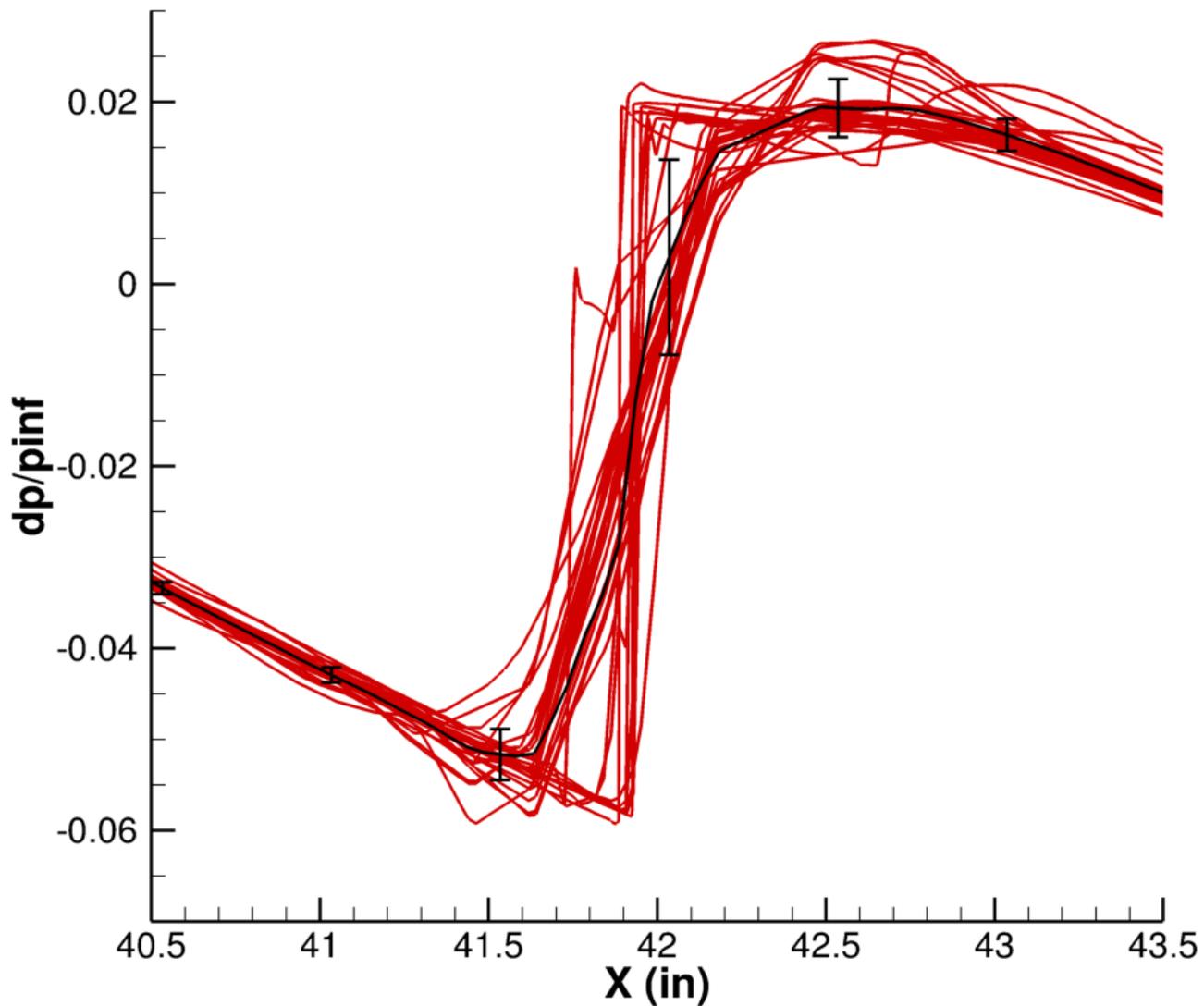
PHI=00

# BICONVEX PLUME



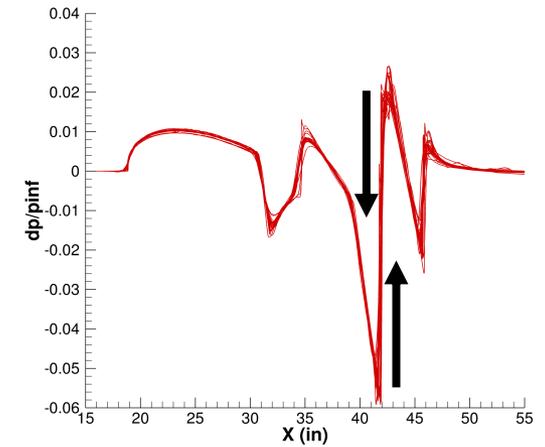
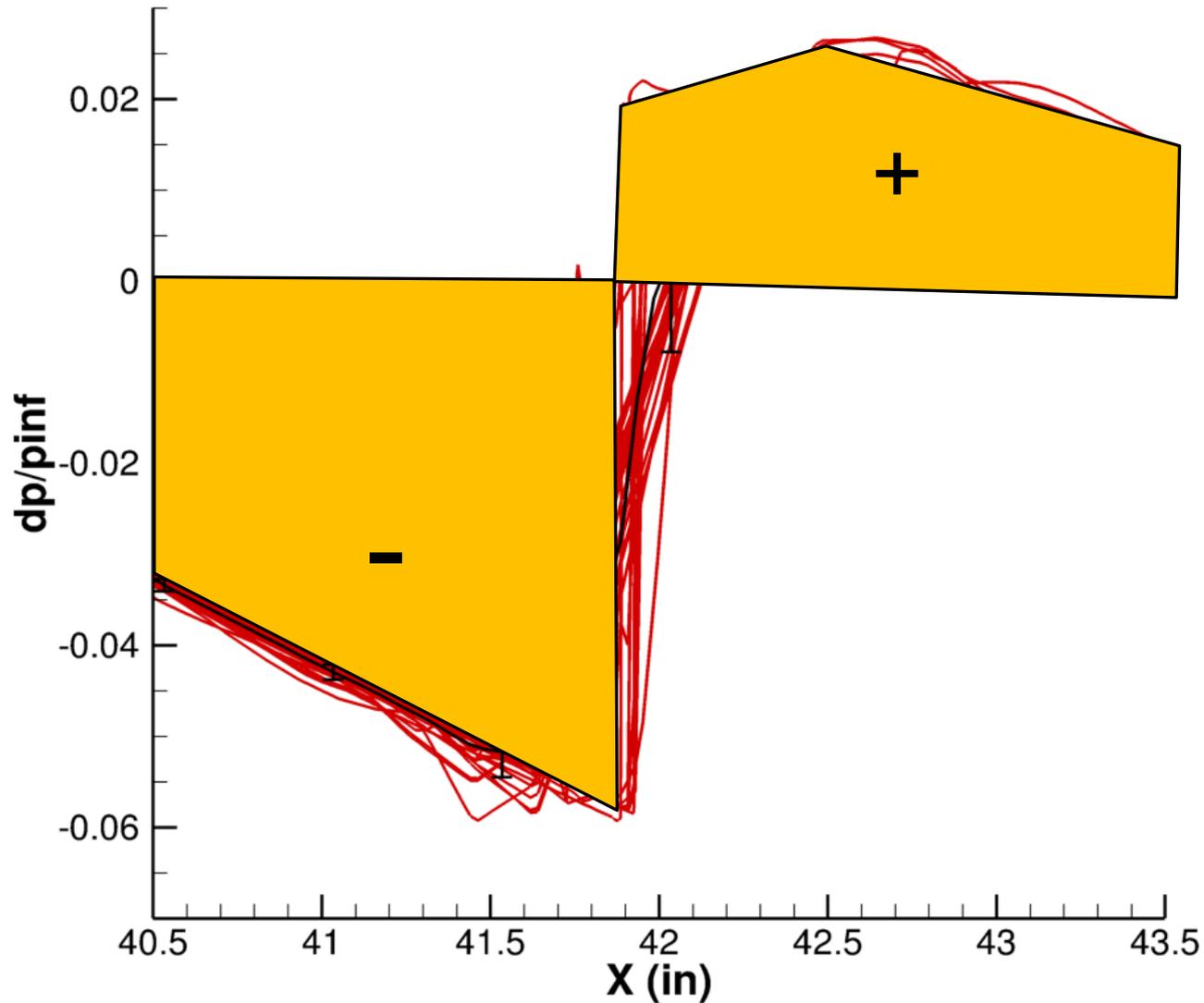
PHI=00

# BICONVEX LIP SHOCK



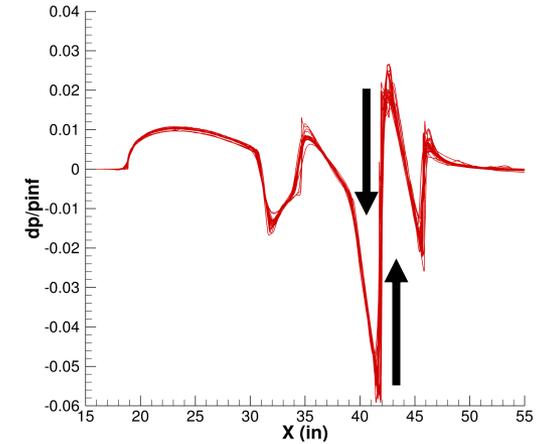
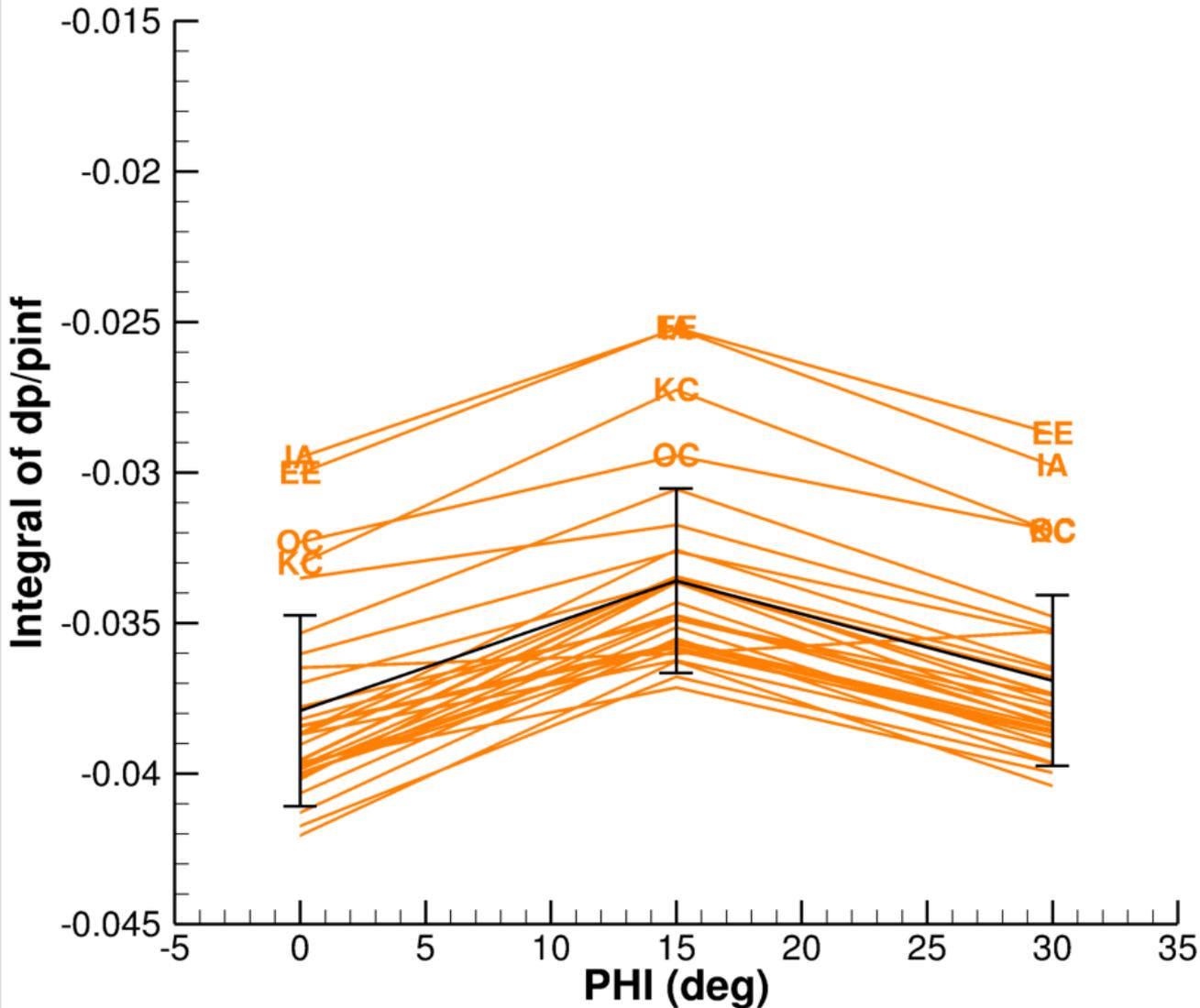
PHI=00

# BICONVEX LIP SHOCK PRESSURE INTEGRAL CALCULATION CARTOON



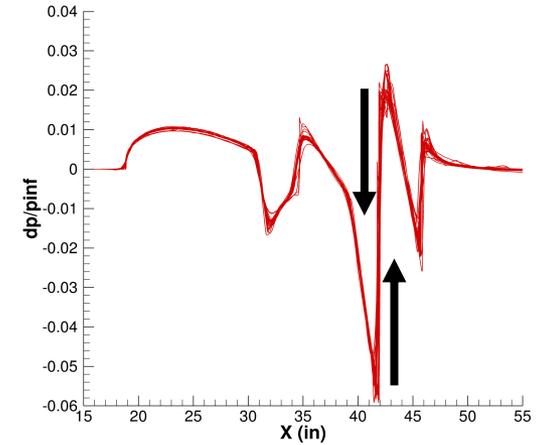
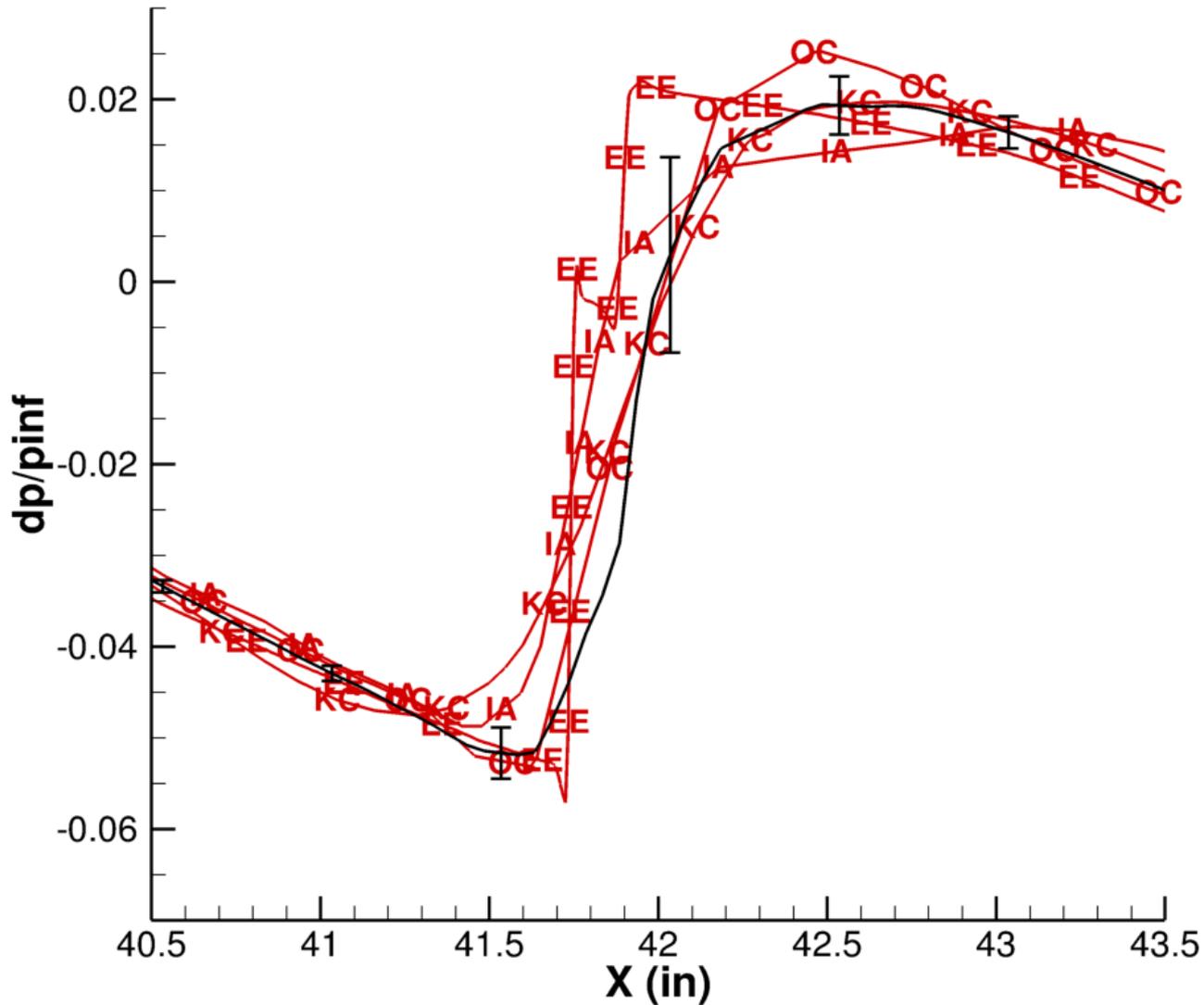
PHI=00

# BICONVEX LIP SHOCK INTEGRAL OUTLIERS



$\text{PHI}=00$

# BICONVEX LIP SHOCK INTEGRAL OUTLIERS



PHI=00

# BICONVEX SUMMARY

- Participant submissions tightly grouped and strict criteria required to identify outliers
- Experiment and participant submissions (displayed as mean and standard deviation) have low variation but do not overlap in multiple locations
  - Possible reference pressure drift in experiment
  - Test section reflection missing from simulation
  - RANS (and Euler) may provide an incomplete simulation of shock-plume interaction physics
  - Simplified boundary conditions may be insufficient

# BICONVEX SUMMARY

- Pressure magnitude used as identifier in smooth regions and pressure integral used as identifier in nonsmooth region with shock
- Many outliers were identified in multiple regions: forebody, lip shock, and plume
  - Forebody outliers were equally high and low
  - Lip shock outliers had forward shock location
  - Plume outliers were high, except one

C608, AN  
EARLY X-59  
PROTOTYPE,  
SUBMISSIONS

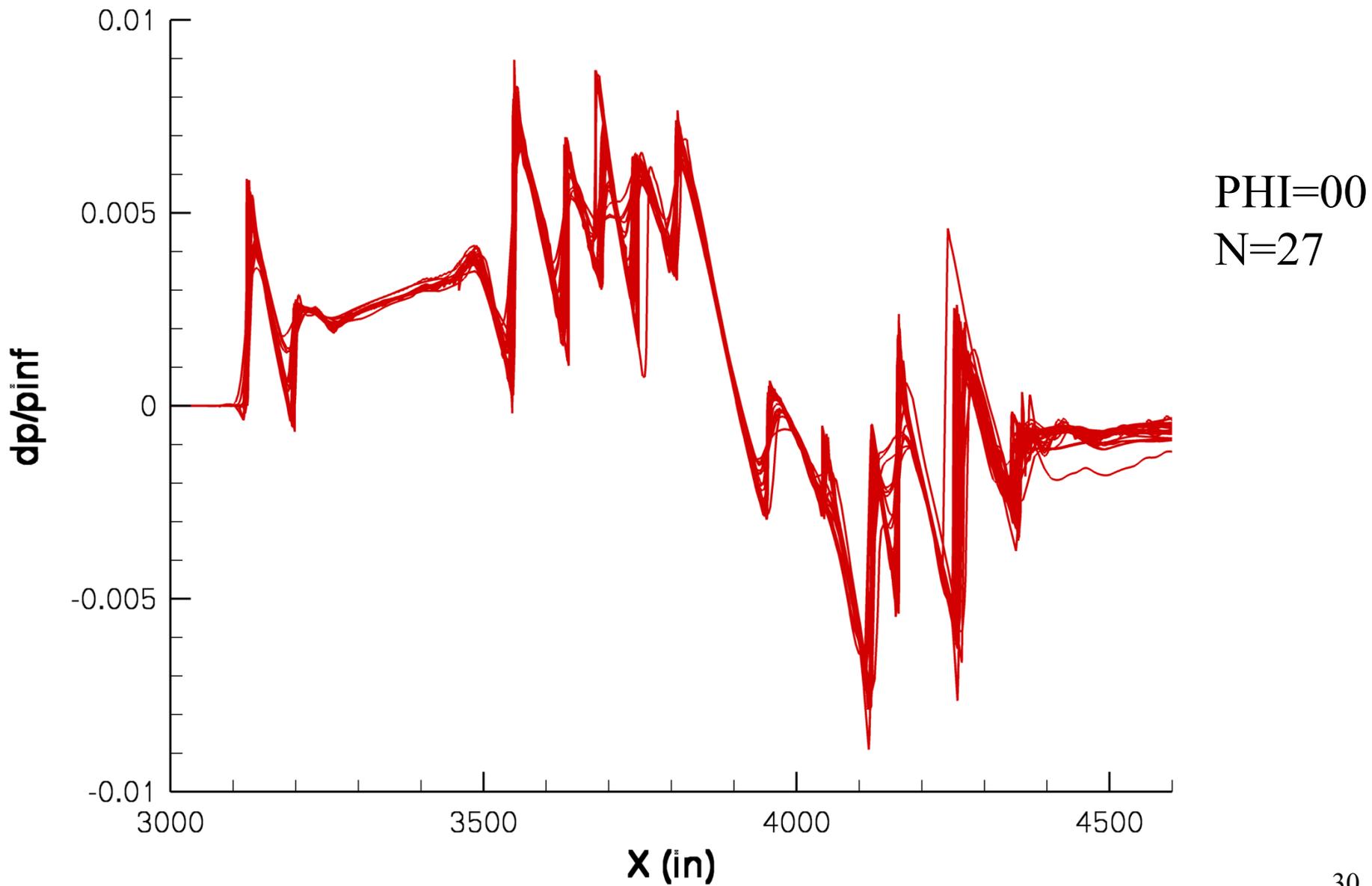
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# OUTLINE

- Near field statistics
- Boom carpets
- Grid convergence
- Pointwise standard deviation of boom carpet
  - Identify outliers
- Details on outliers
  - Nearfield
  - Ground
- Compare to previous workshop for context

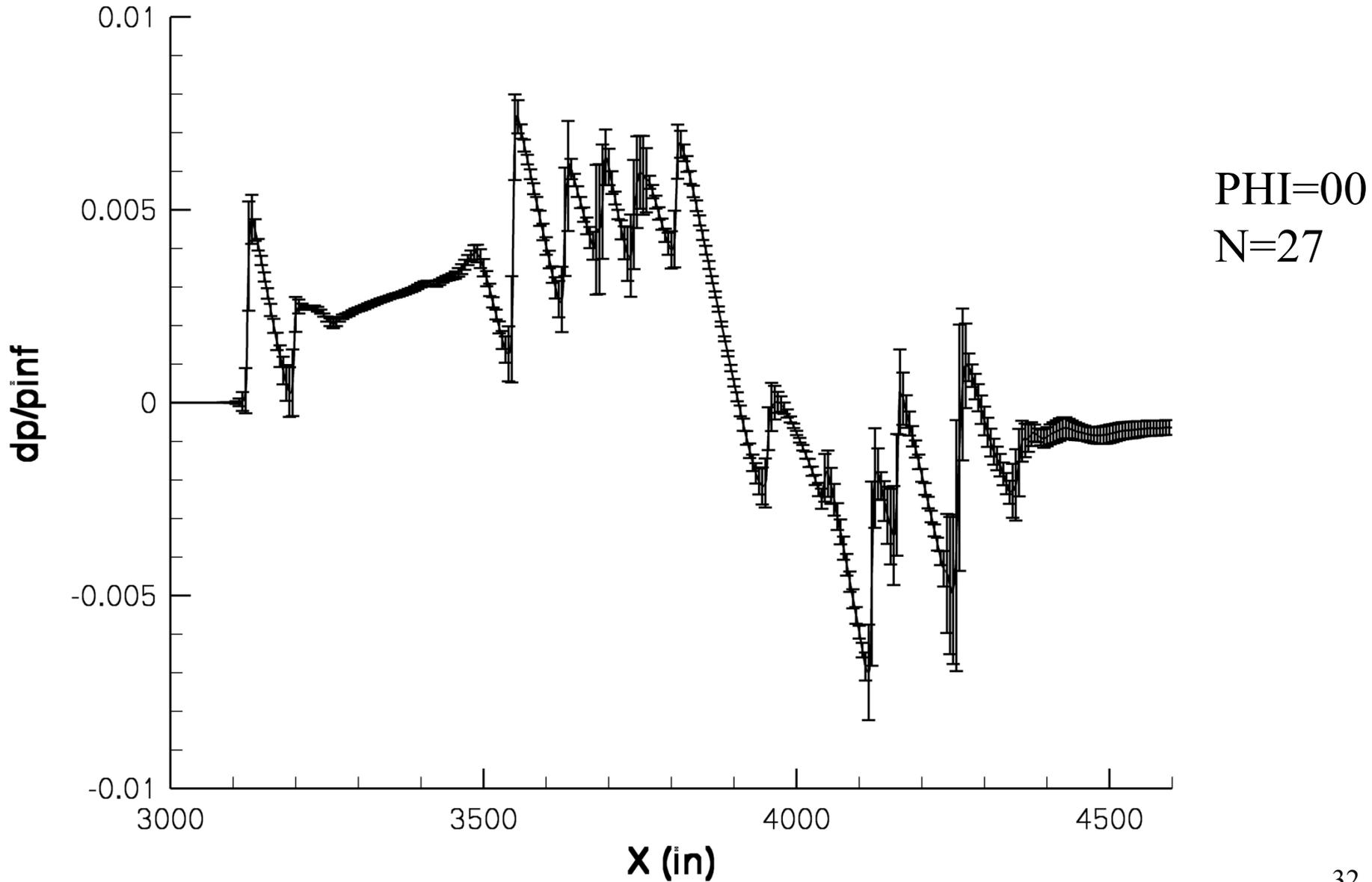
# C608 FINE-GRID ENSEMBLE



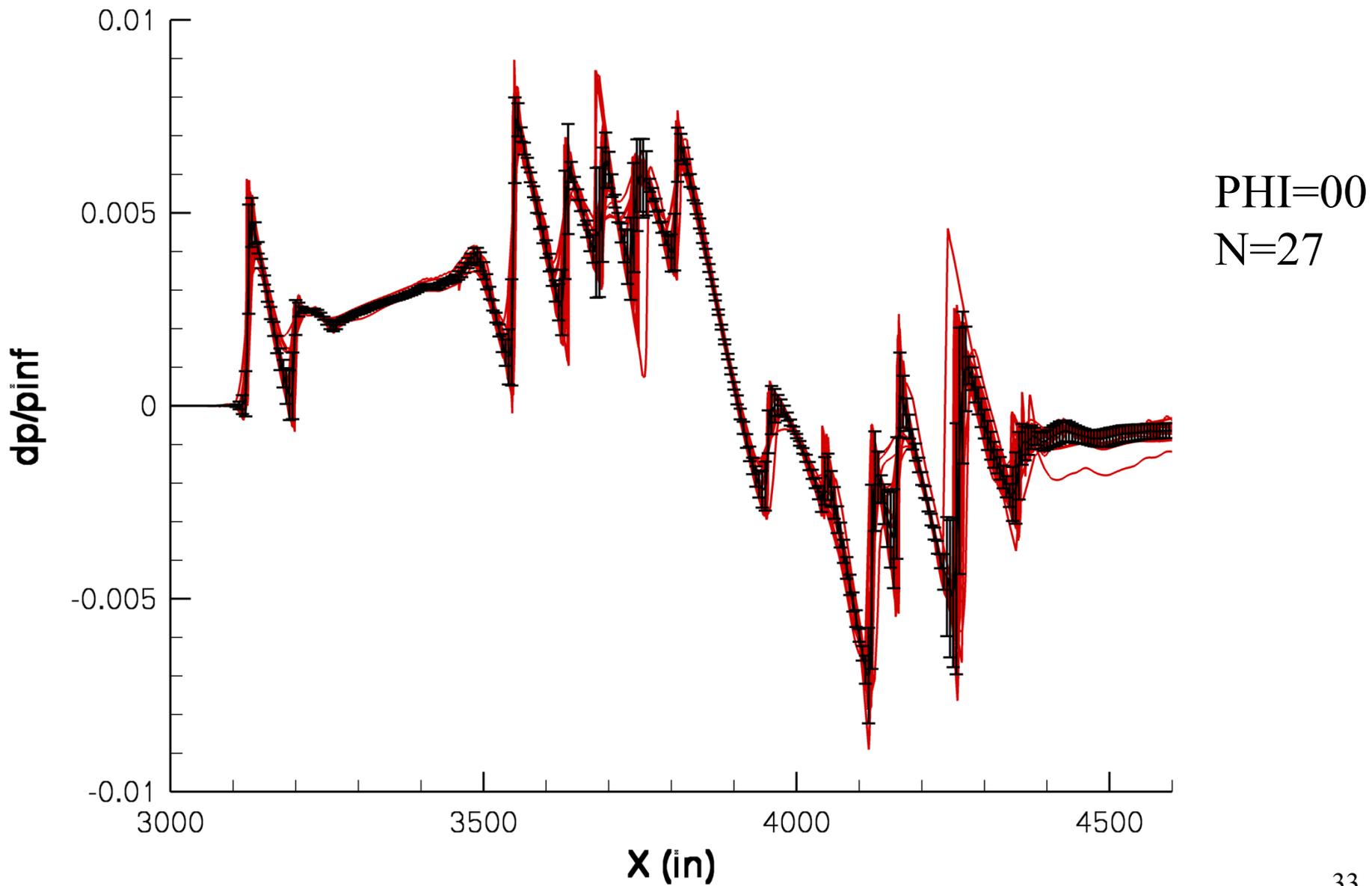
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- Pointwise population mean and standard deviation of interpolated signature every half inch
- Analogous to wind tunnel spatial averaging
- Finest grid solution from each participant (which vary in resolution)
- Outliers impact these statistics

# C608 FINE-GRID ENSEMBLE



# C608 FINE-GRID ENSEMBLE



# GROUND PROPAGATION

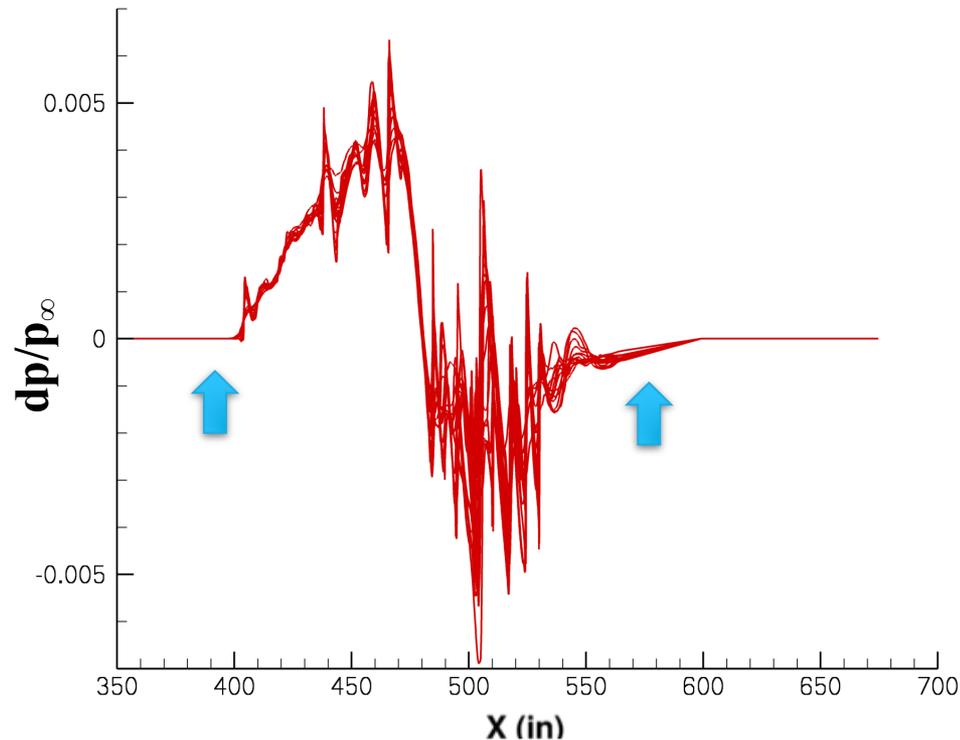
- Geometry and grids provided in “full-scale”
- US Standard atmosphere and ANSI S1.26 Annex C relative humidity from 53200 ft. altitude
- sBOOM version 2.82 (Rallabhandi)
  - Burgers’ equation with molecular relaxation
- Submissions are windowed with fore and aft ramps

# NEARFIELD WINDOW FOR PROPAGATION

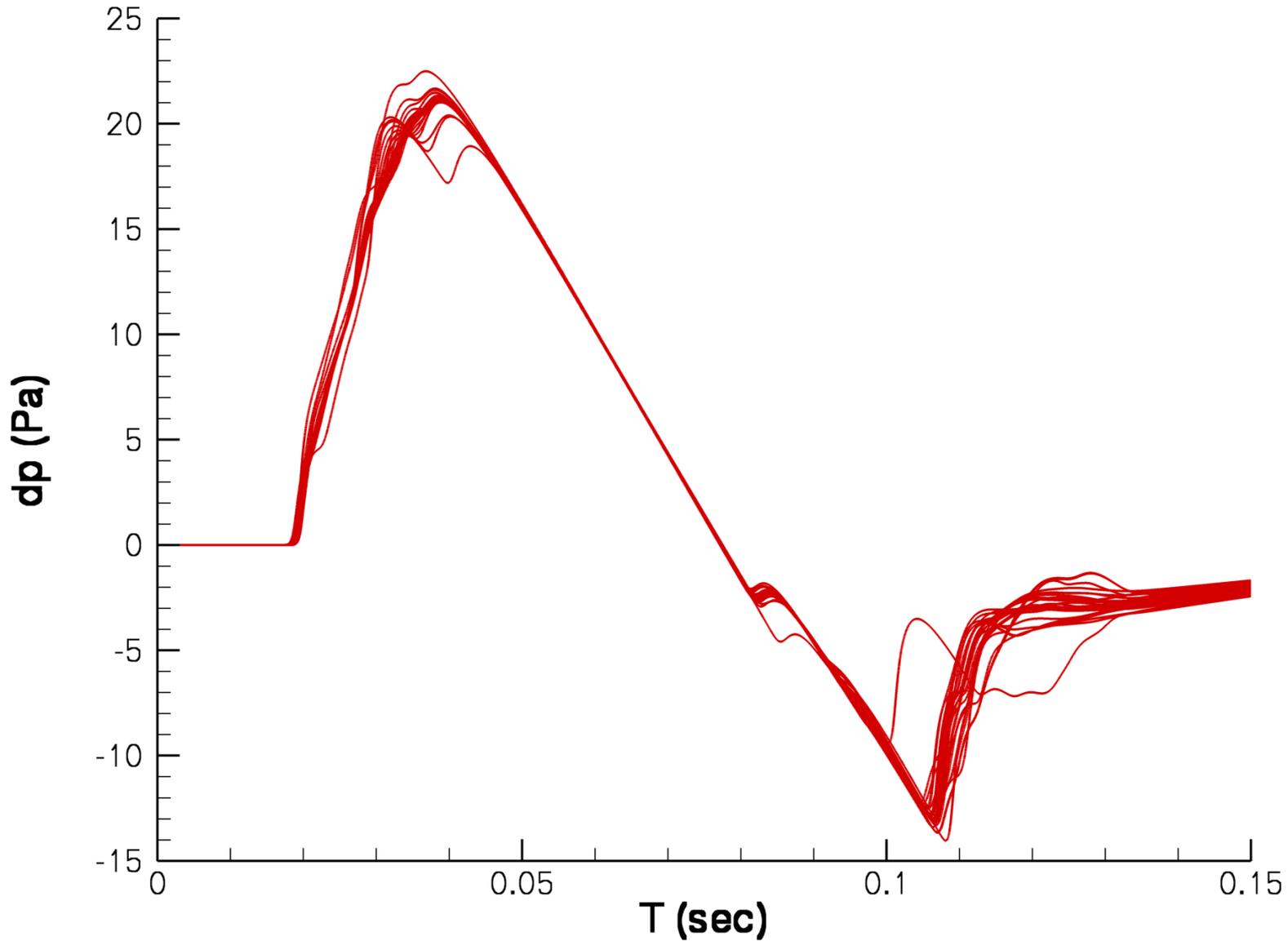
Nearfield submission is ramped to zero

- [2960,3070] ahead of signature
- [4790,5870] aft of signature

Signature is zero-padded outside of that range

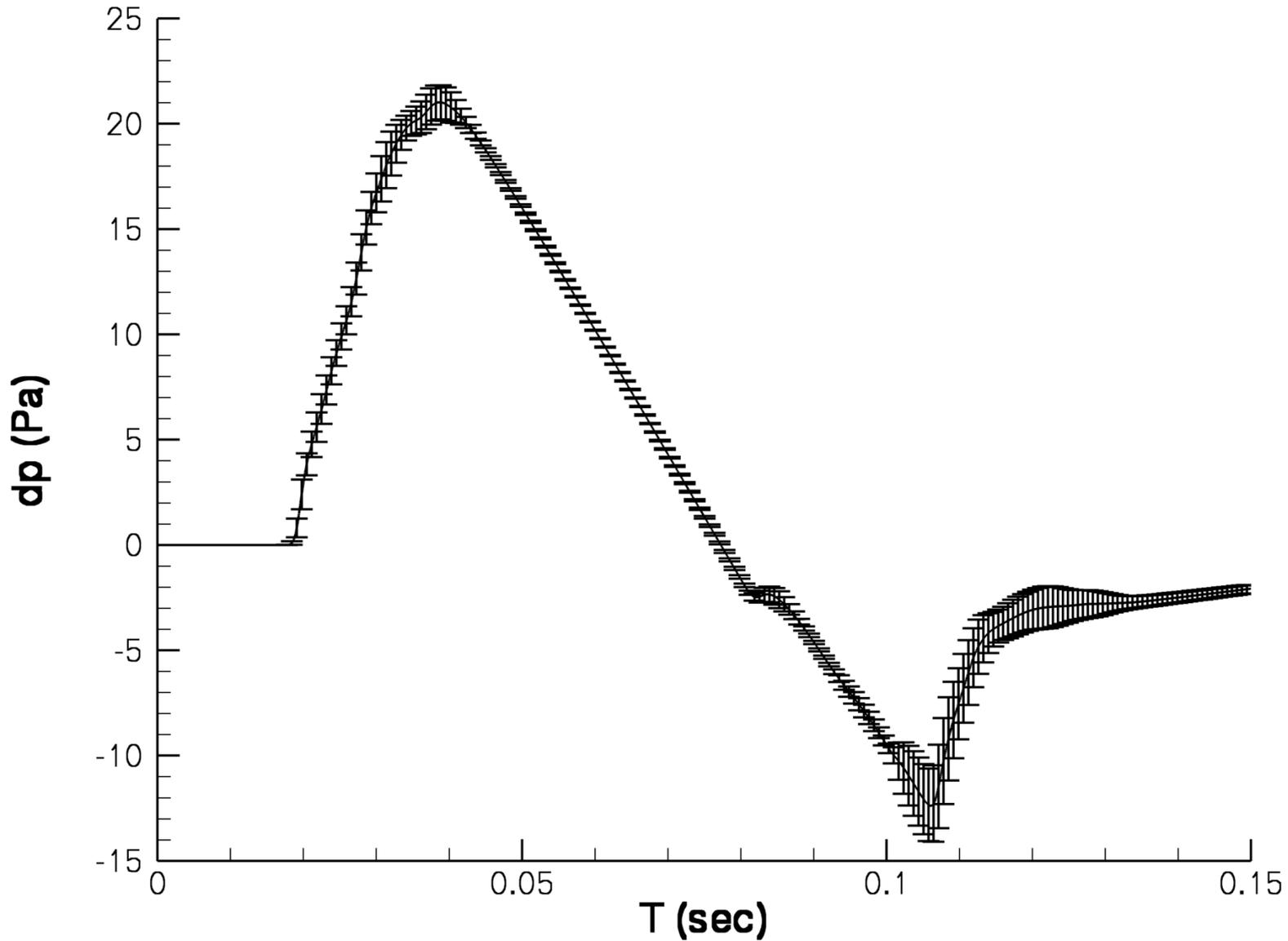


# C608 FINE-GRID GROUND



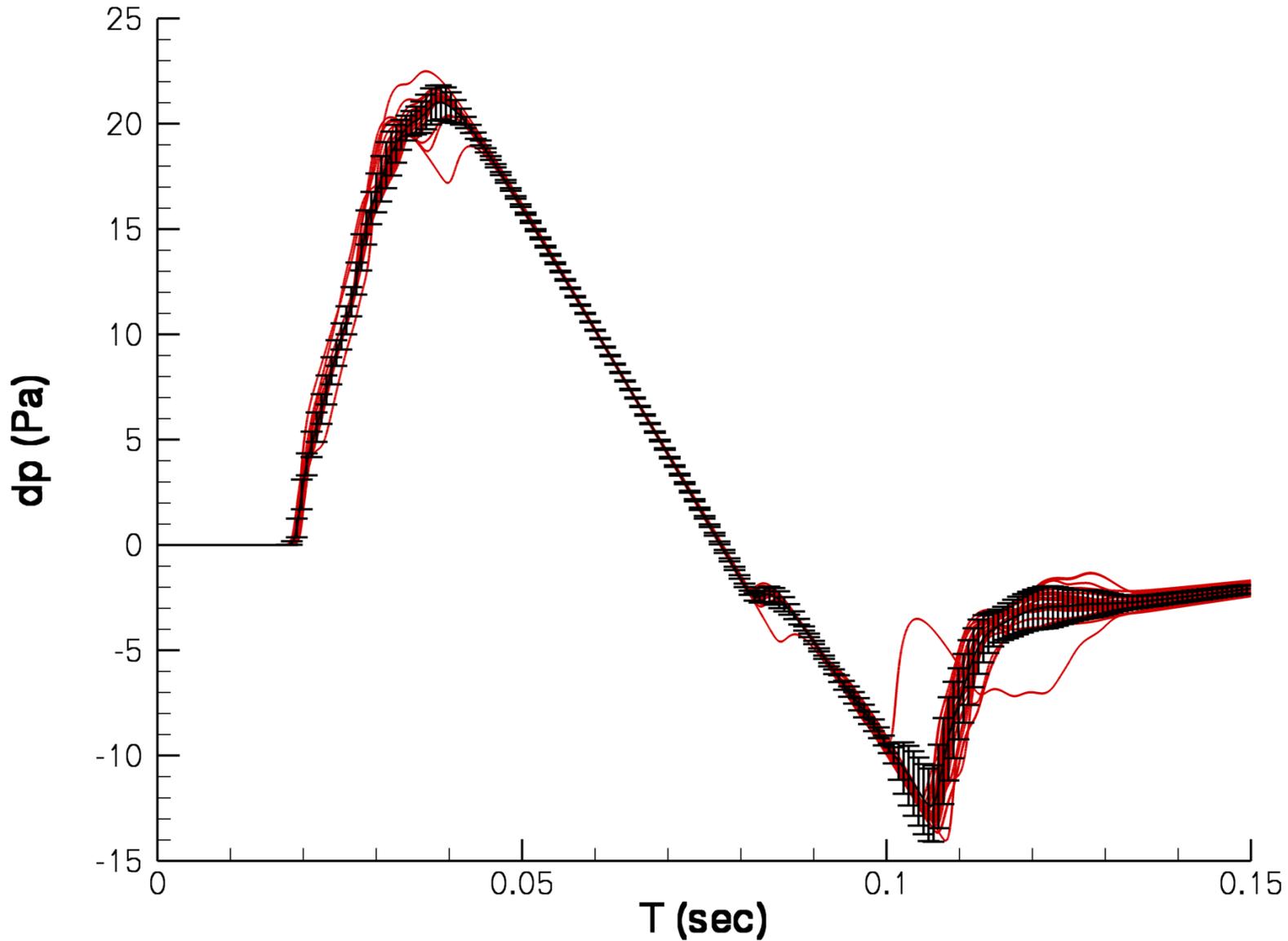
PHI=00  
N=27

# C608 FINE-GRID GROUND



PHI=00  
N=27

# C608 FINE-GRID GROUND



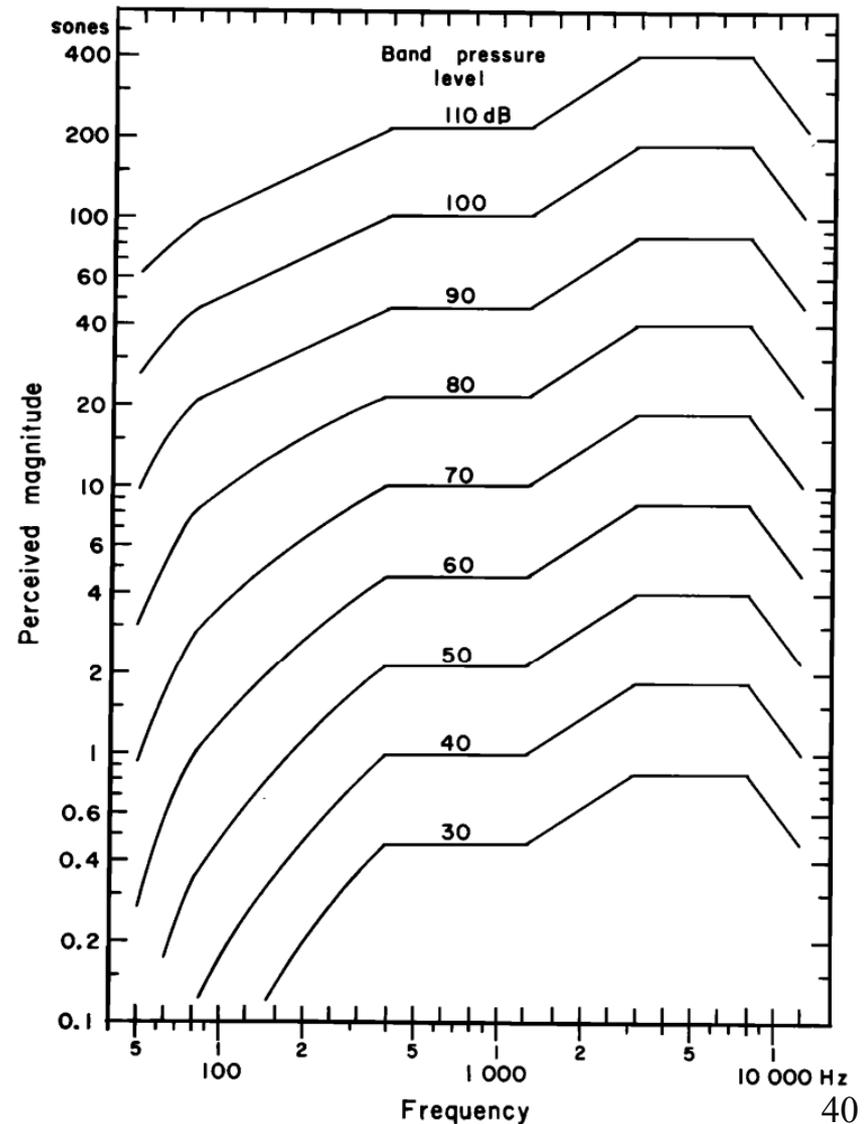
PHI=00  
N=27

# LOUDNESS AND ANNOYANCE

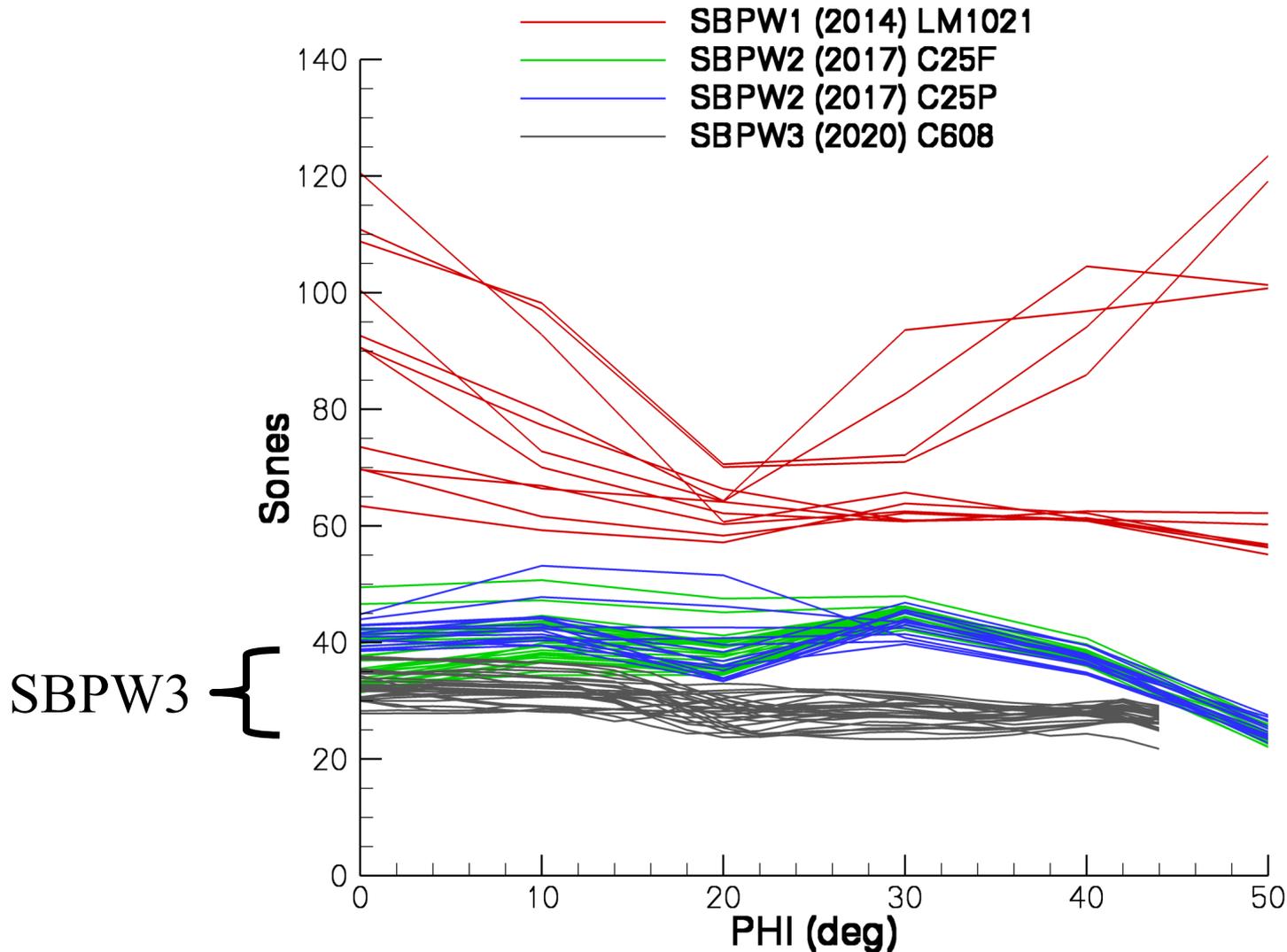
- Subjective metrics
- These human experiences are correlated to noise descriptors through experiments
  - Leatherwood et al. JASA 2002
  - Stevens Mark VII Perceived Level (PL)
  - Loubeau et al. 2nd International Sonic Boom Forum 2015 meta-study

# PERCEIVED LEVEL (PL)

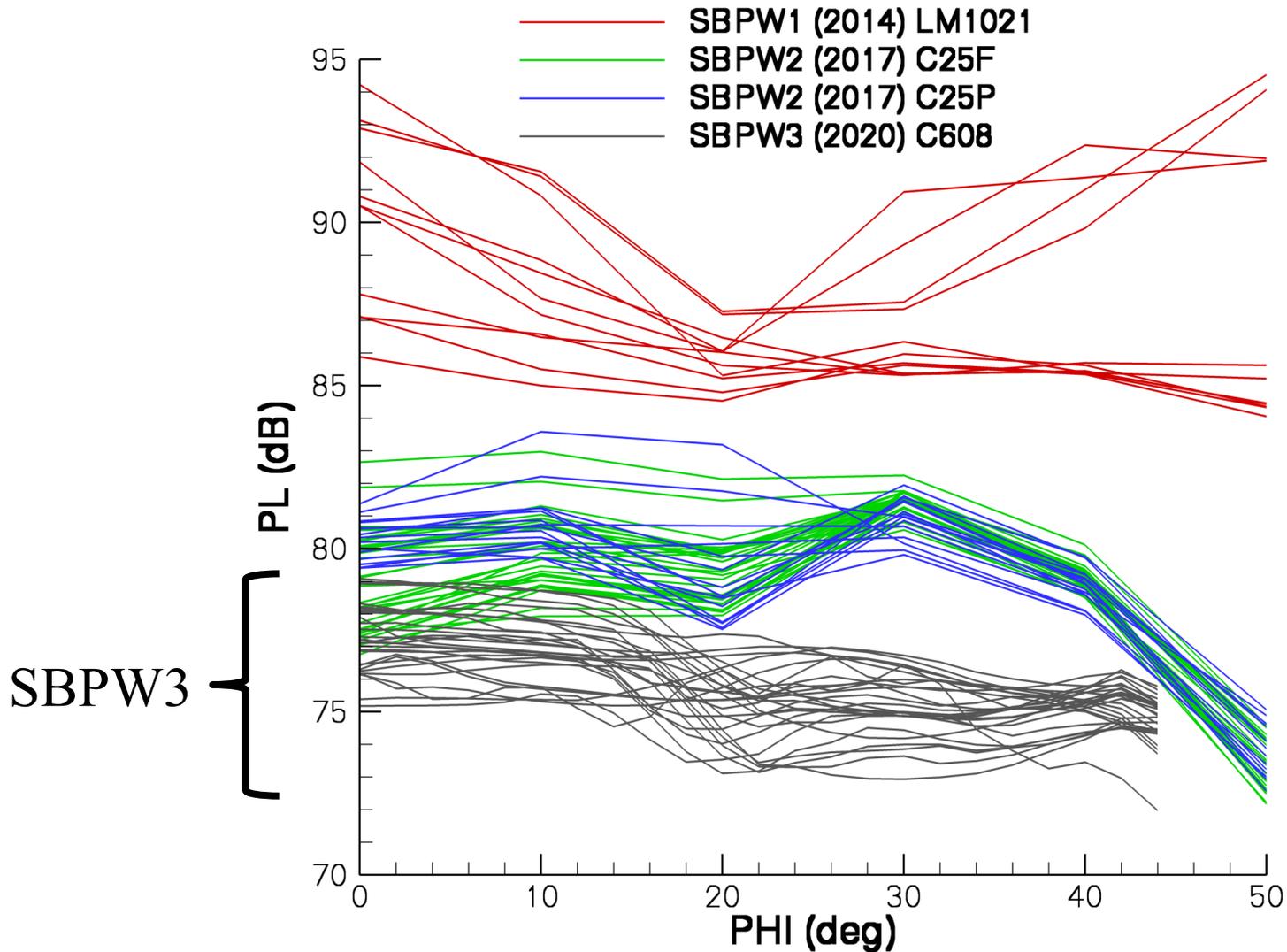
- Signature sound pressure level is gathered into 1/3 octave bands
- Band levels are converted from db into sones (based on subjects perceived loudness)
- Sones from each band are combined
- Sones are converted into PL via logarithm



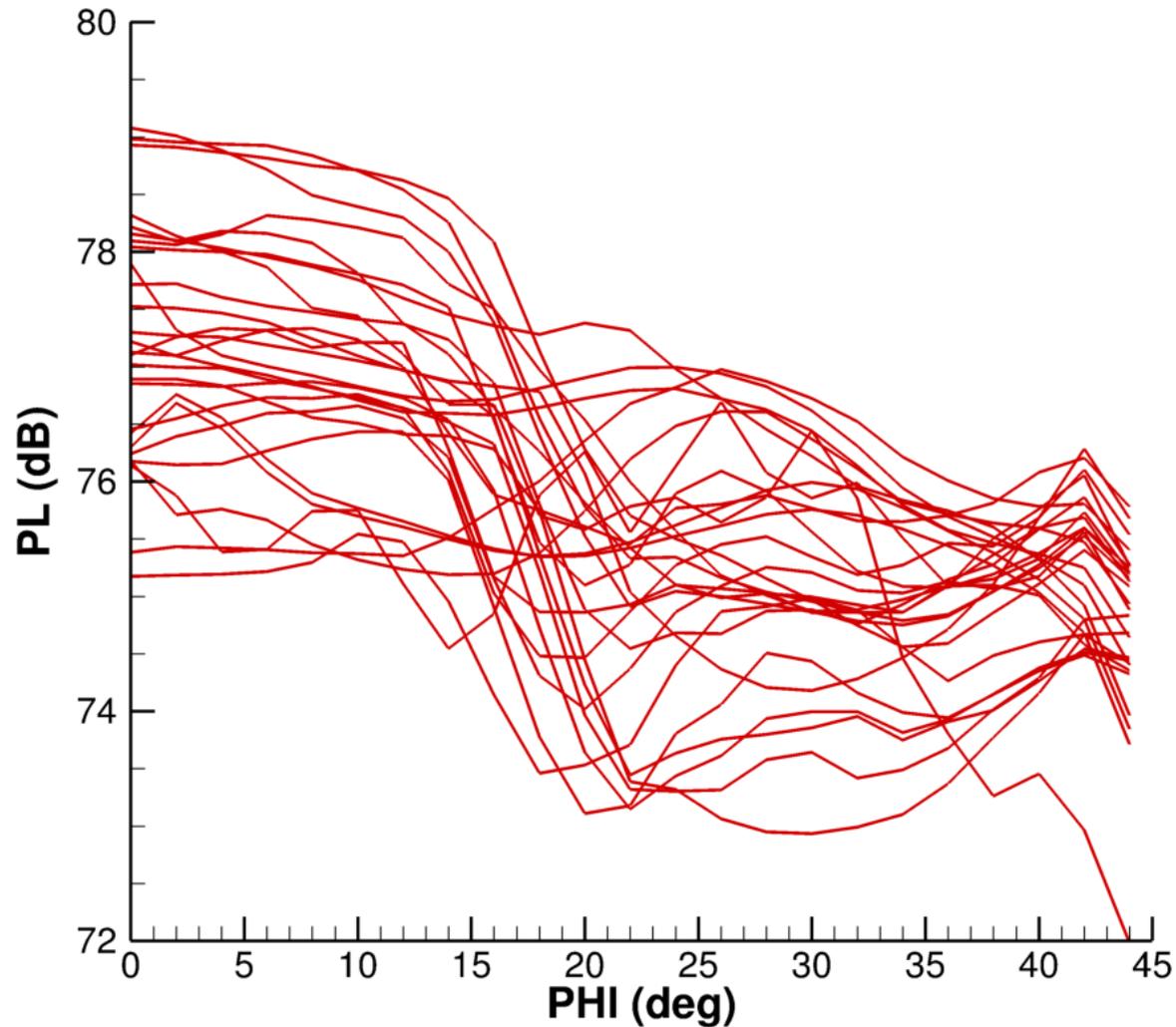
# SBPW FINE-GRID LOUDNESS (SONES) CARPET



# SBPW FINE-GRID PL CARPET

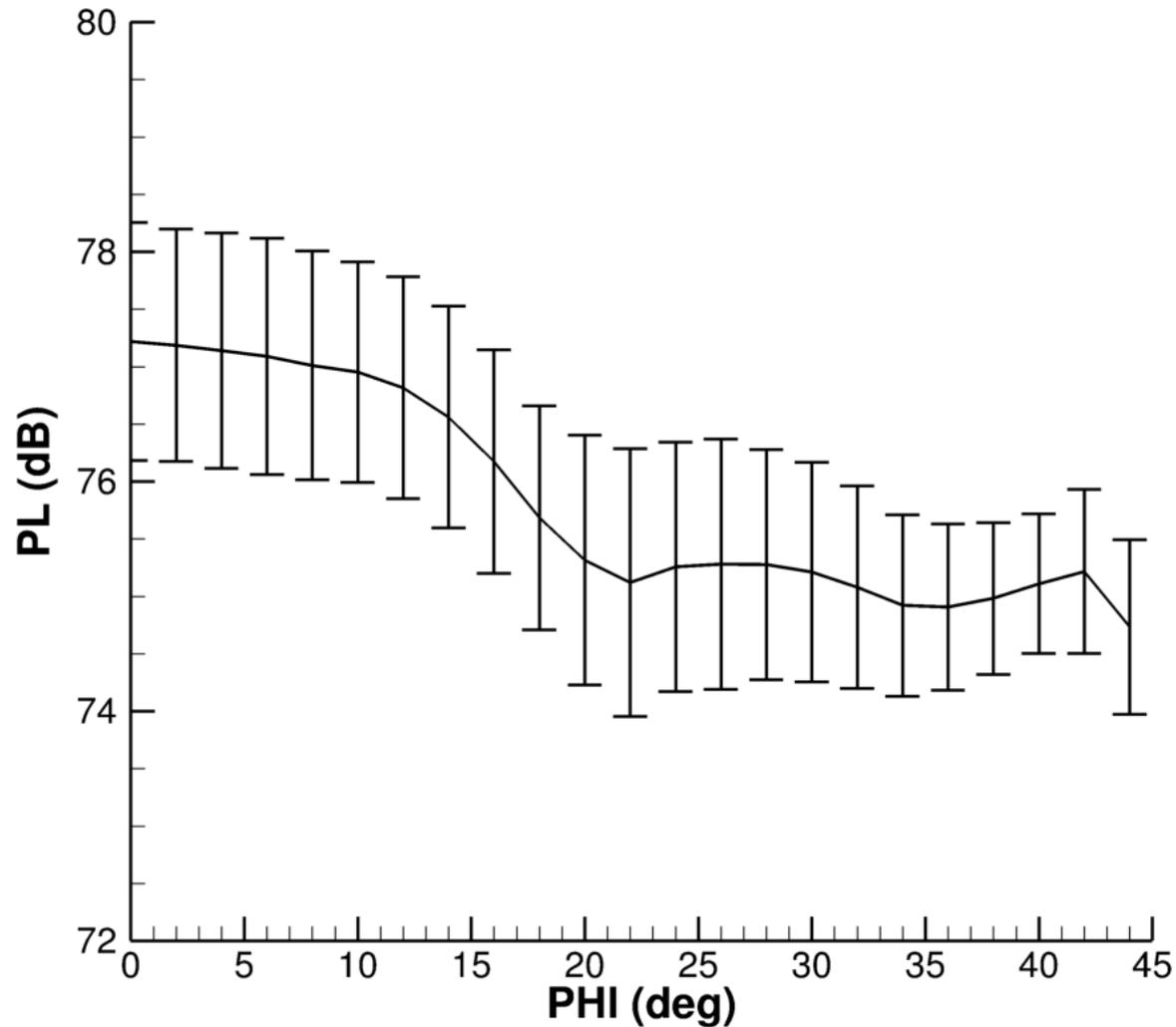


# C608 FINE-GRID PL CARPET



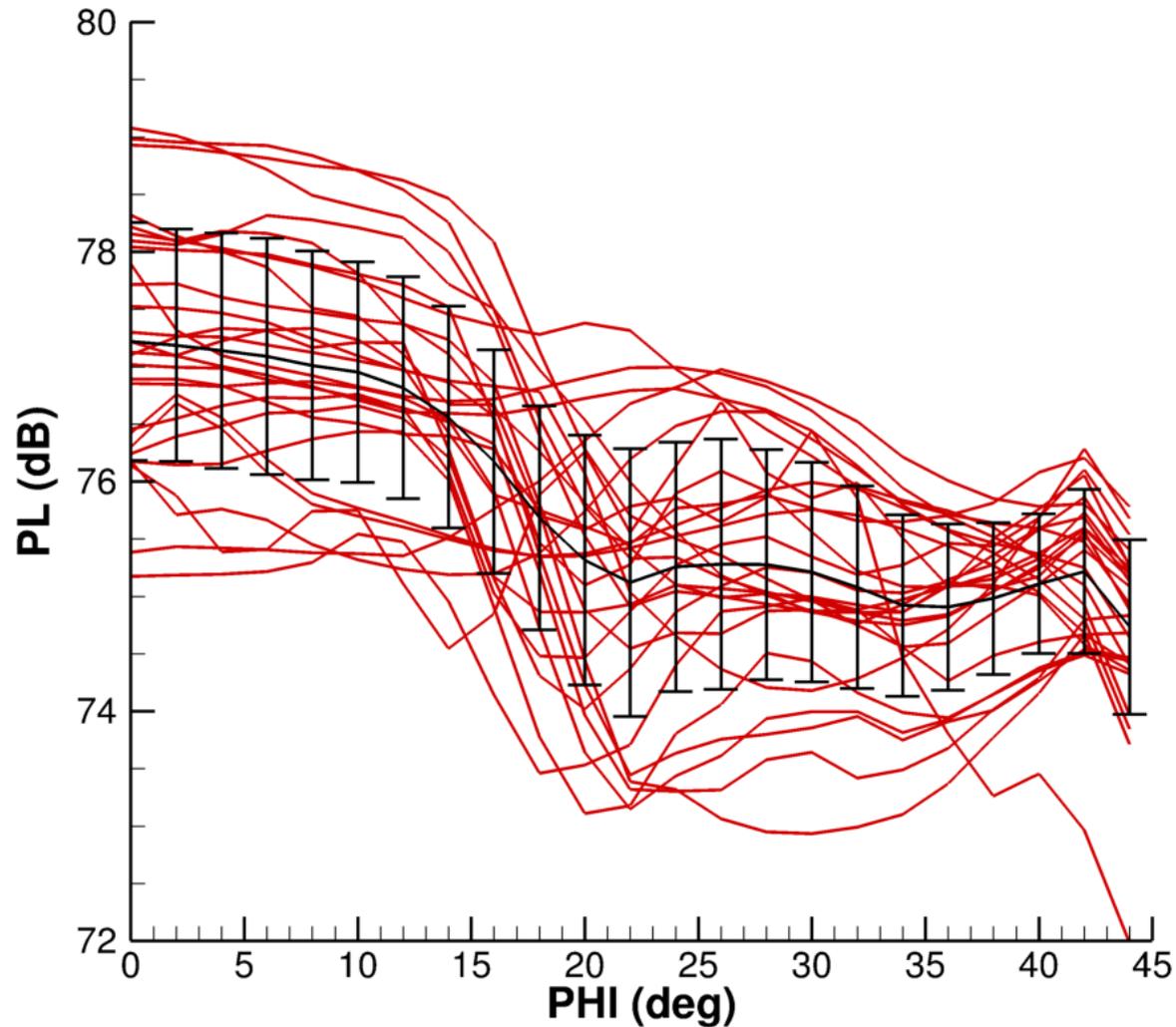
N=27

# C608 FINE-GRID PL CARPET



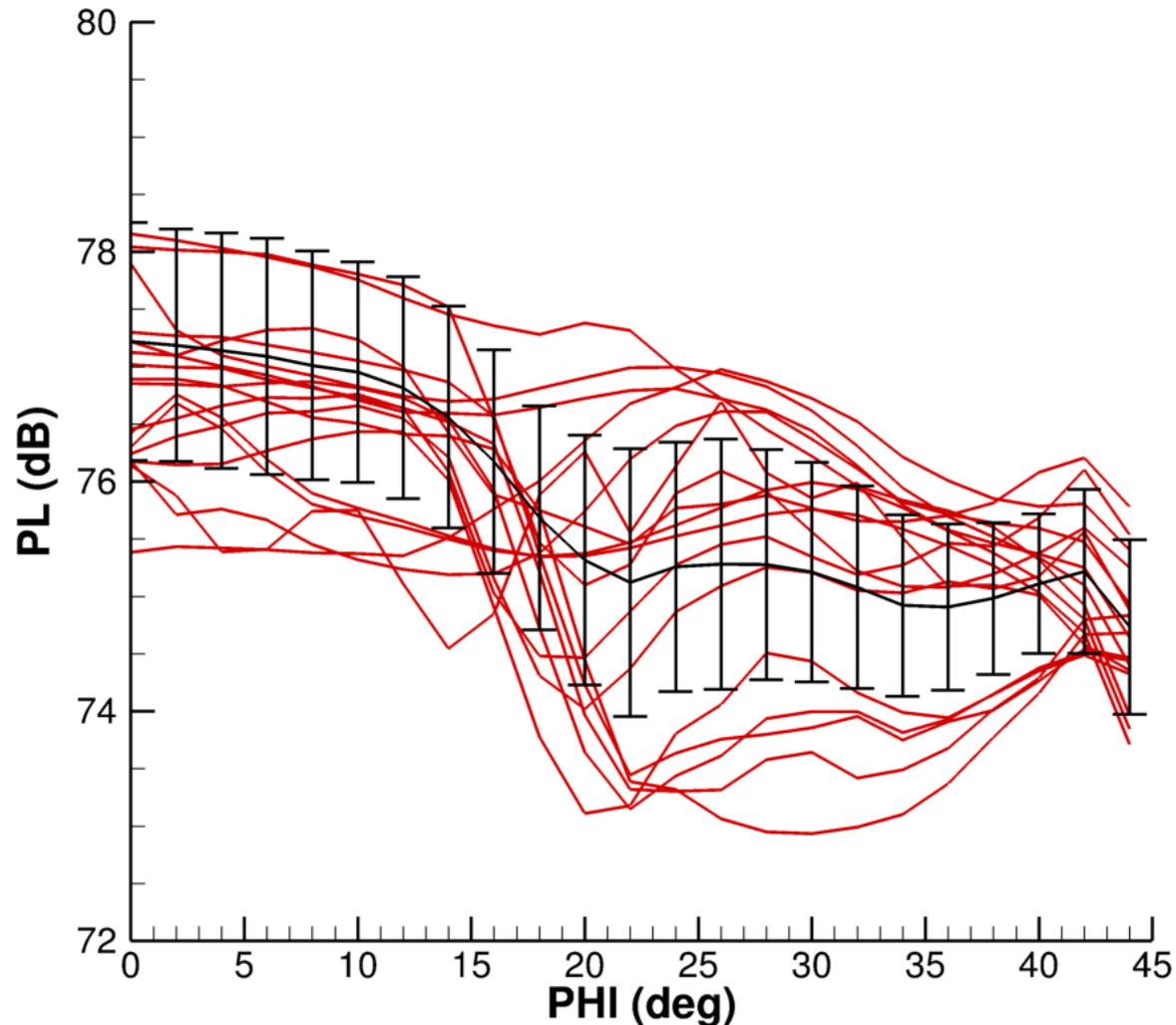
N=27

# C608 FINE-GRID PL CARPET



N=27

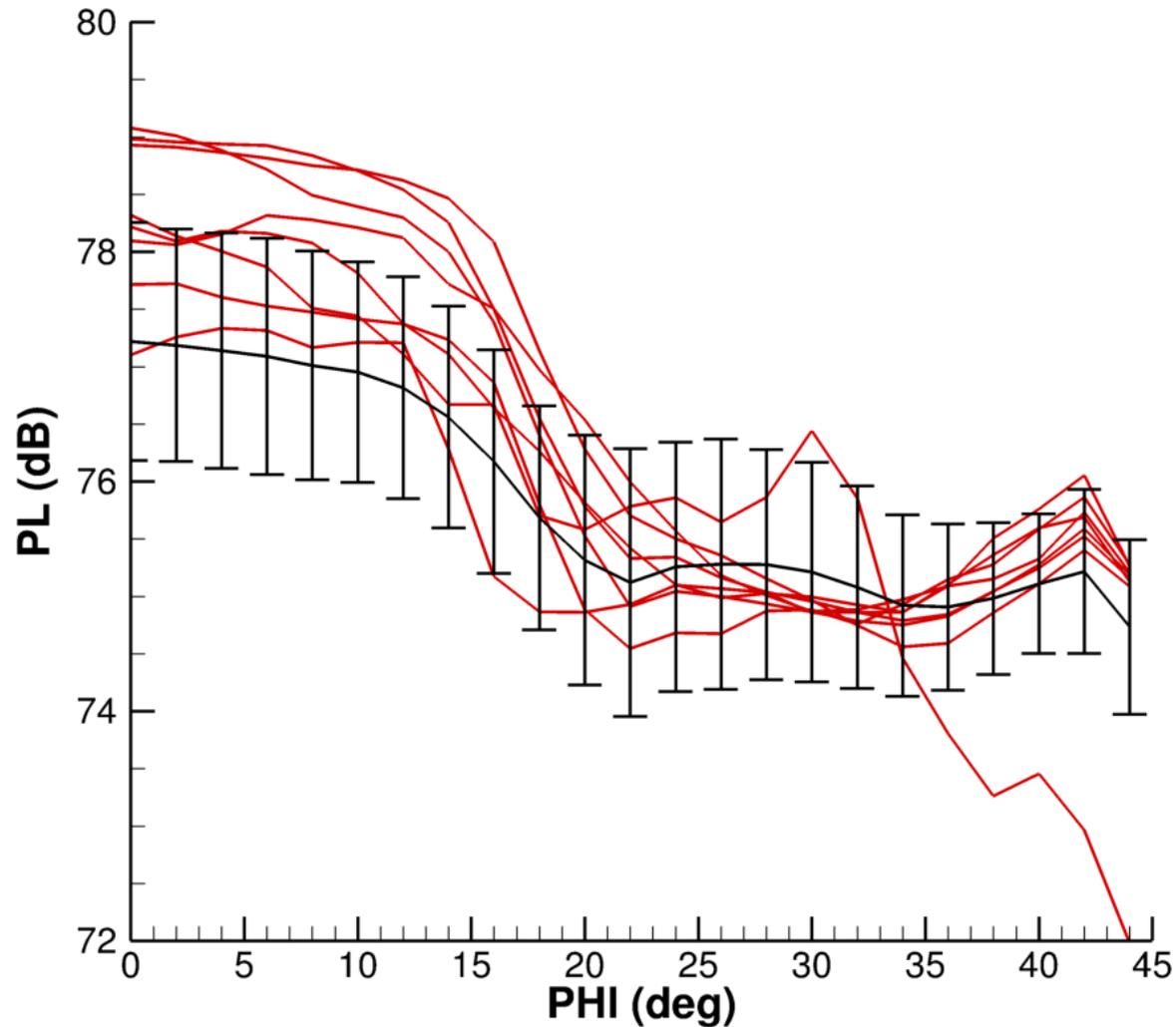
# C608 WORKSHOP PROVIDED FINE-GRID PL



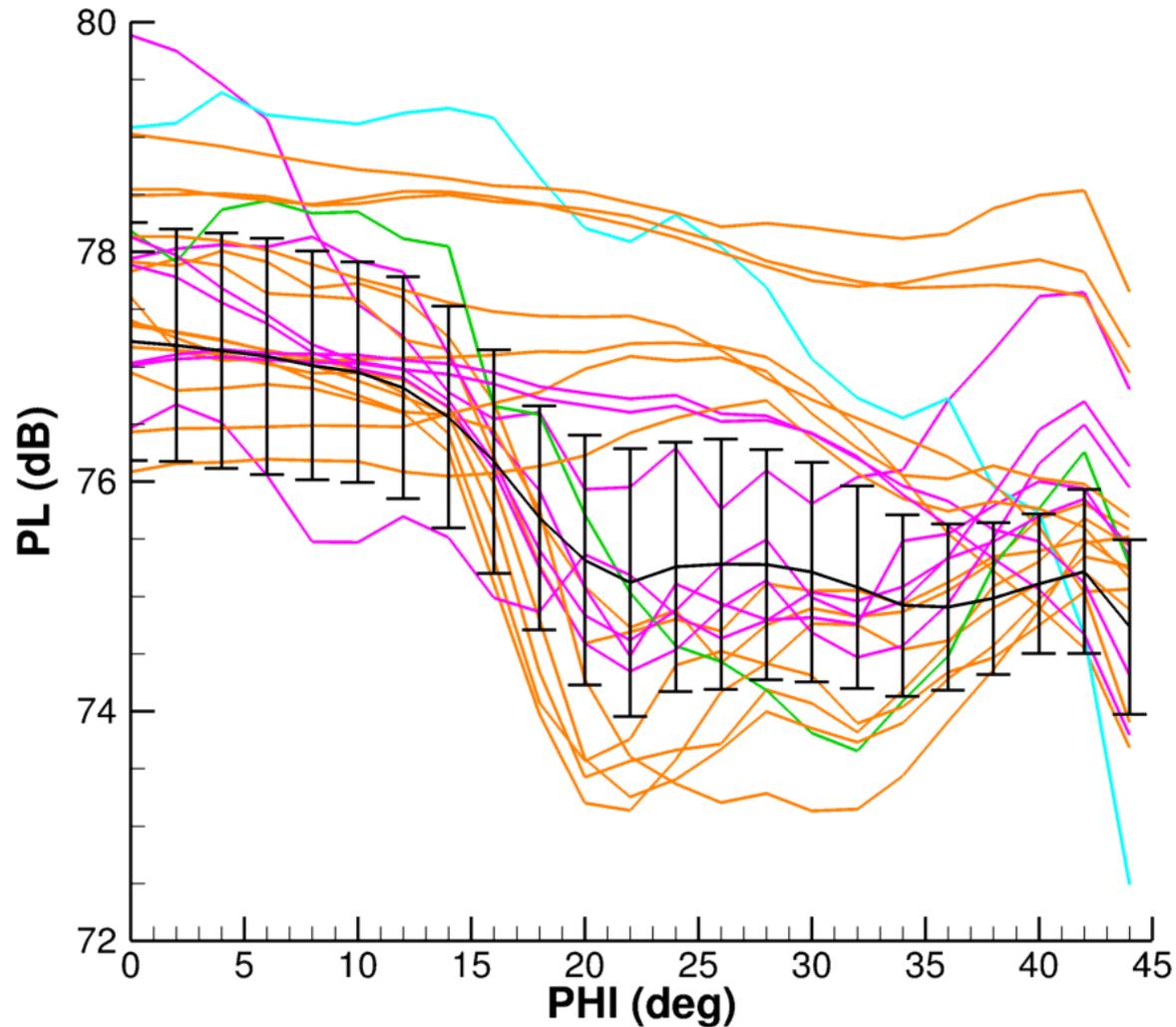
N=19

# C608 PARTICIPANT GENERATED FINE-GRID PL

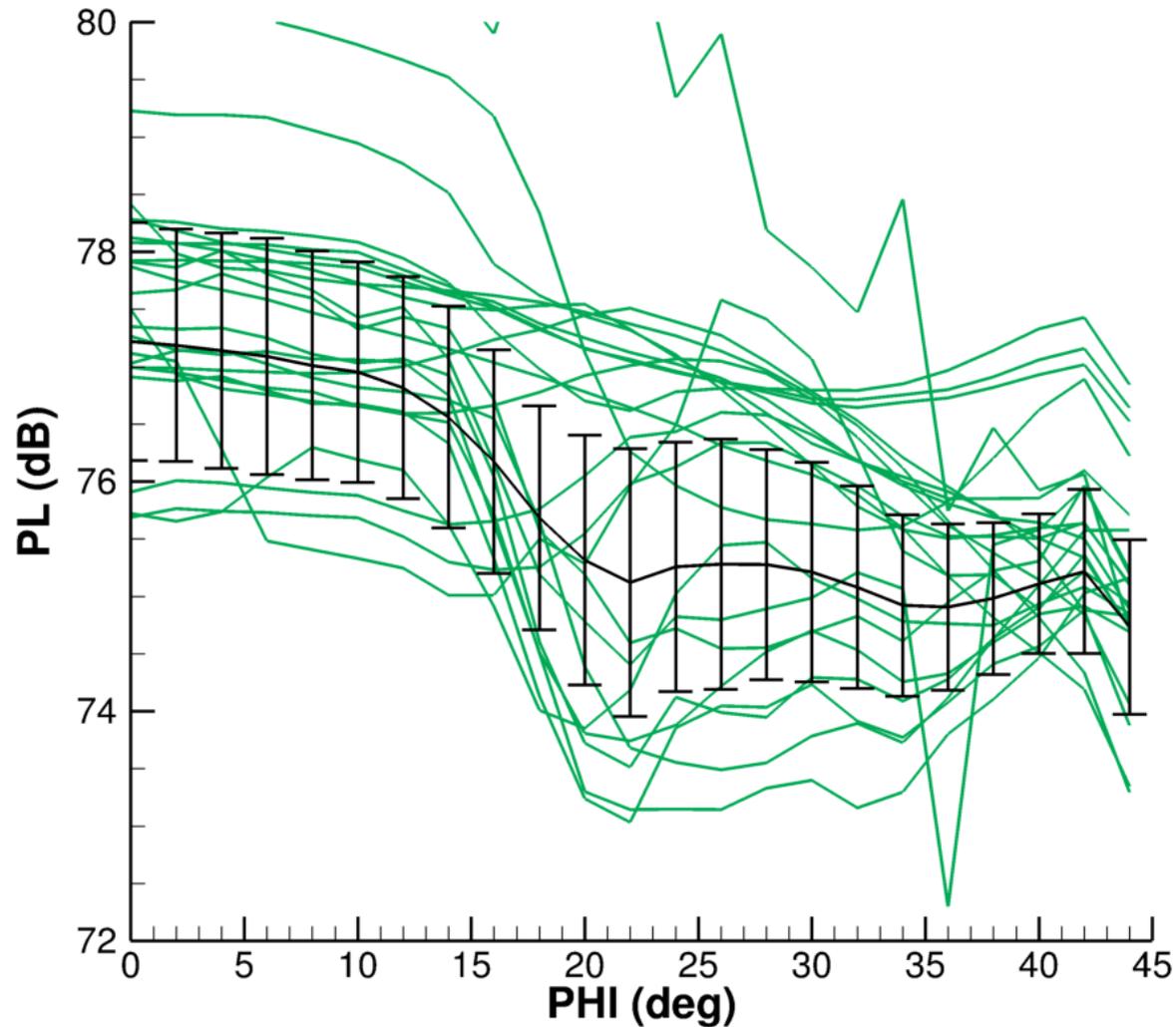
N=8



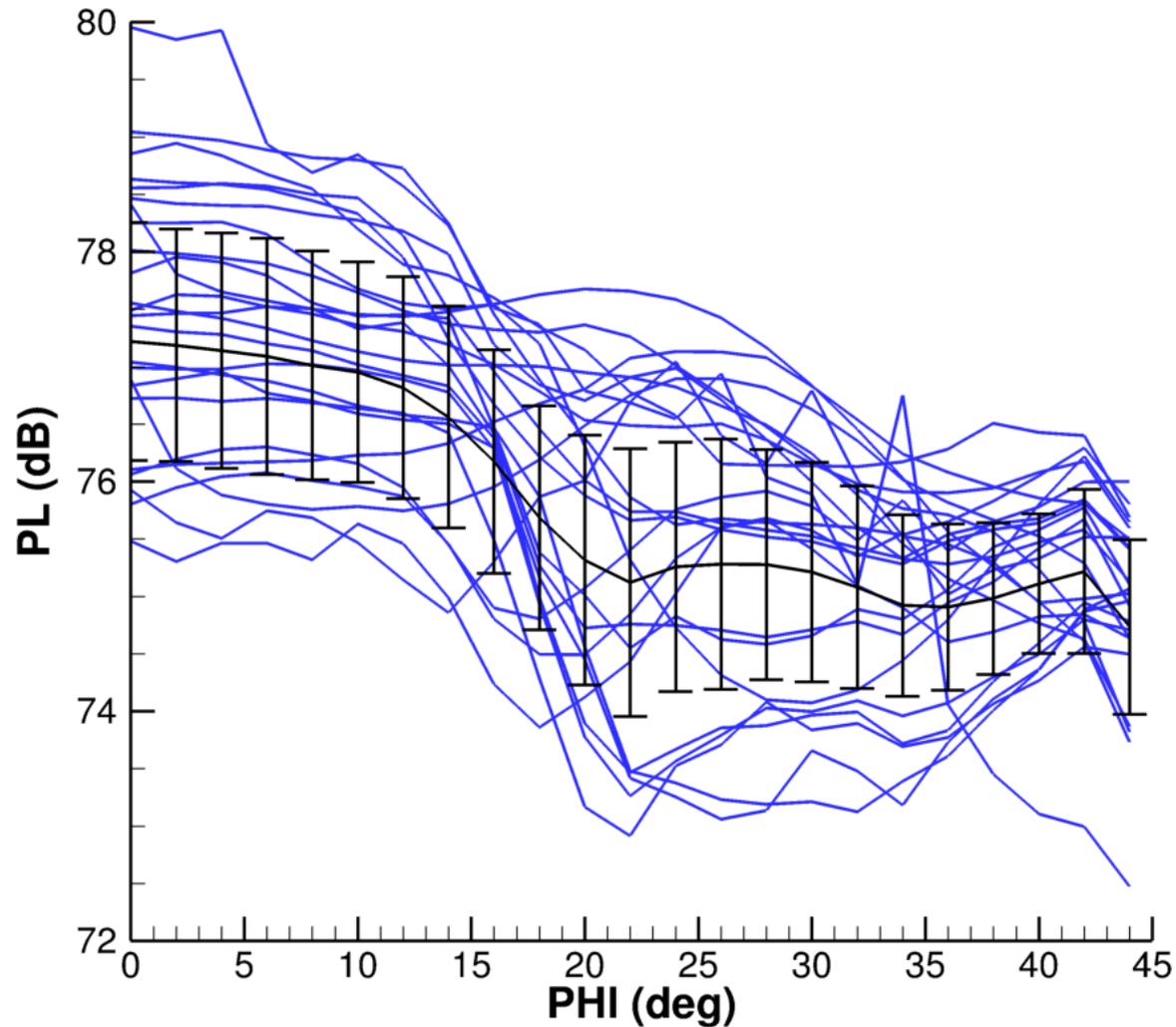
# C608 TINY-GRID PL



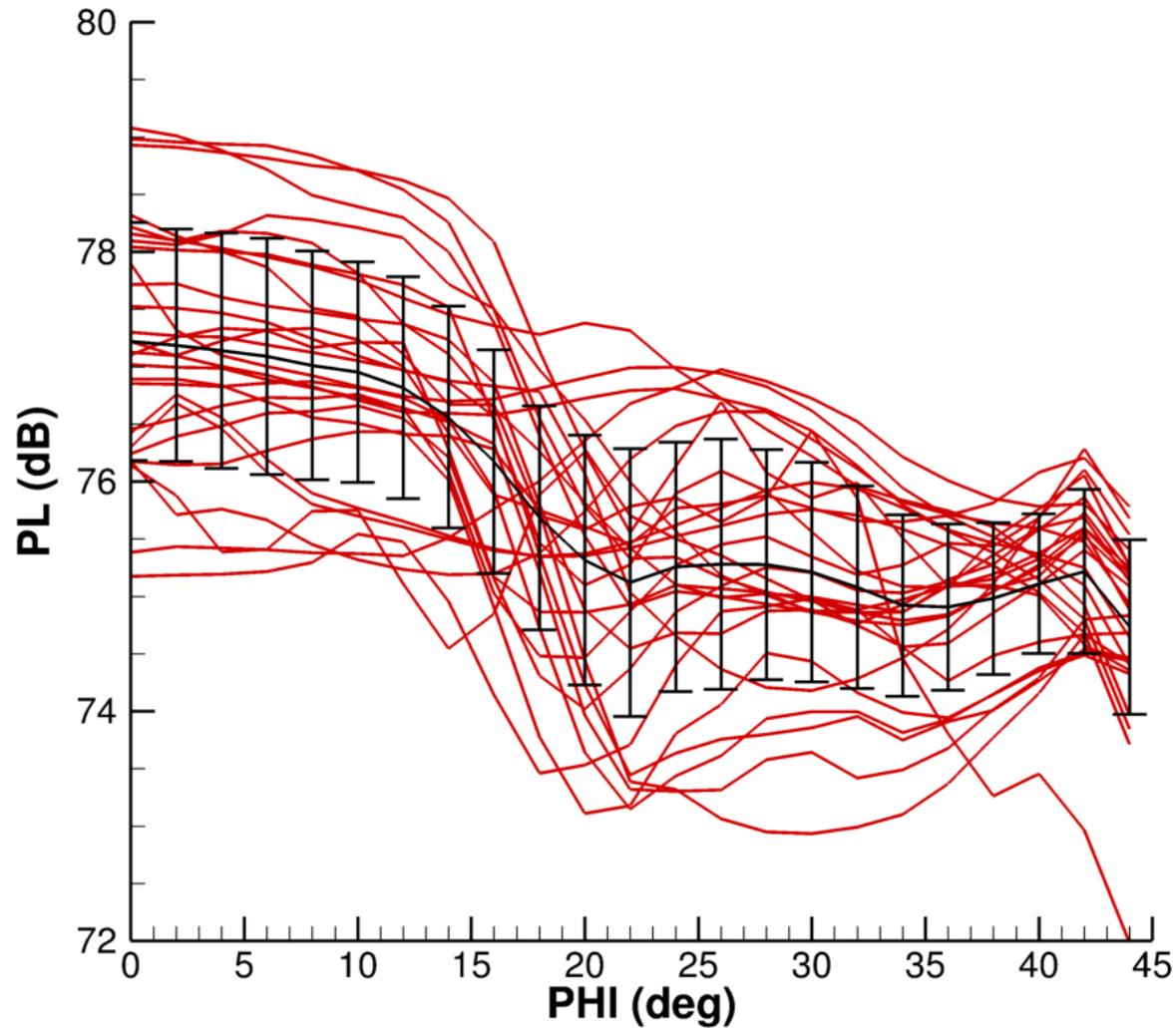
# C608 COARSE-GRID PL



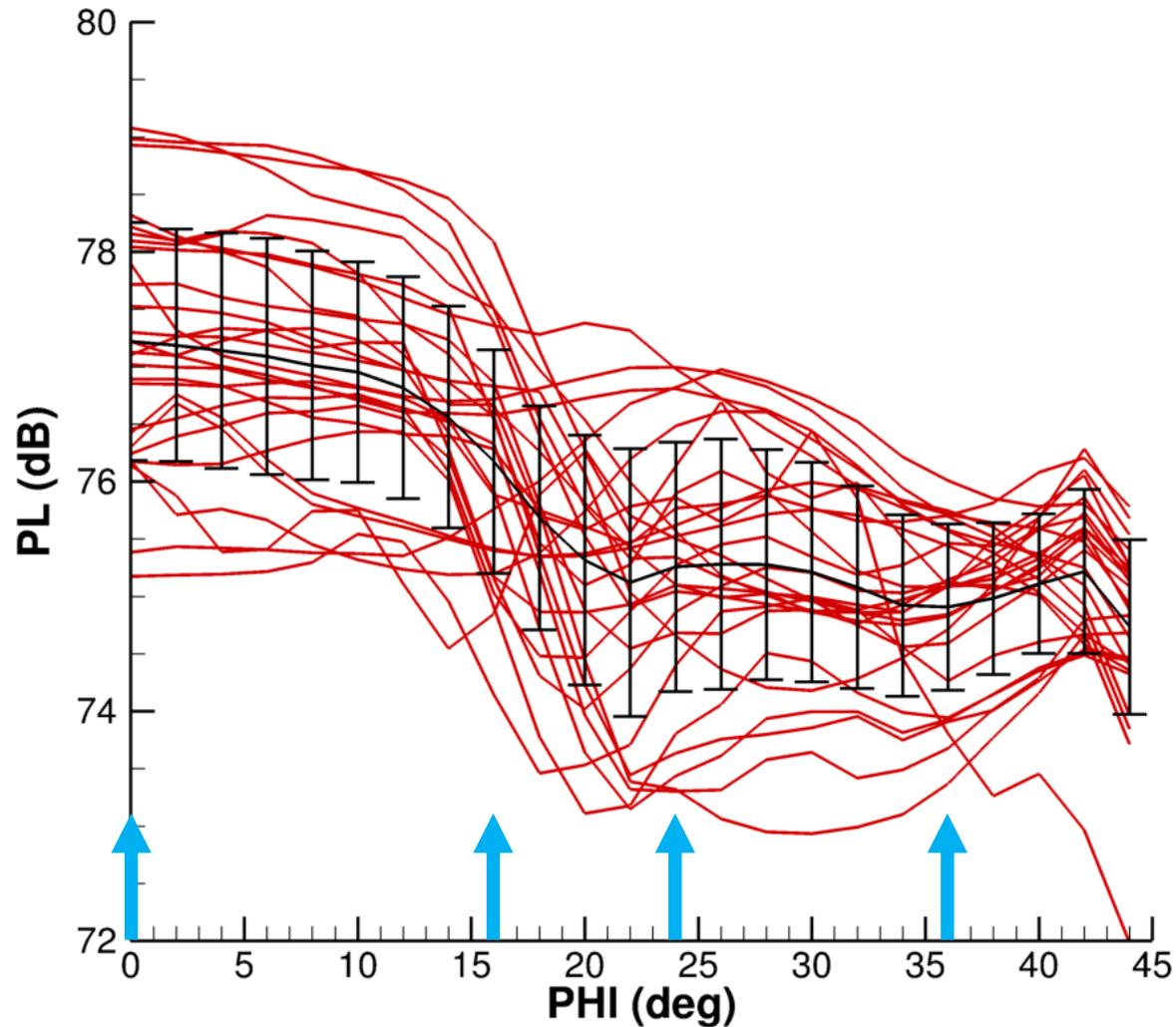
# C608 MEDIUM-GRID PL



# C608 FINE-GRID PL



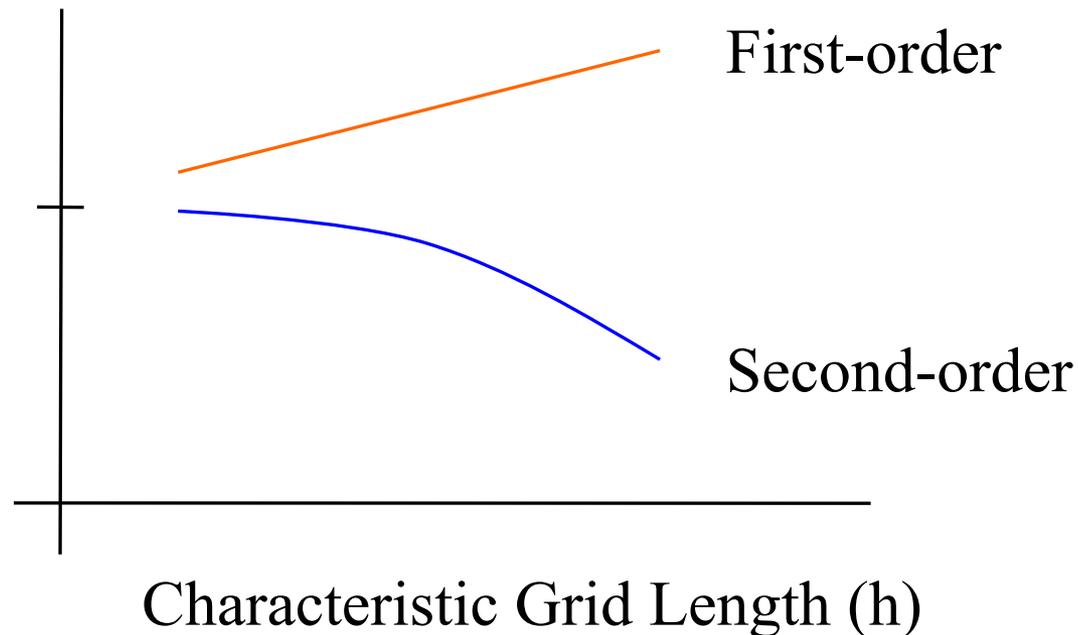
# C608 FINE-GRID PL



# EXPECTED GRID CONVERGENCE

Consistent methods should approach a value as the grid is refined to “zero”  $h$

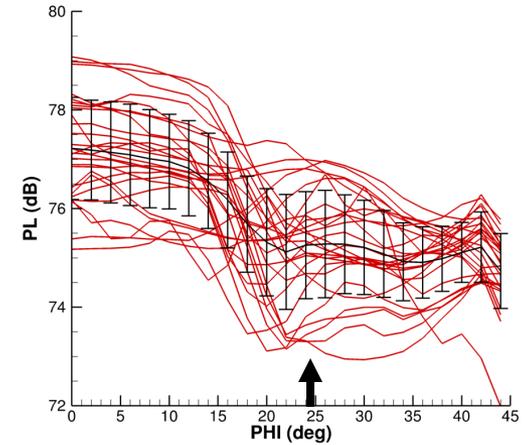
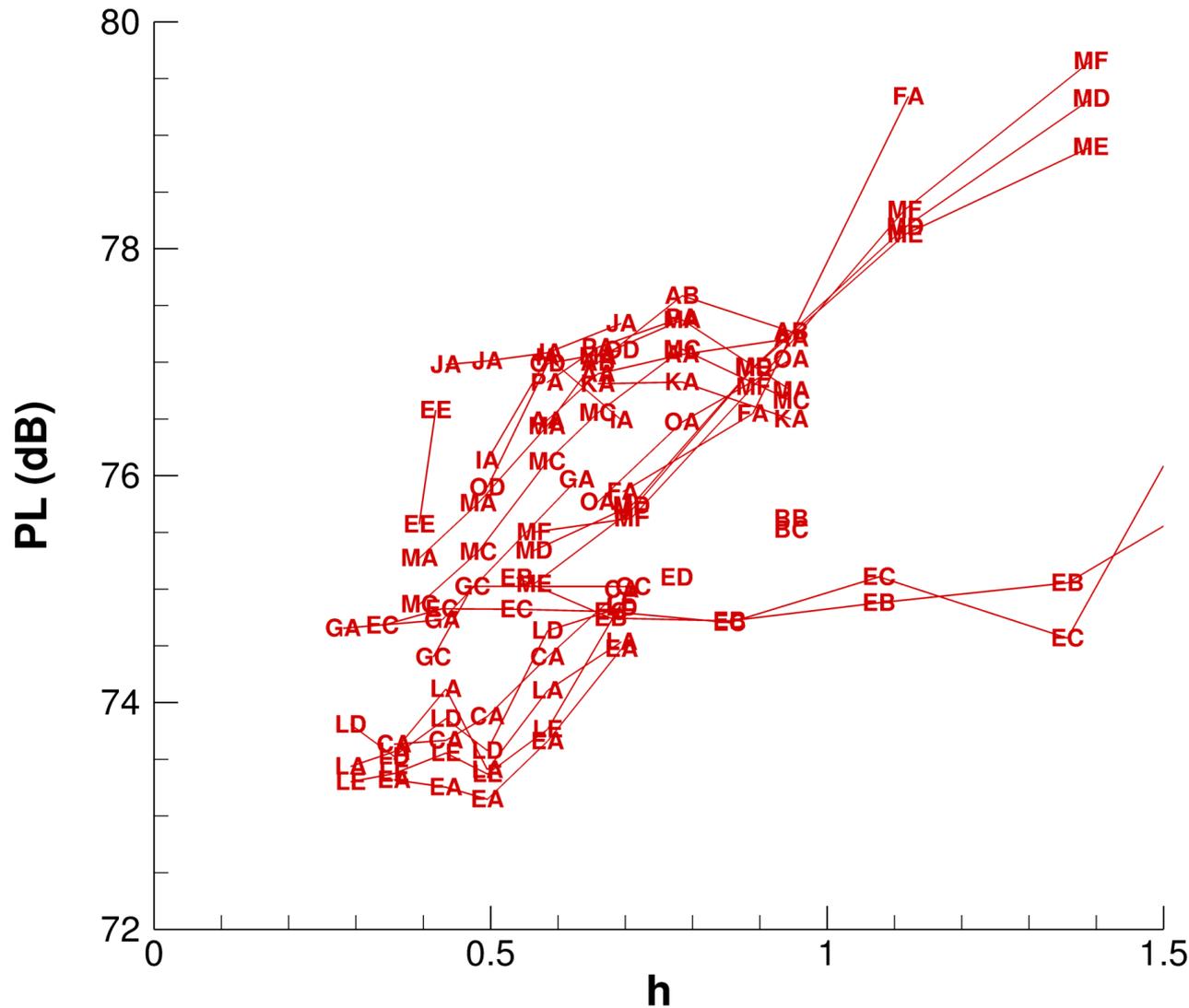
Ten million control volumes is  $h=1$







# C608 PL GRID CONVERGENCE



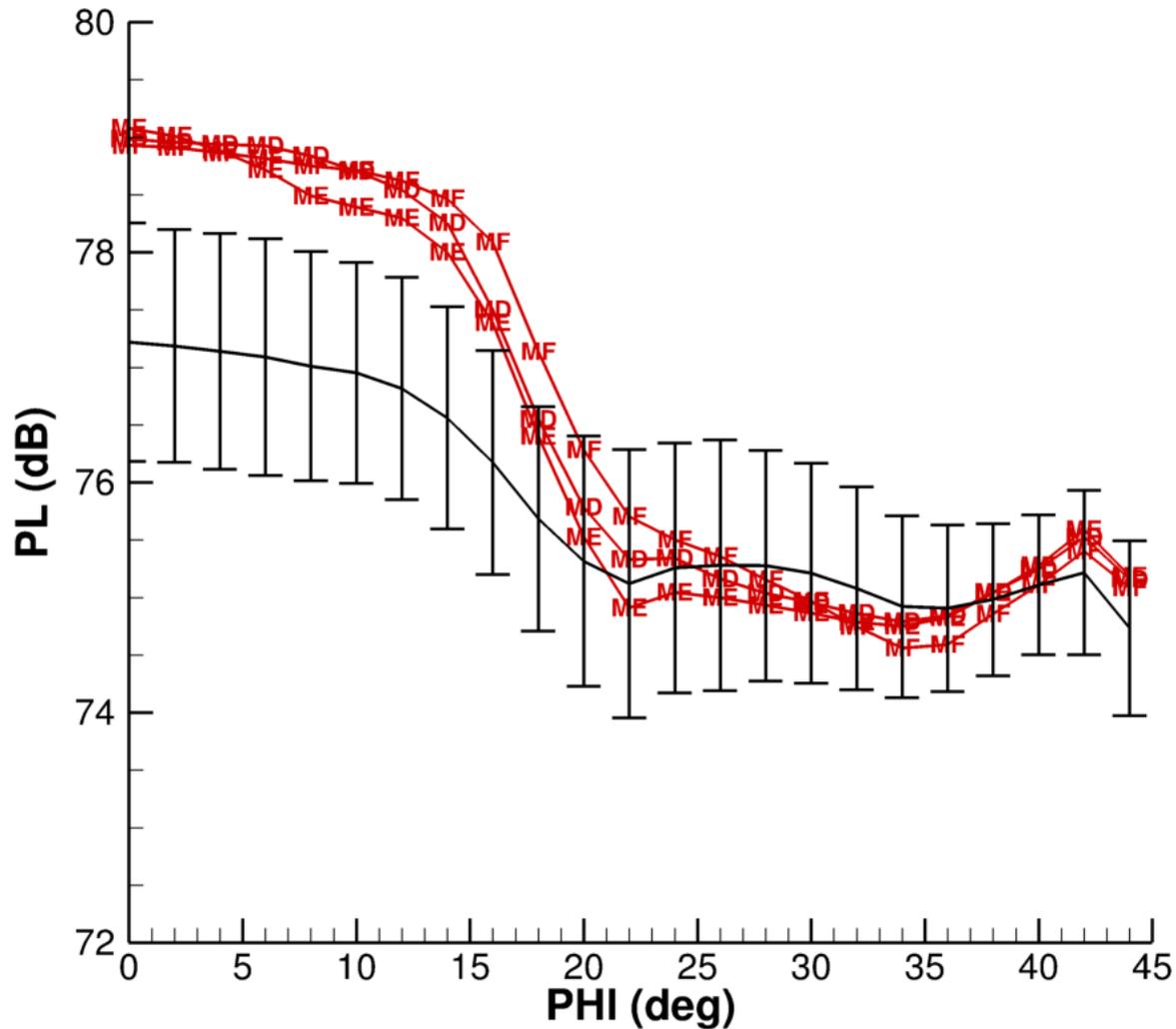
PHI=24



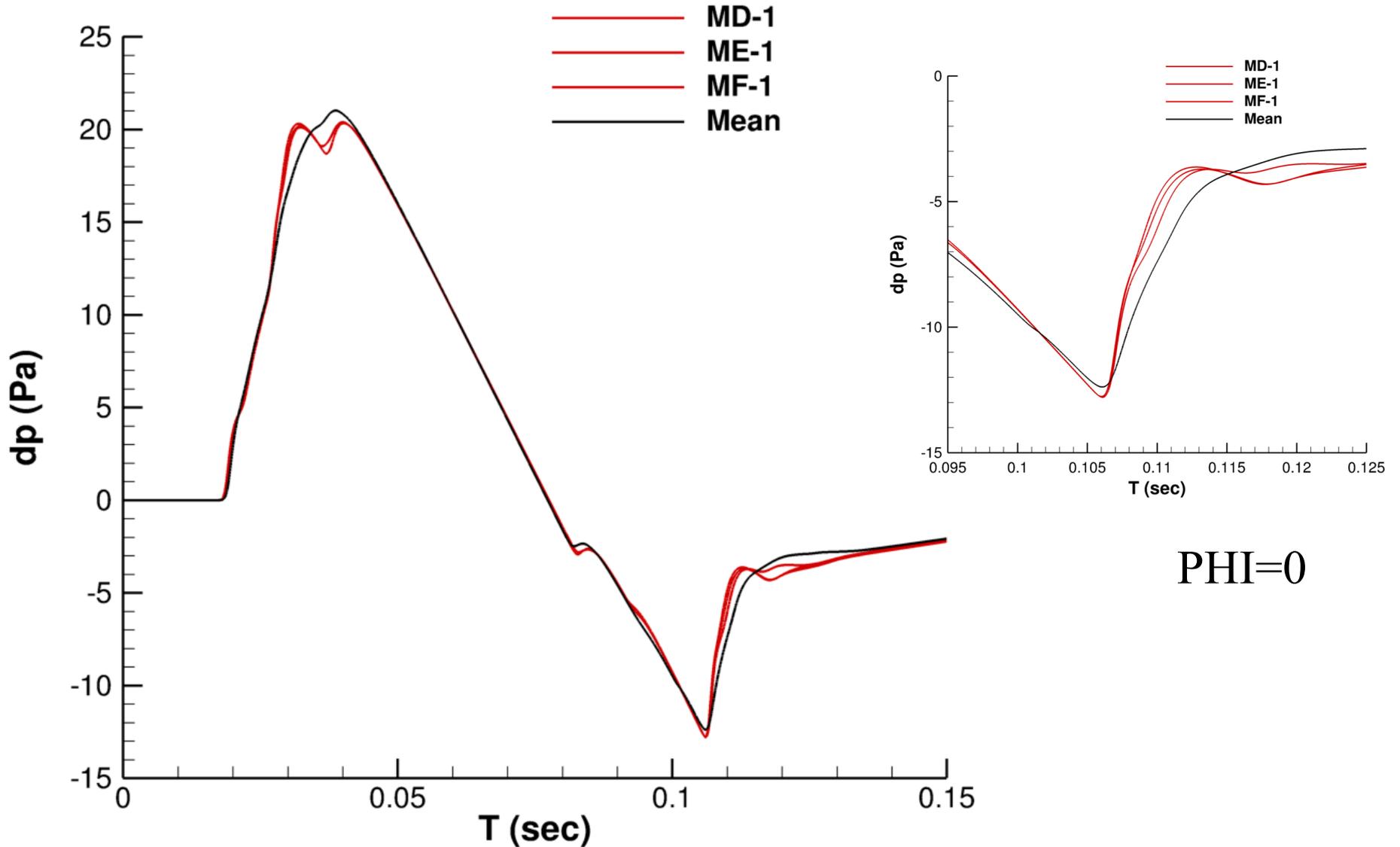
# IDENTIFICATION OF OUTLIERS

- Goal is an objective tool to identify and learn from difference in submissions
- Pointwise standard deviation is an imperfect tool
  - Not suited to small sample size
  - Distribution of submissions is not normal
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- Focus on submissions exceeding 1 standard deviation for  $\text{PHI} < 15$

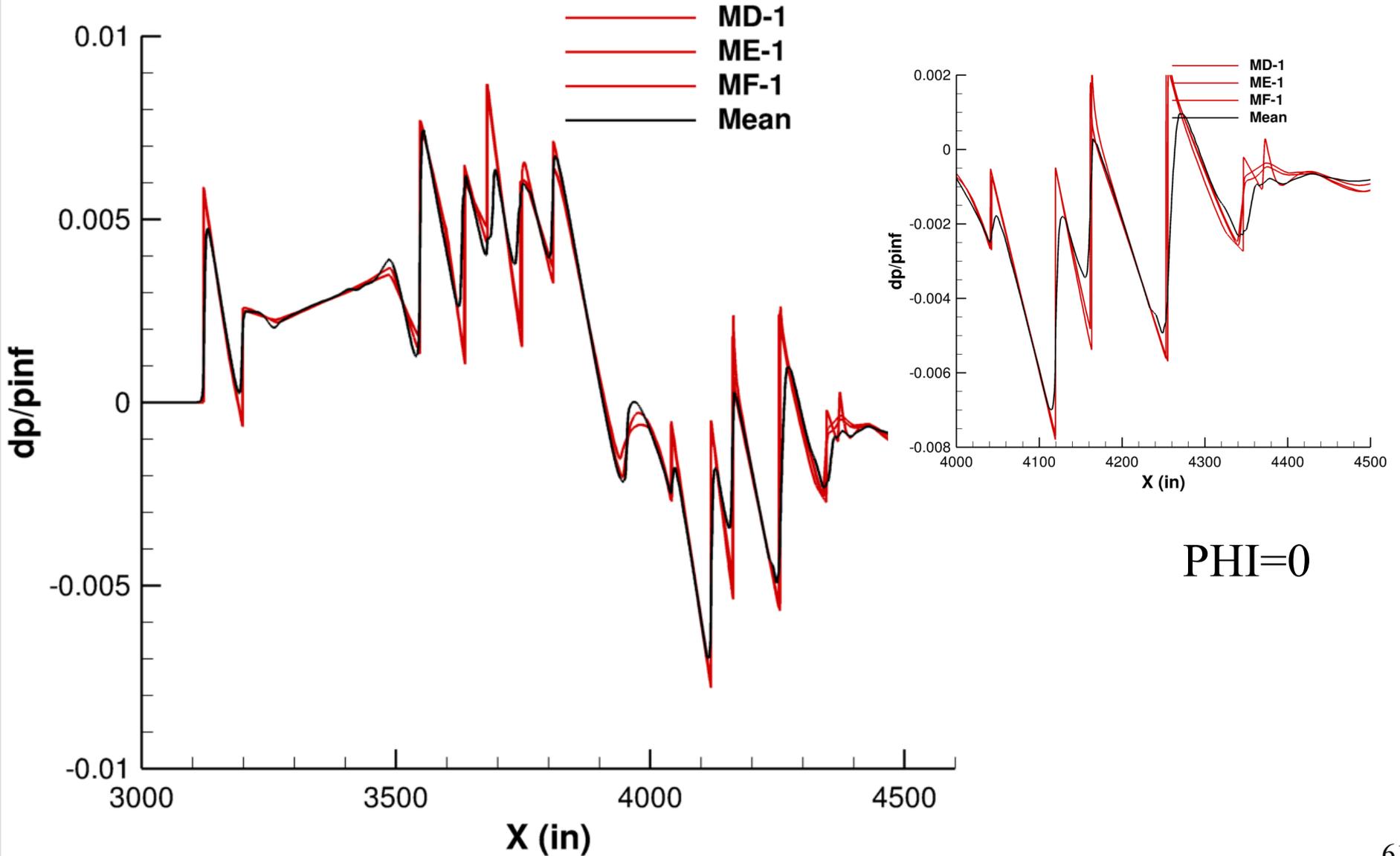
# C608 PL HIGH OUTLIERS



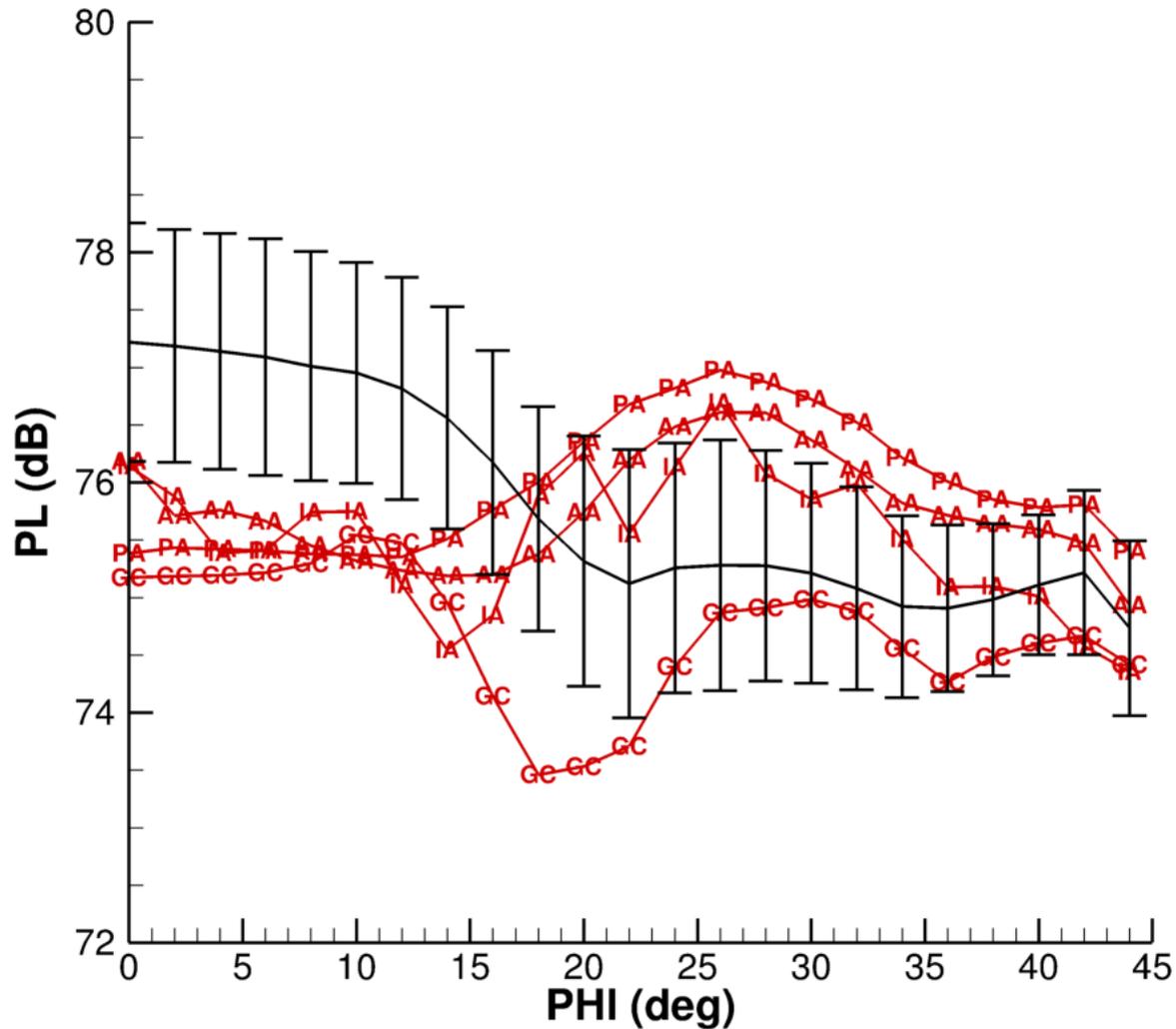
# C608 MD, ME, MF GROUND



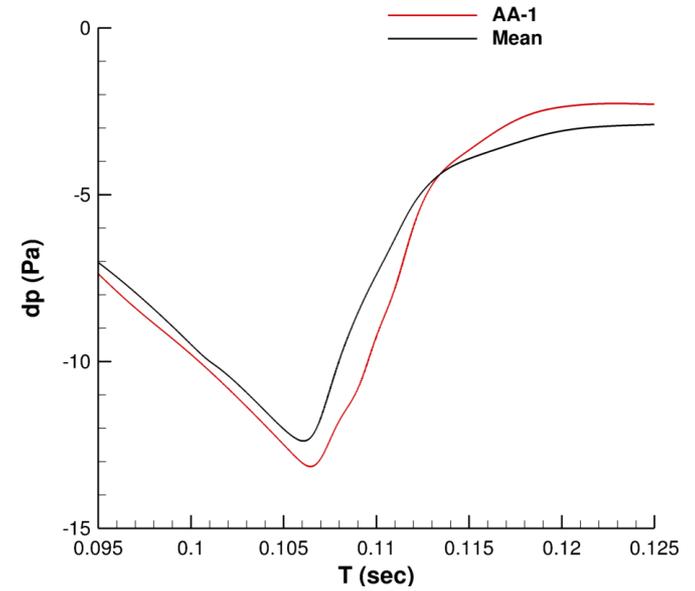
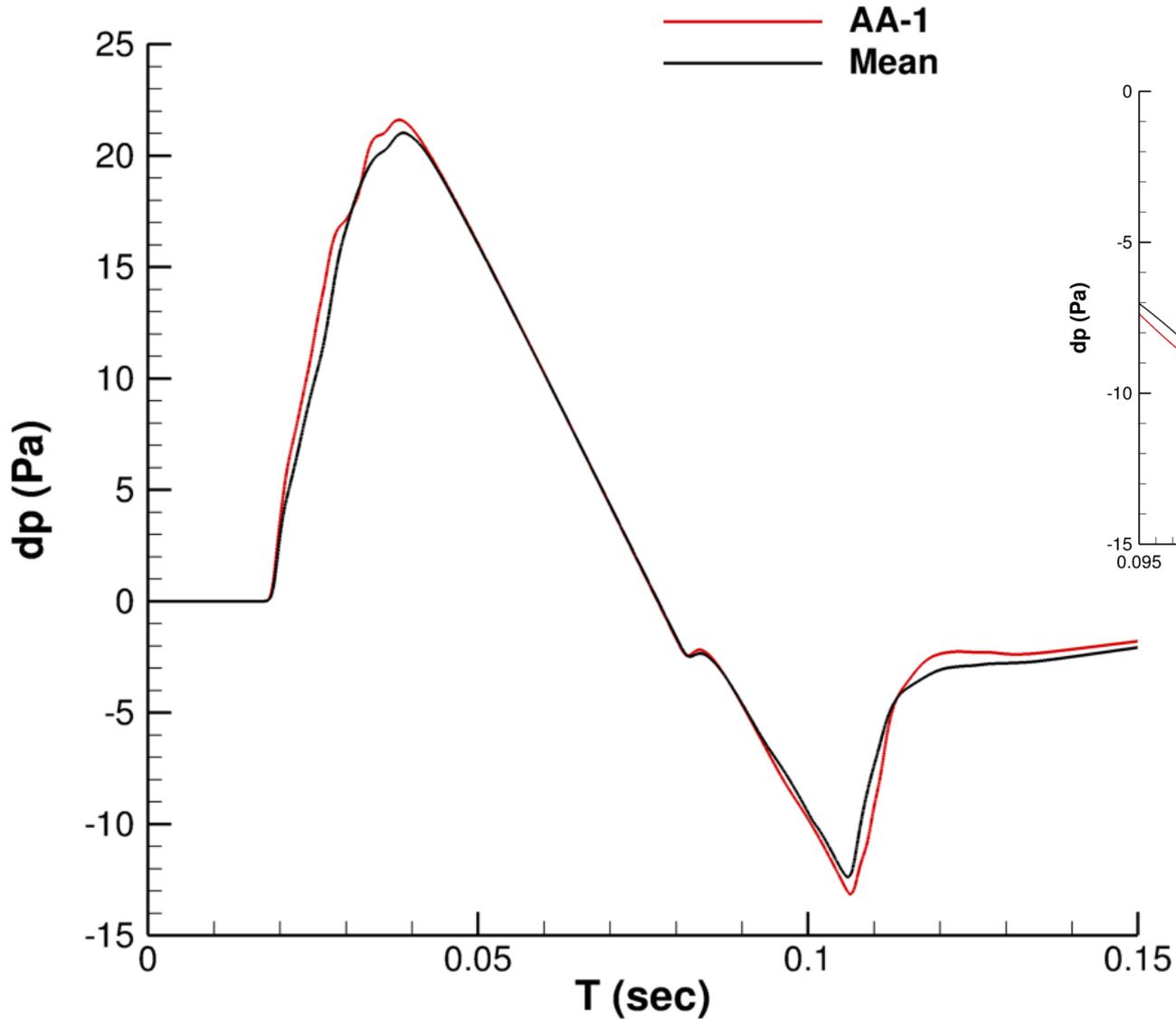
# C608 MD, ME, MF NEARFIELD



# C608 PL LOW OUTLIERS

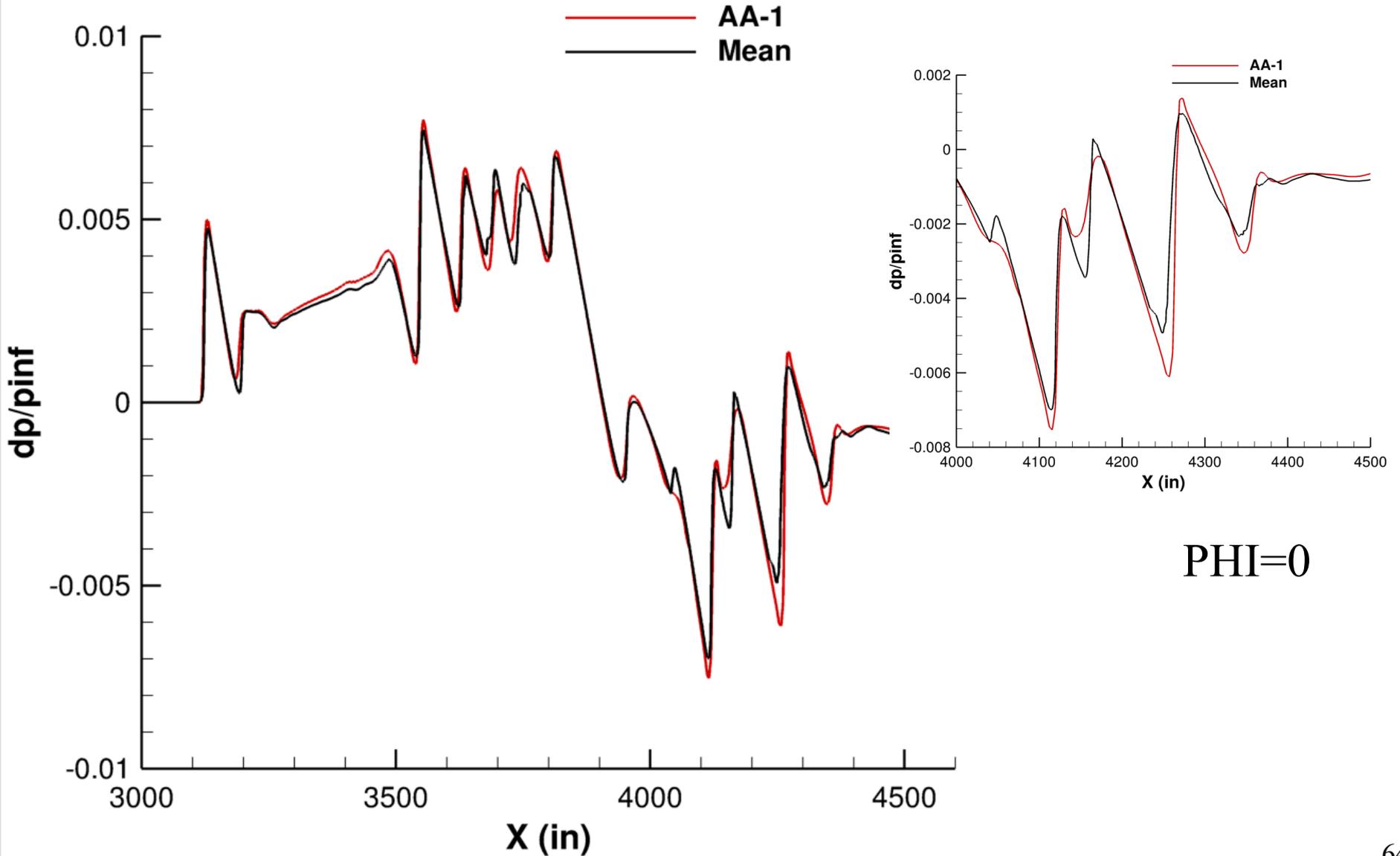


# C608 AA GROUND

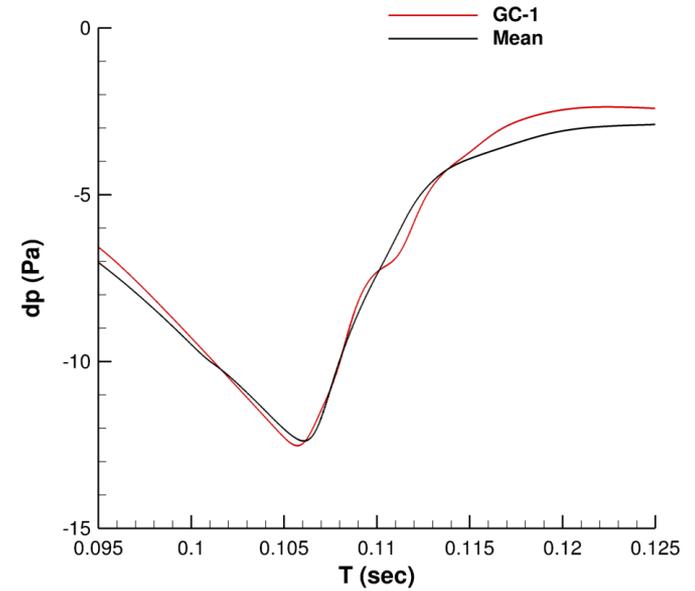
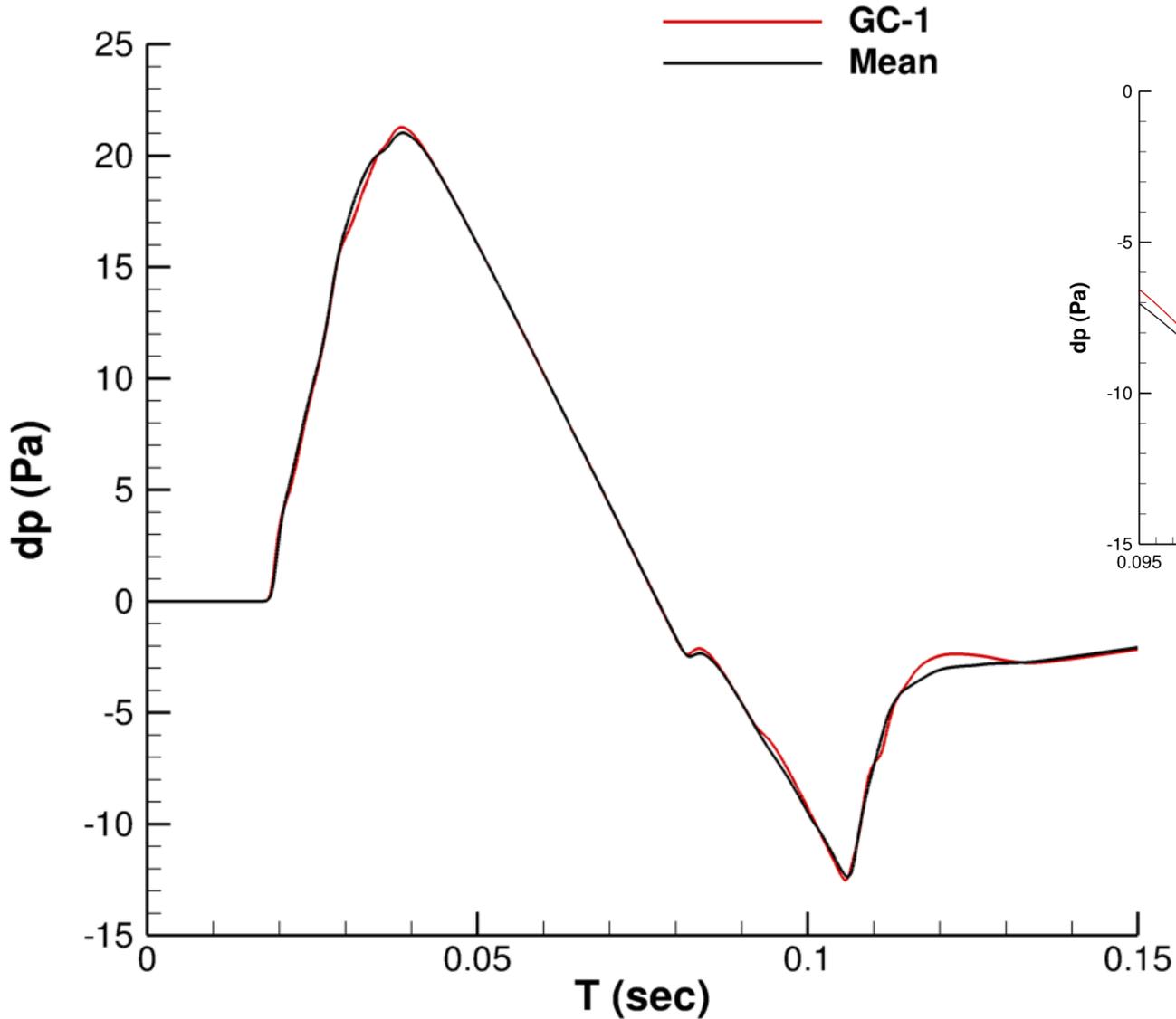


PHI=0

# C608 AA NEARFIELD

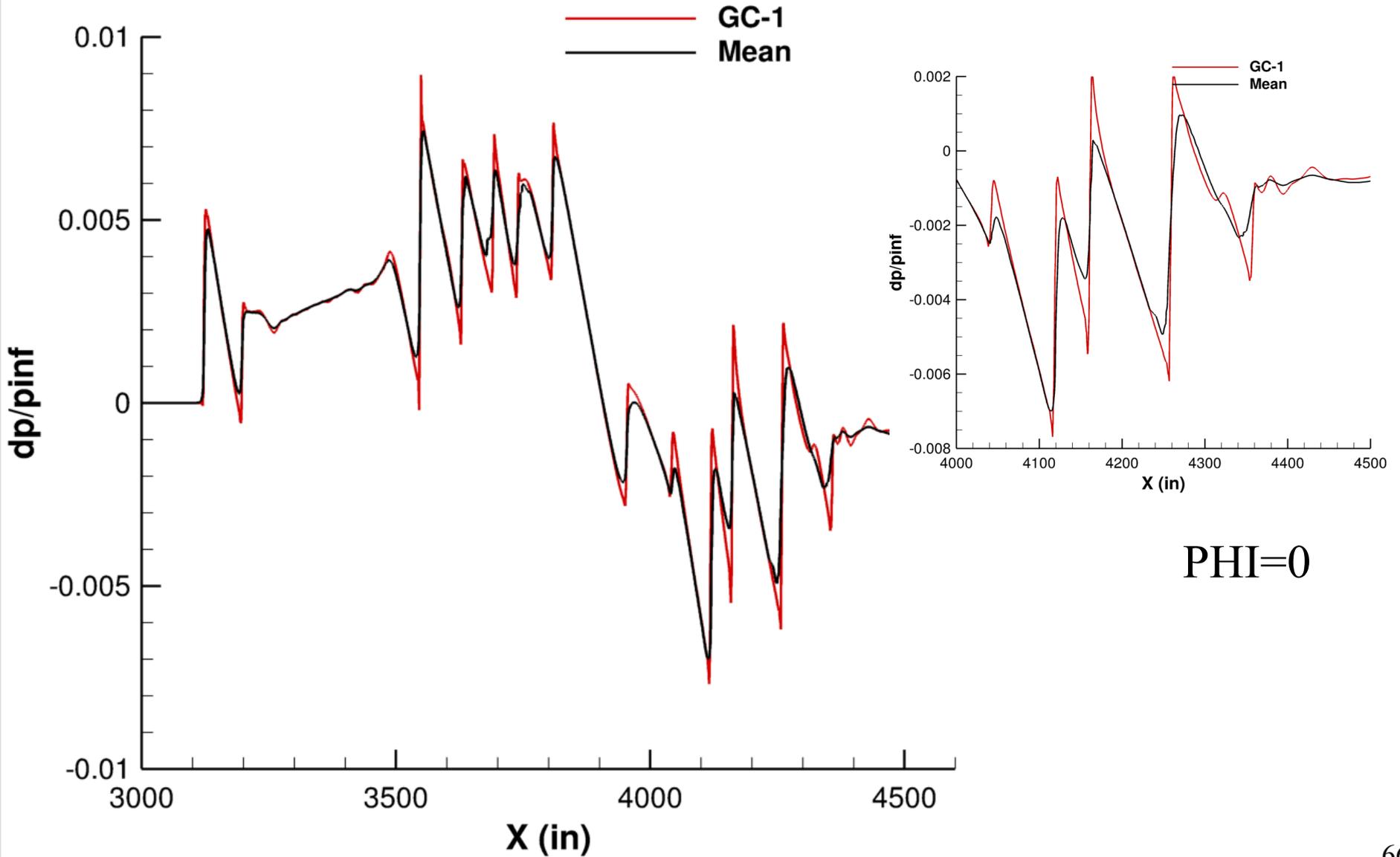


# C608 GC GROUND

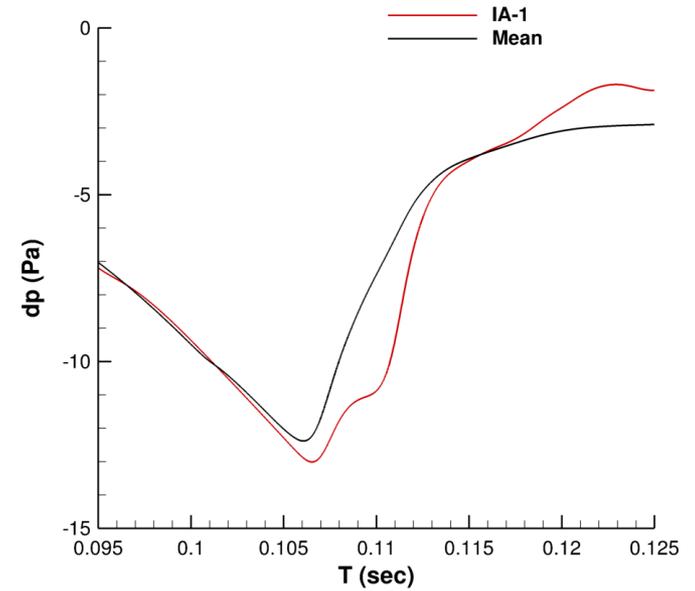
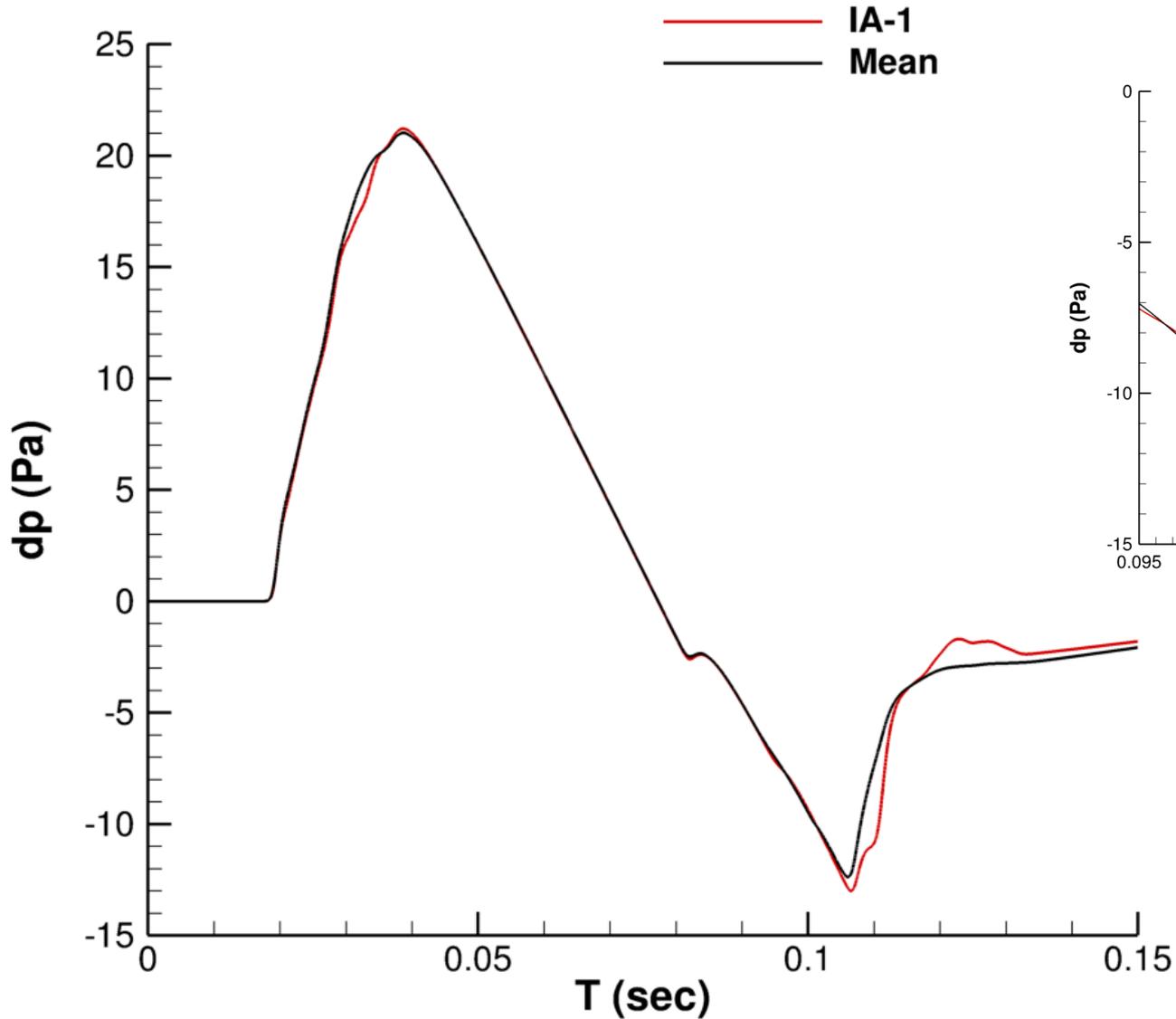


PHI=0

# C608 GC NEARFIELD

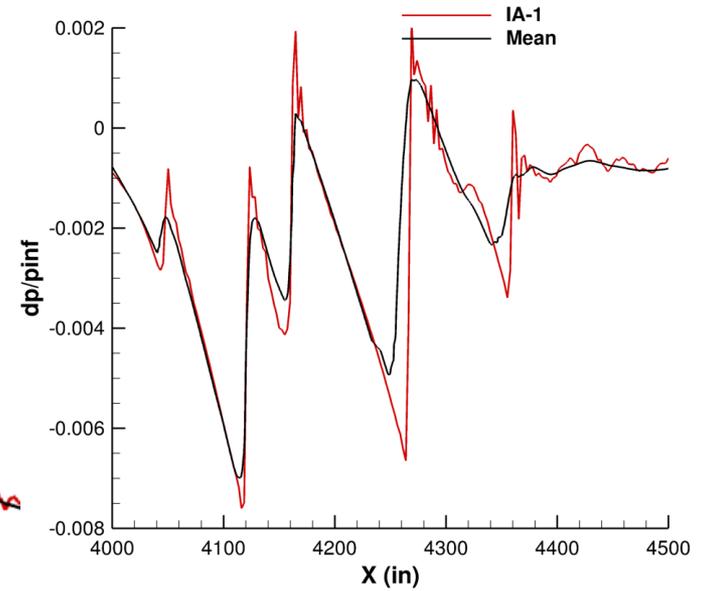
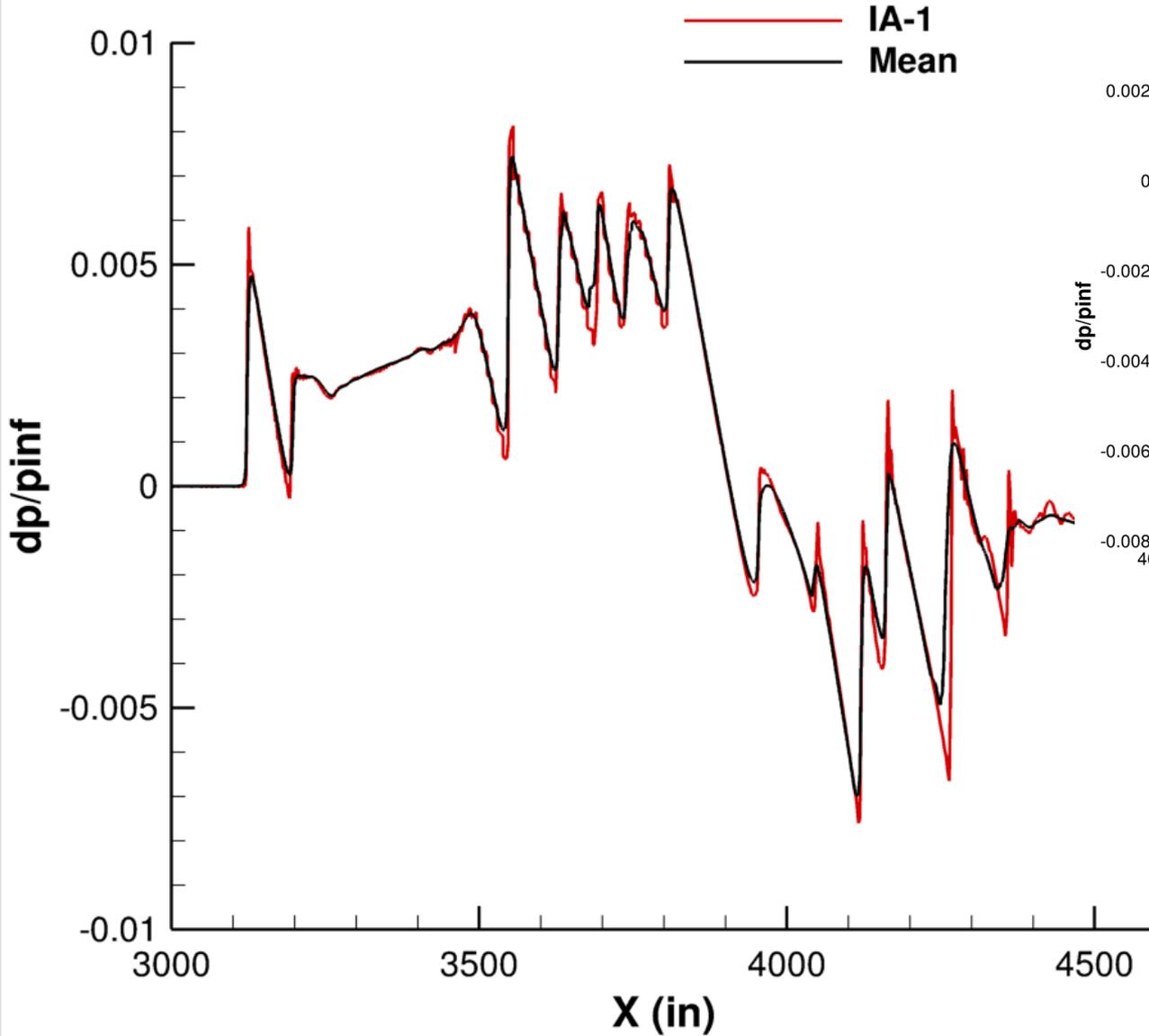


# C608 IA GROUND



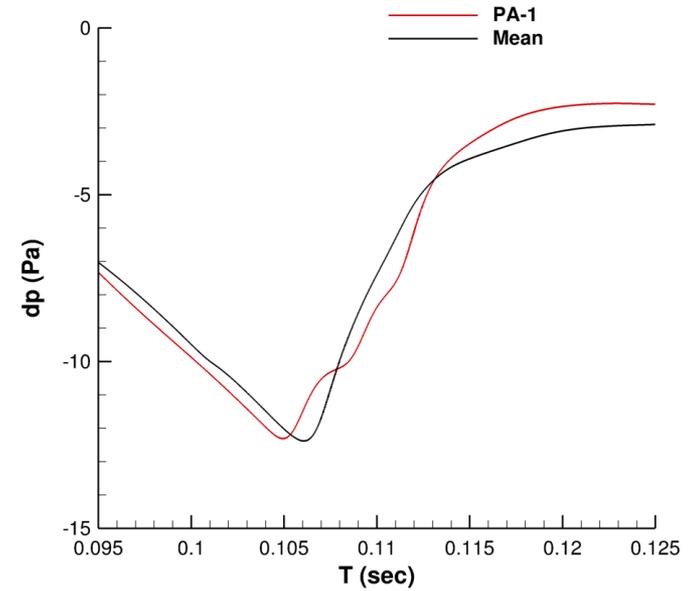
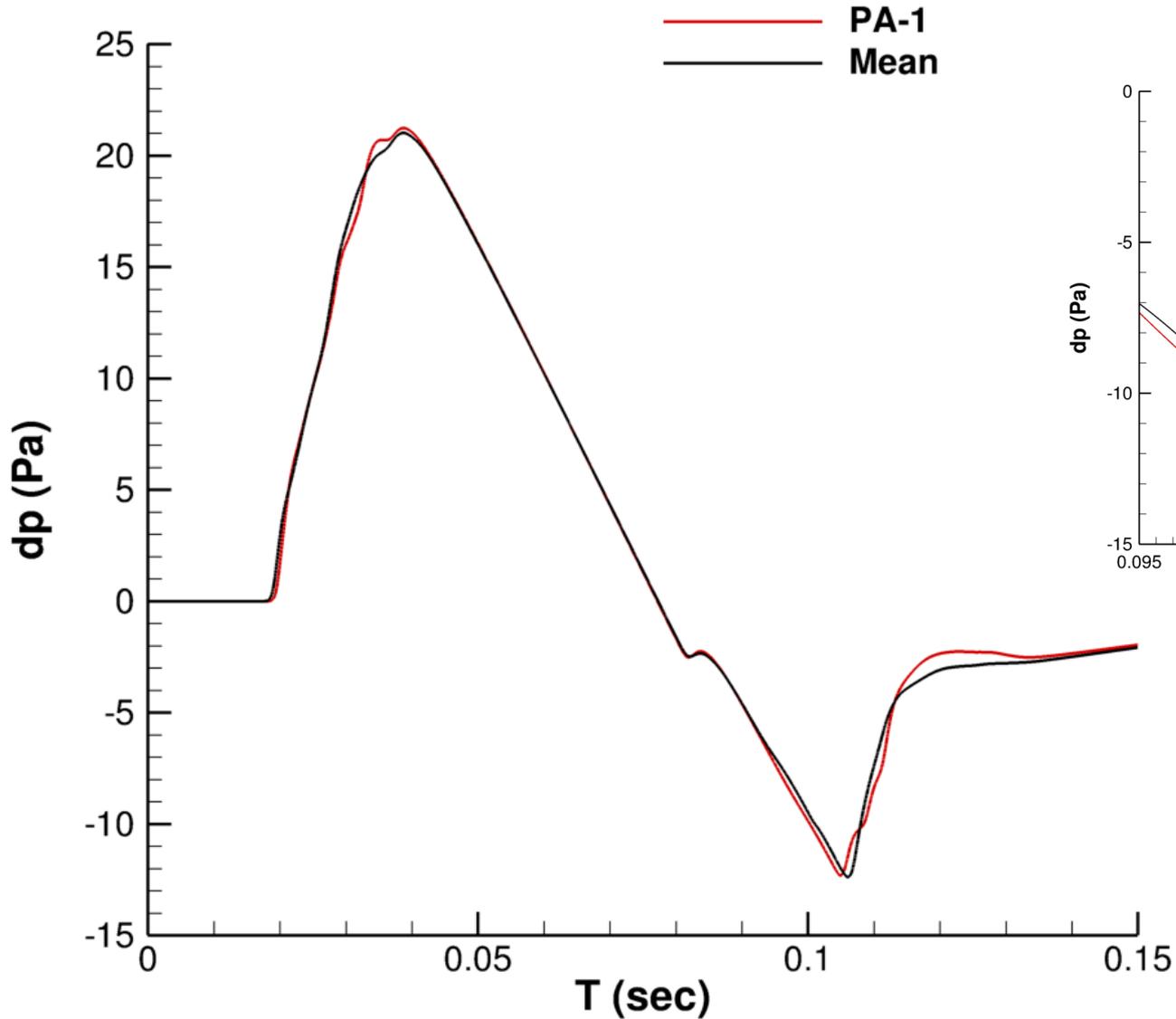
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# C608 IA NEARFIELD



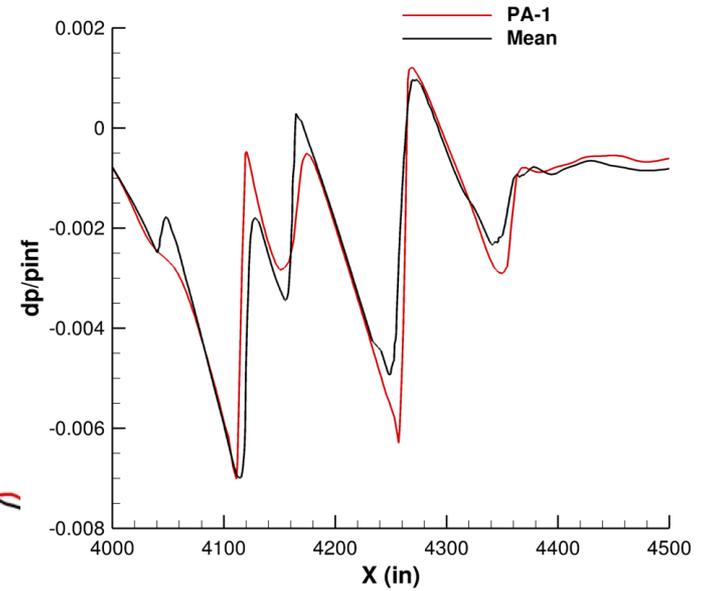
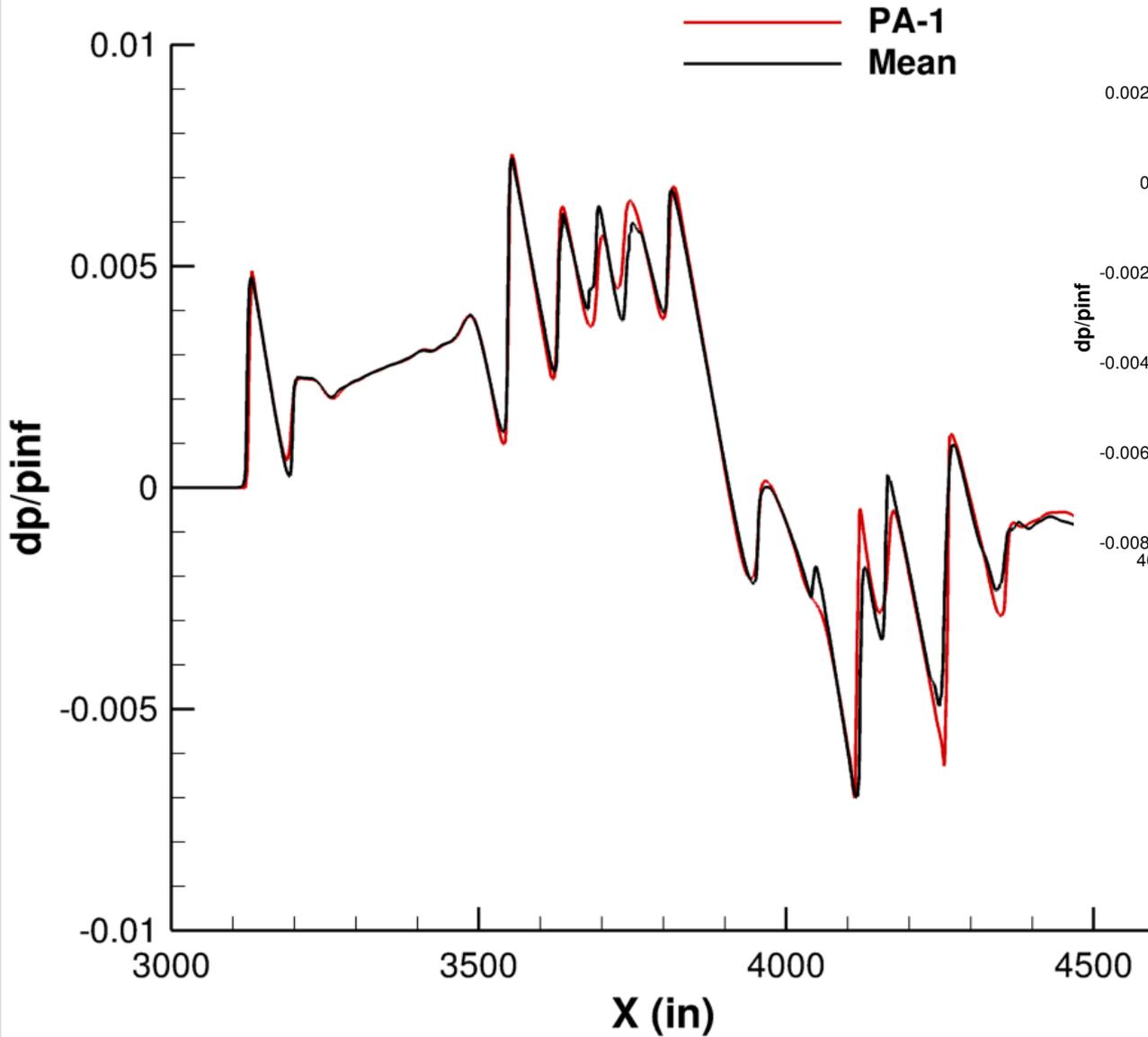
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# C608 PA GROUND



PHI=0

# C608 PA NEARFIELD

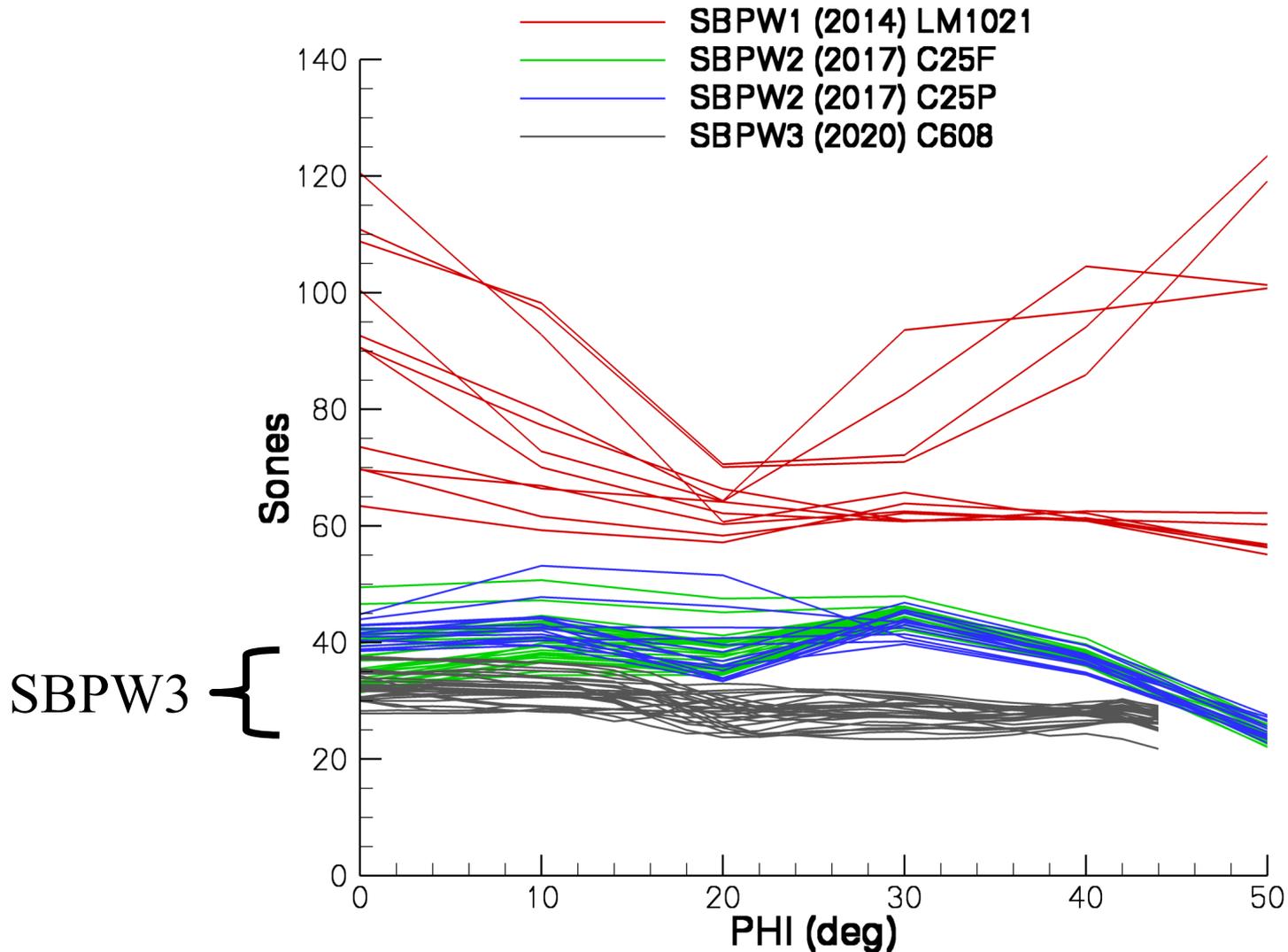


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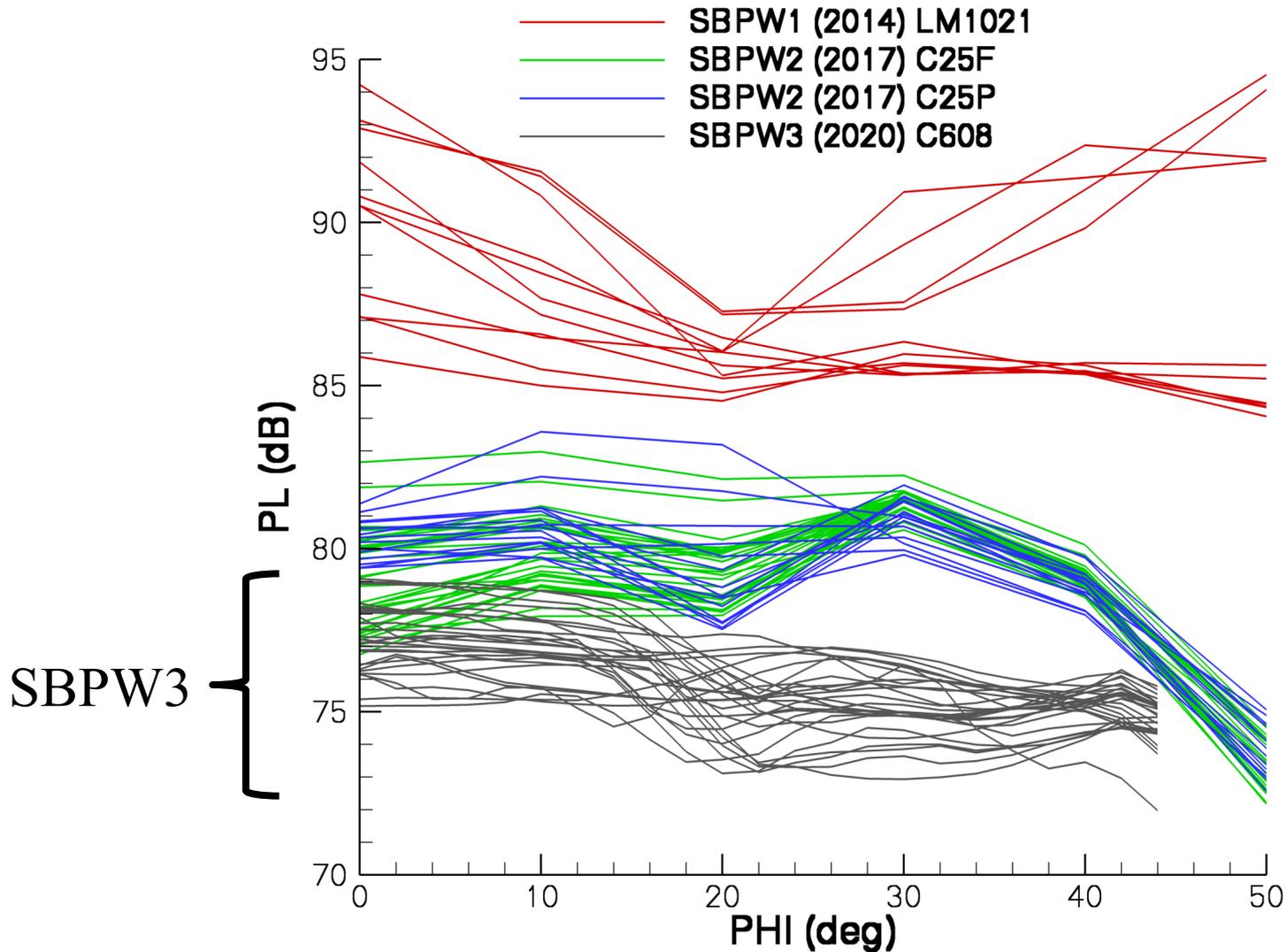
# SUMMARY

- C608 is the quietest (and therefore, the hardest to predict) C608 includes propulsion boundary conditions and a representative level of geometry complexity
- Variation of the aft deck lip shock and other tail shocks has the largest impact on PL
- The variation is lowest of SBPW with the strictest criterion for outliers
  - A coverage factor of one was used (outliers exceed 68% likelihood) via sample mean and standard deviation
  - SBPW-2 coverage factor of two (outliers exceed 95% likelihood) via box and whisker plots

# SBPW FINE-GRID LOUDNESS (SONES) CARPET



# SBPW FINE-GRID PL CARPET



# CONCLUSIONS

- Sincere thank you to all the participants!
- These cases included propulsion boundary conditions and realistic geometry making them the hardest attempted in the workshop series
- The variation is lowest of SBPW, requiring a stricter criterion to identify outliers
  - A coverage factor of one was used (outliers exceed 68% likelihood) via sample mean and standard deviation
  - SBPW-2 coverage factor of two (outliers exceed 95% likelihood) via box and whisker plots

# OPTIONAL CASE PARTICIPATION

How to encourage more participation during or after workshop

- Uncertainty Quantification (UQ) run matrix
- Multiscale Mach adapted grids

# NEXT STEPS

- Participant submission updates (10-FEB-2020)
- Further analysis based on feedback (some surface and volume solutions available)
- AVIATION papers and AIAA Journal of Aircraft Special Section
  - Provide participant submissions and ensemble data to AVIATION authors for comparison plots and independent analysis
- Enable a foundation for research
  - Midfield space marching solver
  - Propagation including over the top or secondary boom
  - Other opportunities?